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CORONA J

PERFORMANCE EVALUATION REPORT

MISSION 1026-1 and 1026-2

FTV 1620; J-25

20 June 1966

Approved:

Mgr.

Advanced Projects

Approved:

Mgr.

Program

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22 July 1966

To: V. Webb
C. Murphy
A. Johnson

Thru:

From:

Subject: MISSION 1026-1 AND 1026-2 FINAL REPORT

Enclosed is the Final Performance Evaluation Report for
Mission 1026-1 and 1026-2.

[REDACTED]
Manager
Advanced Projects

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FOREWORD

This report details the performance of the payload system during the operational phase of the [REDACTED] Flight Test Vehicle 1620.

Lockheed Missiles and Space Company has the responsibility for evaluating payload performance under the Systems Integration and "J" System contracts.

This document is the final payload test and performance evaluation report for Missions 1026-1 and 1026-2 which was launched on 28 October 1985.

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INTRODUCTION

This report presents the final performance evaluation of Missions 1026-1 and 1026-2 of the Corona Program. The purpose of this report is to define the performance characteristics of the J-25 payload system and to identify the source of in-flight anomalies.

The performance evaluation was jointly conducted by representatives of Lockheed Missiles and Space Company (LMSC) and ITEK at the facilities of NPIC and AFSPPF. The off-line evaluation using Corona engineering photography acquired over the United States was performed at the individual contractors plants.

The quantitative data used for this report is obtained from government organizations. The diffuse density data, and MTF/ALM resolution are produced by AFSPPF. The vehicle attitude error and image smear correlation times are made at NPIC who also supply the Payload System Summary and MTF/ALM resolution reports published [REDACTED]

Computer programs developed by A/P are utilized to calculate and plot the frequency distribution of the various contributors to image smear to permit analysis and correlation of the conditions of photography to the information content and quality of the acquired pictures. Computer analysis of the exposure, processing and illumination data provides the necessary data to analyze the exposure criteria selected for the mission.

SECTION I

SYSTEM PERFORMANCE

A. MISSION OBJECTIVES

The payload section of Mission 1026, placed into orbit by Flight Test Vehicle #1620 and LV-2A booster #439, consisted of two panoramic cameras, two Stellar-Index cameras, two Mark 5A recovery capsules and a space structure to enclose the cameras and provide mounting surfaces for all equipments. Figure 1-1 presents an inboard profile of the J-25 payload system. This Corona "J" system is designed to acquire search and reconnaissance photography of selected areas of the earth from orbital altitudes. The planned mission was two, 5 day photographic periods with no deactivate period.

B. MISSION DESCRIPTION

The payload was launched from Vandenberg Air Force Base (VAFB) at 2117:12 Z (1417:12 PDT) on 28 October 1965. Ascent and injection were normal and the achieved orbit was within nominal tolerances. Tracking and command support was effected by the Air Force Satellite Control Facility consisting of tracking and command stations at [REDACTED]

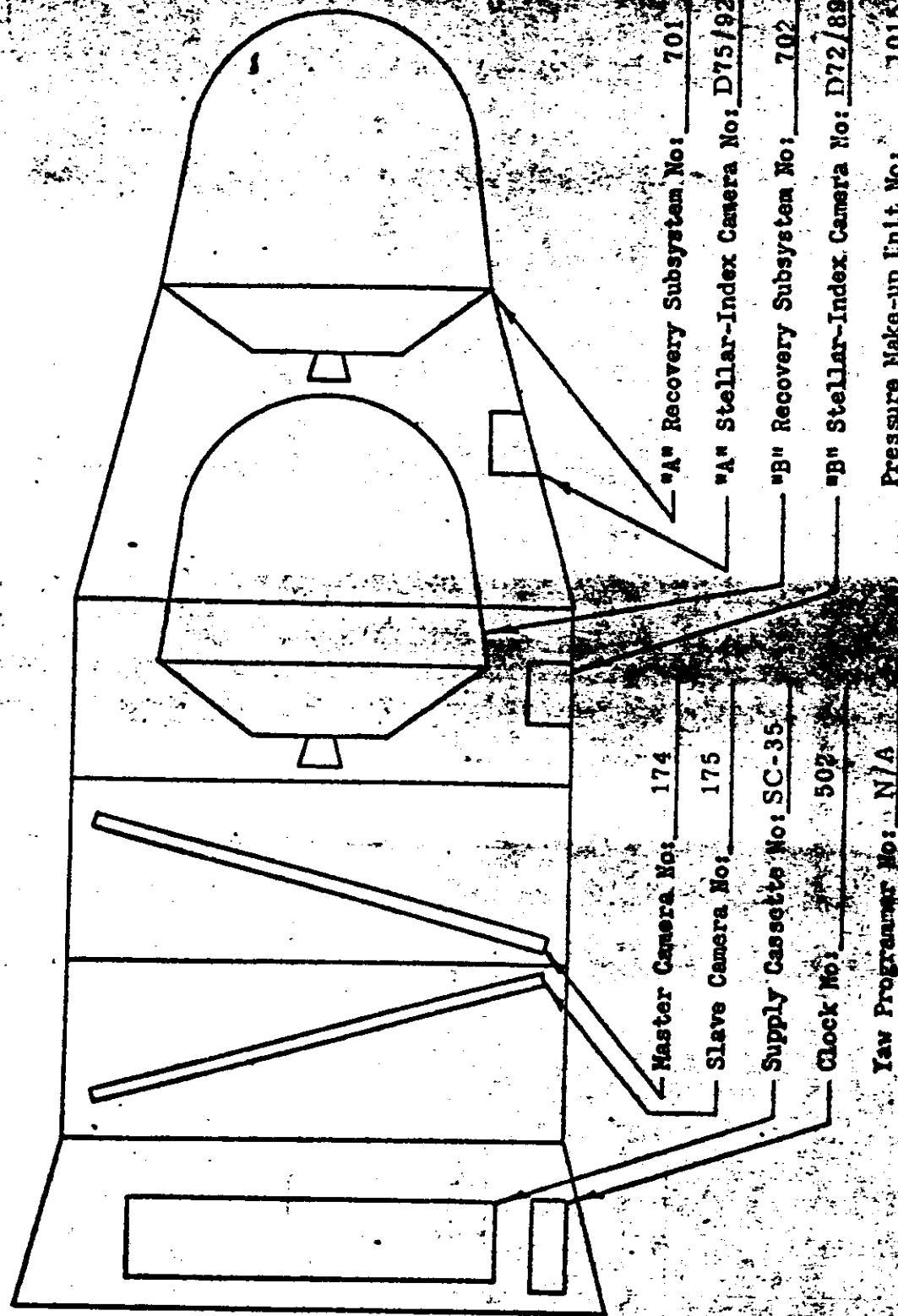
[REDACTED] under central control of the Satellite Test Center at Sunnyvale, California. Mission 1026-1 consisted of a 5 day operation and was completed by air recovery on 2 November 1965. Mission 1026-2 was completed with an air recovery on 7 November 1965 following a 5 day photographic operation.

The comparison of the planned and actual orbit parameters is tabulated as follows:

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SCHEMATIC INBOARD PROFILE - CORONA J SYSTEM

MISSION 1026



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ORBITAL PARAMETERS

<u>Parameter</u>	<u>Predicted</u>	<u>Orbit 1 Actuals</u>
Period (Min.)	90.72	90.77
Perigee (N. M.)	99.8	95.12
Apogee (N. M.)	238.1	240.6
Inclination (Deg.)	75.00	74.98
Perigee Latitude (Deg. N.)	23.00	10.05
Eccentricity	0.0192	0.02041

C. PANORAMIC CAMERAS

The Master instrument operated normally through both missions. The photo quality suffered from heavy haze, low sun elevations and 50% cloud cover.

The Slave instrument photo quality was rated as being slightly better than the Master. The center of format switch operated intermittently from pass D-24 thru D-70.

The horizon cameras on the sun side of the vehicle exhibited veiling on both missions.

D. STELLAR-INDEX CAMERAS

The "A" S/I produced star imagery adequate to determine the vehicle attitude. Four stellar frames were affected by shutter malfunction. The index camera produced no anomalies.

The "B" S/I camera operation is rated as excellent.

E. OTHER SUB-SYSTEMS

The clock, instrumentation, command and thermal control subsystems performed satisfactorily.

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F. CONCLUSIONS

Mission 1026 achieved its objective of search and reconnaissance, however atmospheric conditions obscured some target areas.

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SECTION 2**PRE-FLIGHT SYSTEMS TESTS****A. ENVIRONMENTAL TESTING****1. Test Objective**

As a standard procedure, the J payload systems are subjected to thermal/altitude environmental testing which simulates orbital environment. One of the purposes of this test is to demonstrate the system susceptibility to corona discharge. Such discharge fogs the film thus degrading the operational photography.

2. Test Summary

The J-25 payload system was subjected to an environmental TASC Chamber test from July 10 through July 17, 1965. The environmental test consisted of 13 orbits in the "B" mode. The payload was removed from the chamber on two occasions as a result of two fail-safes. This was done between orbit 13 in the "A" mode and the "A" recovery sequence.

Panoramic camera #1 experienced erratic payload metering, cycle period in excess of the 2.15 sec. maximum at one point, cycle rate errors ranging from 1.9% fast to 2.9% slow, and two fail-safes.

Instrument #2 performance was satisfactory except the cycle rate errors ranged from 0 to 4.5% slow in the "A" mode and 0.4% to 2.6% slow in the "B" mode.

The stellar/index camera performance was satisfactory throughout both the "A" and "B" mode of operation.

The clock performance was out of allowable offset limits.

The instrumentation system performance was satisfactory with the exception of several noisy idler commutators and Sanborn channel setup discrepancies.

The PMU system operation was satisfactory with the average gas depletion rate being somewhat higher than normal.

The temperature environment was cold for the "A" mode and near 70° for the "B" mode.

3. Panoramic Camera Performance

Instrument #1 performance disclosed the following problems:

- a. Erratic metering in the "A" mode.
- b. Cycle rate predictability exceeded the 1% specification limit in the "A" mode.
- c. Two fail-safes occurred in the "A" mode and the payload was pulled out of the rails.
- d. Cycle period exceeded the maximum limit of 2.15 sec. on one operation.

Payload metering was erratic as observed on the takeup idler. Uneven payload motion was also indicated on the supply idler monitor. On orbit 2, "B" mode, the cycle period exceeded the maximum allowable of 2.15 secs. This occurred at 2130 seconds up the ramp of R4A1 and was 1.9% faster than predicted. A cycle period was obtained in orbit 2, "A" mode, which was within 1% of the predicted cycle period.

The instrument had two fail-safes during the test. The first occurred near the end of orbit 13 during the mono portion of a stereo surpress operation. The chamber was brought to atmospheric pressure and the system was inspected and no anomalies were detected. The system was reinstalled in the environmental chamber and the test was restarted at orbit 12 of the "A" mode. Fail-safe occurred again during the stereo portion of the stereo surpress operation in orbit 13. The system was again removed from the chamber. The master instrument film was found to be out of the rails and the takeup cassettes of the "A" SRV was found to be rubbing. Erratic payload metering was evident on both instruments prior to each fail-safe. The system

was transferred to the "B" mode by performing a cut and wrap recovery sequence. A confidence run was made and the system reinstalled in the chamber. Instrument #1 completed the test without further problems.

Instrument #2 performance was satisfactory except cycle rate errors ranged from 0 to 4.5% slow in the "A" mode and 0.4 to 2.6% slow in the "B" mode. The chamber temperature was low in the "A" mode and more nearly normal in the "B" mode. Analysis is continuing to establish the degree of correlation, if any, between cycle rate error and temperature.

The supply cassette rotation monitor indicated supply cassette rotation in the latter half of the "B" mode of operation.

A tabulation of cycle rate data from the chamber test is included as Table 2-1.

4. Stellar/Index Performance

Both the "A" and the "B" stellar/index cameras performed properly throughout the test. The "A" index idler T/M was noisy.

5. Clock Performance

The clock offset was 1 millisecond per hour in both the "A" and "B" mode when compared to IRIG C. This exceeds the offset limit of 2 milliseconds per 12 hours.

6. Instrumentation Performance

Instrumentation performance throughout the test was satisfactory. Minor problems such as improper Sanborn calibrations and noisy idler contacts occurred.

The status commutator and temperature commutator has no open points.

The film footage pots and cycle counters showed good correlation throughout the test.

J-25 TASC CHAMBER TEST NO 1

TEST CYCLE RATES

REV/MODE	RAMP	T.U.R.	ACT.	INST 174 SEEM. CALIB.	DEV.	ACT.	INST 175 SEEM. CALIB.	DEV.	DIFF.	174/175
0 A	8	2	0	5.570	5.479	-1.66	5.600	5.539	-1.10	0.54
1 A	7	7	400	3.525	3.494	-0.90	3.520	3.489	-0.88	-0.14
2 A	4	1	1430	2.190	2.216	1.18				
2 A	4	1	1610				2.208	2.207	-0.07	
2 A	4	1	2130	2.170	2.189	0.89	2.200	2.200	0.01	1.38
3 A	5	8	730	2.893	2.879	-0.48	2.887	2.869	-0.61	-0.21
3 A	5	8	1550	2.458	2.446	-0.47	2.460	2.437	-0.95	0.08
4 A	7	7	2280	2.580	2.559	-0.83	2.580	2.549	-1.22	0
5 A	8	2	340	5.190	5.076	-1.07	5.178	5.117	-1.19	0.90
6 A	5	8	1090	2.733	2.693	-1.47	2.753	2.683	-2.60	0.73
6 A	5	8	1440	2.518	2.493	-1.01	2.535	2.483	-2.09	0.68
7 A	7	7	1180	2.927	2.875	-1.81	2.957	2.865	-3.21	1.02
7 A	7	7	1590	2.610	2.562	-1.86	2.627	2.552	-2.92	0.65
8 A	7	7	2530	2.767	2.723	-1.63	2.793	2.713	-2.97	0.94
9 A	4	1	1015	2.767	2.743	-0.86	2.823	2.733	-3.29	2.02
9 A	4	1	3200	3.500	3.405	-2.78	3.540	3.400	-4.13	1.14
9 A	11	1	890	4.585	4.459	-2.82	4.680	4.477	-4.53	2.07
10 A	11	1	1880	2.283	2.271	-0.53	2.320	2.263	-2.54	1.62
10 A	11	1	2970	4.470	4.384	-1.95	4.560	4.400	-3.64	2.01

TABLE 2-1

REV/MODE	RAMP	T.U.R.	INST-174			INST 175			174/175	
			ACT.	MEAN	DEV.	ACT.	MEAN	DEV.	DIFF.	
10	A	7 7	130	3.700	3.598	-2.83	3.760	3.595	-4.58	1.62
11	A	7 7	1910	2.527	2.476	-2.05	2.550	2.467	-3.38	0.91
11	A	7 7	2270	2.603	2.554	-1.92	2.630	2.544	-3.37	1.04
12	A	8 2	1045	3.263	3.229	-1.04	3.333	3.222	-3.45	2.15
12	A	8 2	1850				2.265	2.229	-1.60	
13	A	8 2	835	3.785	3.726	-1.58	3.870	3.726	-3.88	2.25
13	A	4 1	2625	2.390	2.336	-2.30	2.420	2.327	-3.98	1.26
	A	11 1	1755	2.385	2.319	-2.86	2.350	2.310	-1.74	-1.47
1	B	7 7	390	3.523	3.499	-0.68	3.575	3.495	-2.29	1.48
2	B	4 1	2130	2.148	2.189	1.89	2.218	2.200	-0.81	3.26
2	B	5 8	730	2.875	2.879	0.15	2.893	2.869	-0.82	0.63
3	B	5 8	1550	2.440	2.446	0.26	2.468	2.437	-1.27	1.15
4	B	7 7	2380	2.615	2.614	-0.02	2.635	2.604	-1.18	0.76
5	B	8 2	1795	2.178	2.225	2.13				
5	B	8 2	2315				2.318	2.307	-0.46	
5	B	11 1	1460	2.640	2.653	0.47	2.658	2.643	-0.59	-0.68
6	B	11 1	2040	2.255	2.282	1.17	2.295	2.273	-0.96	1.77
6	B	5 8	1190	2.613	2.632	0.74	2.637	2.622	-0.56	0.92
6	B	5 8	1545	2.425	2.448	0.95	2.450	2.439	-0.46	1.03
7	B	7 7	1280	2.767	2.780	0.47	2.787	2.770	-0.61	0.72
7	B	7 7	1690	2.505	2.520	0.59	2.520	2.510	-0.40	0.60
8	B	7 7	0	3.630	3.608	-0.61	3.675	3.605	-1.94	1.24

TABLE 2-1

REV/MODE	RAMP	T.U.R.	INST 174			INST 175			174/175	
			ACT.	CALIB.	DEV.	ACT.	CALIB.	DEV.	DIFF.	
8	8	8 2	0	5.490	5.479	-0.20	5.610	5.539	-1.28	2.19
8	8	4 1	1115	2.587	2.581	-0.23	2.613	2.571	-1.63	1.01
9	8	4 1	3410	3.850	3.849	-0.03	3.900	3.851	-1.28	1.30
9	8	11 1	1090	3.690	3.693	0.07	3.745	3.691	-1.45	1.49
10	8	11 1	2080	2.288	2.297	0.39	2.313	2.288	-1.07	1.09
10	8	11 1	3120	5.170	5.114	-1.10	5.290	5.157	-2.59	2.32
10	8	7 7	305	3.570	3.542	-0.79	3.615	3.538	-2.17	1.26
13	8	4 1	2630	2.363	2.344	-0.80	2.373	2.335	-1.69	0.91
13	8	11 1	1975	2.260	2.268	0.34	2.277	2.259	-0.76	0.75

DEV. AND DIFF. ARE IN PERCENT
 THE (-) SIGN INDICATES THAT THE INST IS SLOWER THAN
 PREDICTED OR THAT INST 1 IS SLOWER THAN INST 2

TABLE 2-1

7. Pressure Make-Up System Performance

The pressure make-up system performed satisfactorily. The gas consumption rate was a little higher than normal at 9.0 psia/minute of operate for the "A" mode. A flow rate check of the system after the test showed the flow rate to be high but within allowable tolerances.

Because of the two chamber dives, the "B" mode chamber pressure did not go below 1 micron.

8. Temperature Summary

The chamber temperatures were cold during the "A" mode and near normal for the "B" mode. Instrument mainplate temperature is shown in Figure 2-1.

B. RESOLUTION TEST

Resolution and theodolite tests were performed on 23 July 1965. Results of the thru-focus resolution tests of pan instruments 174 and 175 show the following characteristics:

Master Pan Instrument No. 174

Maximum high contrast resolution 194 lines/mm at + .001 focal position.

Maximum low contrast resolution 120 lines/mm at + .001 focal position.

Slave Instrument No. 175

Maximum high contrast resolution 194 lines/mm at + .001 focal position.

Maximum low contrast resolution 120 lines/mm at + .001 focal position.

The test data for both instruments is shown in Figures 2-2 and 2-3. Both instruments met the system requirements specification.

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PANORAMIC CAMERA TEMPERATURES vs ORBIT

Temp °F

80

70

60

50

40

30

20

Orbit Number

"A" "B"

○ Avg of T/S 7 & T/S 8
on the Master Instr.

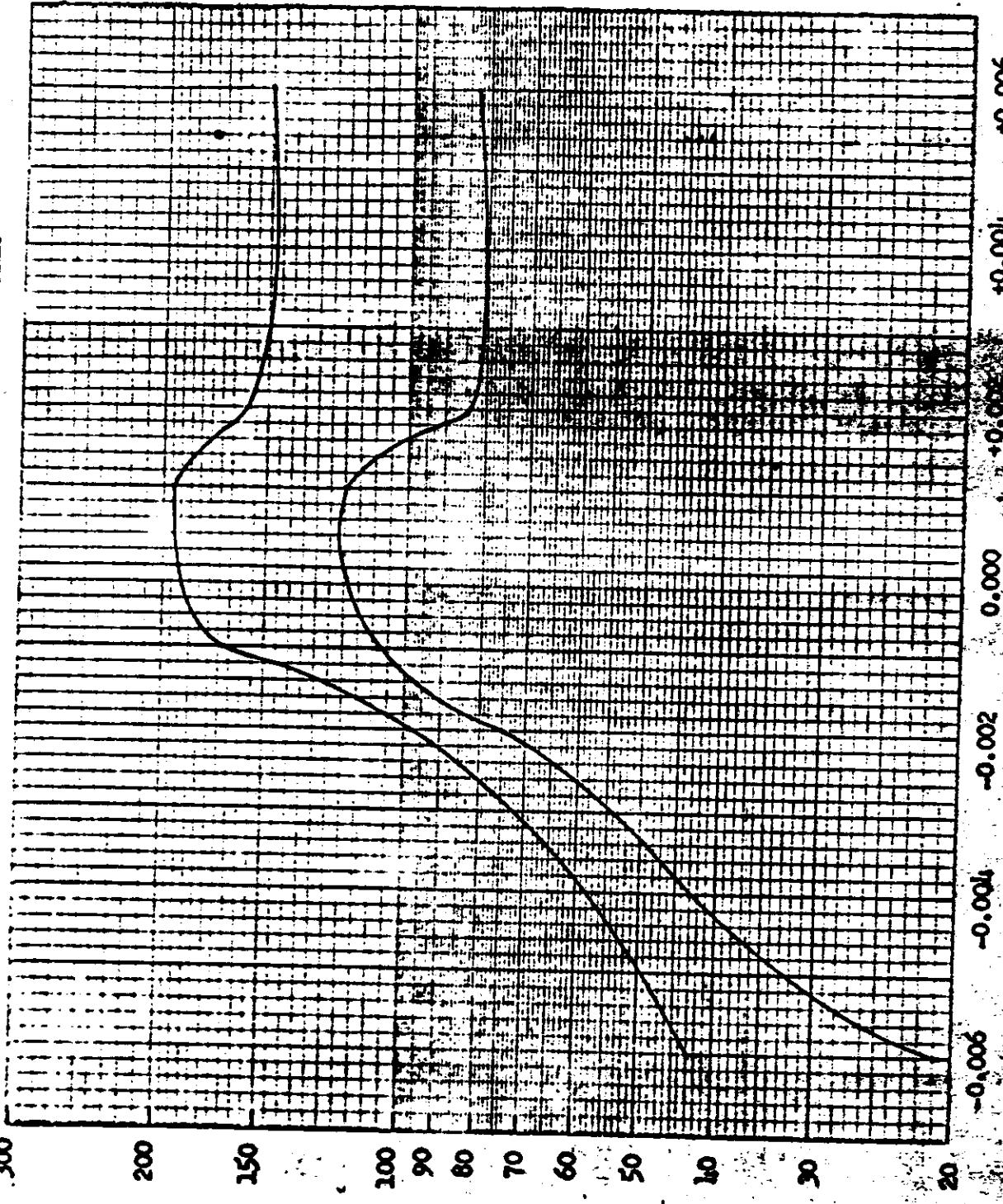
△ Avg of T/S 7 & T/S 8
on the Slave Instr.

8 888
6 666
4 444
2 222

No.

TOP SCREEN

PRE-FLIGHT DYNAMIC RESOLUTION

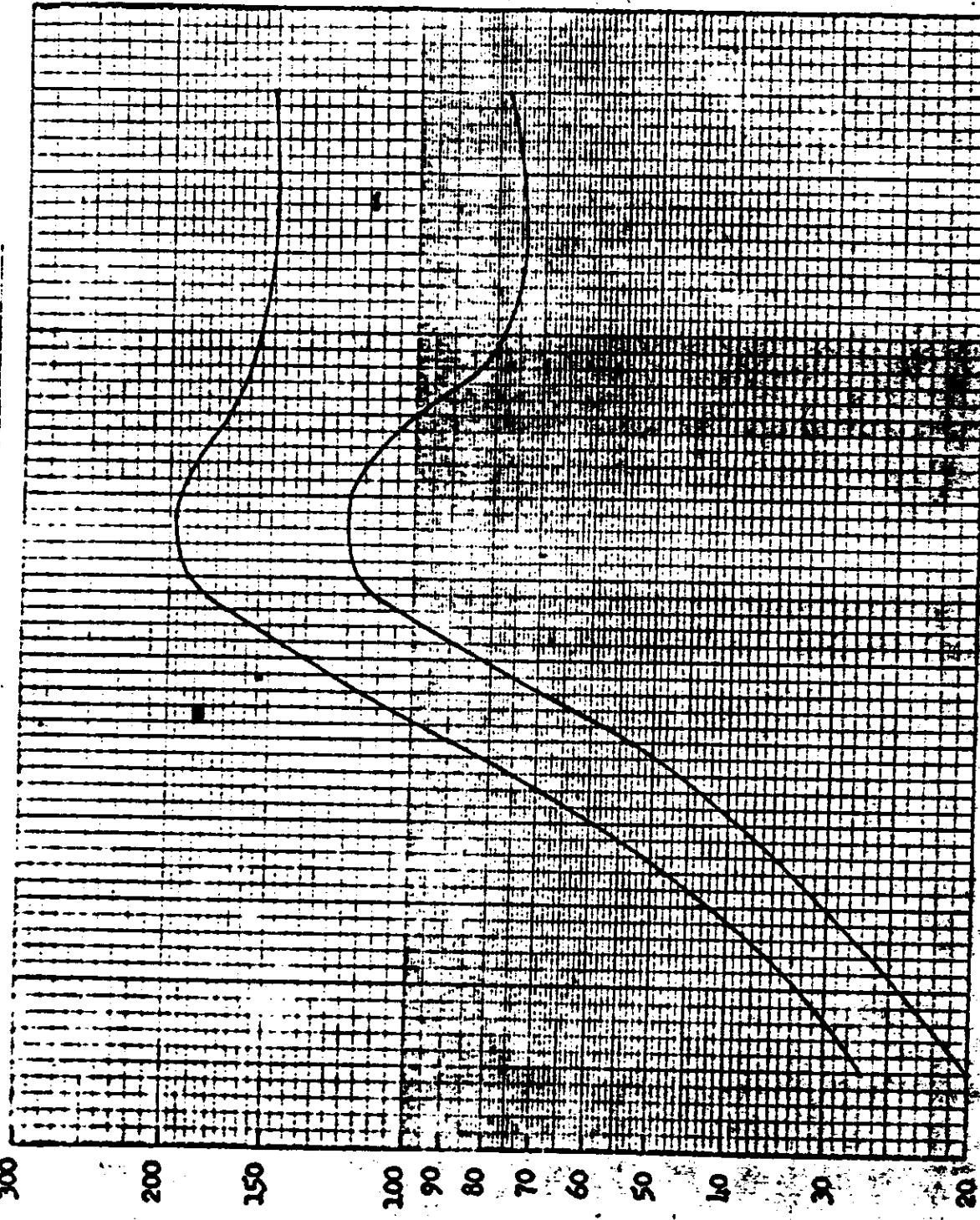


PHOTOGRAPHIC RESOLUTION (Lines per Millimeter)

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No.

PRE-SPLIT DYNAMIC RESOLUTION



PHOTOGRAPHIC RESOLUTION (lines per millimeter)

-0.006 -0.004 -0.002 -0.001 0.000 +0.001 +0.002 +0.004 +0.006 +0.008 +0.009 +0.009

THROUGH FOCUS INCREMENTS (mm)

REO SPECTRUM

PICTURE

C. LIGHT LEAK TEST

The J-25 system was tested for light leaks on 1 September 1965.

A minor light leak fogged three small areas (2 x 1/4 inches each) of the test film in instrument 174 and 175. The first photomultiplier search failed to detect the light leak. The second photomultiplier search located the light leak at the drum/felt seal of the master instrument.

Fog marks measured approximately 0.4 to 0.6 density as a result of the 1/2 hour light leak test and 0.04 density as a result of the 1/4 hour light leak test exposure. The felt seal was inspected to assure a flight acceptable configuration.

D. FLIGHT LOADING AND CERTIFICATION

The J-25 flight readiness test was performed and the resulting pan instrument payload was examined at VAFB on 21 October 1965. The initial readiness test payload showed the master instrument end-of-pass mark to be missing and two minus density streaks due to foreign matter were on the filter. Investigation showed the EOP lamp to be burned out and lint was found on the filter. These items were corrected and the readiness test was re-run. In this test, the condition of the master instrument payload was satisfactory. The slave instrument payload was satisfactory in both tests.

The loading of flight payload into the supply cassette was performed on 25 October 1965. After final system assembly, payload splices and tracking were observed to be normal.

SECTION 3

FLIGHT OPERATIONS

A. INSTRUMENTATION AND COMMAND SYSTEM PERFORMANCE

All instrumentation performed satisfactorily throughout the mission with the exception of the fairing temp sensor No. 1. The telemetry monitor indicated an out of band low condition for the duration of Phase I.

Command System Performance

The command system operation was satisfactory throughout both phases of the mission.

Both emergency operate programs (Programs 10 and 11) were utilized on engineering operations during acquisitions on orbit 167 and 174. All instrument functions operated satisfactory.

The following operational problems were encountered during this mission.

1. A vehicle H-timer reset was issued 100 seconds early on orbit 33 which caused all events on orbit 34 and 35 to occur 100 seconds early.
2. The V/h programmer failed to operate on orbit 88. An RTC 10 command (V/h ramp start delay) was issued on orbit 87 stepping from position 9 to position 8, while brush 27 (V/h programmer start brush) was stepping between position 8 and 9.

B. PANORAMIC CAMERA PERFORMANCE

Both panoramic camera system dynamic operation, 99/101% clutch operation, startup, shutdown, and transport functions were normal for all passes monitored. The slave instrument center of format switch (S-108) failed to close on 37 frames in the -1 mission causing missing data block, horizon optics, blanking pulse and errors in cycle counter readings. The switch closure was observed to fail randomly from D-24 to D-70. There was no reported failure of the switch during the -2 mission.

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No. [REDACTED]

Cycle periods taken from telemetry data during [REDACTED] acquisition are listed in Table 3-1. The cut and wrap sequence performed satisfactorily.

Panoramic Film Consumption - Frames

These data are based on cycle counters and nominal supply length.

	<u>Nominal</u>	<u>Actual</u>	
		<u>Master</u>	<u>Slave</u>
Preload samples	18	17	17
Prelaunch checkout	100	101	100
-1 Mission	2950	2929	2897*
-2 Mission	2981	3002	3055
TOTAL	6049	6049**	6049**

*Slave counter known to be missing counts, therefore unreliable.

**Depletion was predicted near or during orbit 151 descending.

C. FMC MATCH

In general, FMC was accomplished within the desirable 5% tolerance, although on the slow side. However, driving the system well into the cycle rate limiter was necessitated by perigee displacement (4.7 N. M. low, 13° south); this could produce rates nominally as much as 5% slow for orbit match.

The limiter affected operations for a 30 to 35 degree (latitude) range; between 225° - 305° early in the mission, to between 250° to 215° late in the mission.

Certain operations occurred outside the expected tolerance:

Revs 1 thru 6 - Commanding occurred previous to complete orbit definition; only partial compensation made for perigee shift - errors up to 10%.

Revs 34 and 35 - An intolerable reset on Rev 33 caused all functions to occur 100 sec. early; displacing operations northward and yielding approximately 5 - 6% error for actual photography.

Rev 87 - Due to a command problem, the programmer was not started, yielding very high indicated error for the operation run at the "bottom of the ramp".

J-23/1620 MISSION 1026, ENGINEERING OPERATIONS CYCLES RATES

REV/MODE	RAMP	T.U.R.	INST 174			INST 175			174/175 DIFF.
			ACT.	CAL.	DEV.	ACT.	CAL.	DEV.	
009	1	7 3	0	4.760	4.755 0.10S	4.765	4.764 0.39F	0.11	
016	1	7 3	1554	2.340	2.295 1.97S	2.322	2.286 1.57S	-0.77	
031	1	7 3	1625	2.258	2.239 0.85S	2.279	2.240 1.72S	0.93	
040	1	7 3	123	4.722	4.725 0.06F	4.755	4.752 0.06S	0.70	
047	1	7 3	1630	2.265	2.236 1.31S	2.262	2.240 0.98S	-0.13	
063	1	7 3	1659	2.264	2.233 1.39S	2.282	2.238 1.98S	0.80	
079	1	7 2	1695	2.233	2.221 0.92S	2.255	2.228 1.22S	0.99	
88	2	7 2	252	4.888	4.864 0.49S	4.936	4.896 0.81S	0.98	
94	2	7 2	1685	2.240	2.222 0.80S	2.243	2.229 0.74S	-0.27	
94	2	7 2	1825	2.224	2.214 0.45S	2.240	2.222 0.83S	0.62	
103	2	7 2	286	4.845	4.804 0.85S	4.875	4.834 0.84S	0.62	
110	2	7 2	1855	2.240	2.213 1.21S	2.240	2.221 0.86S	-0.00	
126	2	7 2	1856	2.235	2.213 0.98S	2.244	2.221 1.04S	0.40	
142	2	7 2	1850	2.250	2.213 1.66S	2.250	2.221 1.31S	-0.00	
158	2	6 3	1925	2.223	2.210 0.58S	2.225	2.218 0.30S	0.09	

DEV. AND DIFF. ARE IN PERCENT.
 THE (-) SIGN INDICATES THAT INST 1 IS SLOWER THAN INST 2
 F=FAST AND S=SLOW

TABLE 3-1

D. STELLAR/INDEX CAMERA PERFORMANCE

Both the -1 and -2 Stellar/Index cameras operated properly throughout the mission. Metering was normal on all engineering passes and shutter pulses were present on all daytime operations. The -1 index film was not depleted on orbit 79 [REDACTED] but depleted prior to good T/M on cut and wrap slew operation. The -2 index film was depleted at orbit 158 [REDACTED].

E. CLOCK PERFORMANCE

Clock system performance was normal throughout both phases of the mission. Good time correlation between system time and clock time was obtained. A tabulation of the time correlation data is included in Table 3-2.

F. PRESSURE MAKEUP SYSTEM PERFORMANCE

The pressure makeup system performed satisfactorily throughout both phases of the mission. The average gas consumption was 7.4 lbs/min. for a total operate time of 262 minutes, which compares favorably with previous systems.

G. THERMAL ENVIRONMENT

Tables 3-3 and 3-4 contain the temperature data acquired from the [REDACTED] acquisitions. The actual system temperature versus the predicted system temperature for major components is included in Figure 3-1, 3-2 and 3-3.

J-25 MISSION 1026 CLOCK/SYSTEM TIME CORRELATION

SYS TIME I/P	CL TIME I/P	COMP SYS TM	DELTA ST	REV	STA
0.37945138U 05	0.705812700D 04	0.3794513710 05	0.0009	9	
0.77522878U 05	0.4663587000 05	0.7752280000 05	-0.0020	16	
0.33150058U 05	0.8866304300 05	0.3315005290 05	0.0051	24	
0.72799238U 05	0.1283122260 06	0.7279923590 05	0.0021	31	
0.33755785U 05	0.1756687740 06	0.3375574380 05	0.0012	40	
0.73293018U 05	0.2152060080 06	0.7329301770 05	0.0003	47	
0.34130240U 05	0.2624432340 06	0.3413024360 05	-0.0036	56	
0.73844077U 05	0.3021570690 06	0.7384407850 05	-0.0015	63	
0.34853128U 05	0.3495661160 06	0.3485312550 05	0.0025	72	
0.74351678U 05	0.3890646720 06	0.7435168140 05	-0.0034	79	
0.35339748U 05	0.4364527460 06	0.3533975530 05	-0.0073	88	
0.69484758U 05	0.4705977480 06	0.6948475720 05	0.0008	94	
0.30361668U 05	0.5178746570 06	0.3036166620 05	0.0010	103	
0.69910361U 05	0.2055244200 05	0.6991036310 05	-0.0021	110	
0.30771483U 05	0.6781355700 05	0.3077147800 05	0.0050	119	
0.70302455U 05	0.1073445370 06	0.7030245790 05	-0.0029	126	
0.31036218U 05	0.1544782980 06	0.3103621880 05	-0.0008	135	
0.70545878U 05	0.1939879590 06	0.7054587980 05	-0.0018	142	
0.31443048U 05	0.2412851280 06	0.3144304870 05	-0.0007	151	
0.70861774U 05	0.2807038470 06	0.7086176760 05	0.0064	158	

A0= 0.30887010100 05 A1= 0.999999981740 00

SIGMA=0.00317 NO. POINTS= 20

RATIO OF CLOCK TIME TO SYS TIME= 0.1000000001030 01

SYS TIME I/P	CL TIME I/P	COMP SYS TM	DELTA ST	REV	STA
0.37945138U 05	0.705812700D 04	0.3794513930 05	-0.0013	9	
0.77522878U 05	0.4663587000 05	0.775228160 05	-0.0036	16	
0.33150058U 05	0.8866304300 05	0.3315005390 05	0.0041	24	
0.72799238U 05	0.1283122260 06	0.7279923630 05	0.0017	31	
0.33755785U 05	0.1756687740 06	0.3375578370 05	0.0013	40	
0.73293018U 05	0.2152060080 06	0.7329301720 05	0.0008	47	
0.34130240U 05	0.2624432340 06	0.3413024280 05	-0.0028	56	
0.73844077U 05	0.3021570690 06	0.7384407750 05	-0.0005	63	
0.34853128U 05	0.3495661160 06	0.3485312420 05	0.0038	72	
0.74351678U 05	0.3890646720 06	0.7435168010 05	-0.0021	79	
0.35339748U 05	0.4364527460 06	0.3533975400 05	-0.0060	88	
0.69484758U 05	0.4705977480 06	0.6948475600 05	0.0020	94	
0.30361668U 05	0.5178746570 06	0.3036166510 05	0.0029	103	
0.69910361U 05	0.2055244200 05	0.6991036220 05	-0.0012	110	
0.30771483U 05	0.6781355700 05	0.3077147750 05	0.0055	119	
0.70302455U 05	0.1073445370 06	0.7030245780 05	-0.0028	126	
0.31036218U 05	0.1544782980 06	0.3103621920 05	-0.0012	135	
0.70545878U 05	0.1939879590 06	0.7054588070 05	-0.0027	142	
0.31443048U 05	0.2412851290 06	0.3144305030 05	-0.0023	151	
0.70861774U 05	0.2807038470 06	0.7086176990 05	0.0041	158	

A0= 0.30887012460 05 A1= 0.9999999802430 00

A2= 0.21800048741270-13

SIGMA=0.00295 NO. POINTS= 20

TABLE 3-2

TABLE 3-3
J-25 TEMPERATURE SUMMARY

<u>SENSOR</u>	<u>ORBITS ACQUIRED</u>																		
	9	16	24	31	40	47	56	63	72	79	88	94	103	110	119	126	135	142	151
<u>Master Camera</u>	78	75	79	76	79	76	77	74	76	74	71	67	68	64	66	64	66	64	67
3	85	81	85	80	85	81	83	80	81	79	79	73	75	69	73	69	73	69	73
4	91	87	91	88	90	87	89	86	87	84	83	78	78	75	77	74	75	74	76
5	92	88	91	89	90	87	88	85	86	84	80	77	76	73	73	72	73	71	71
6	90	87	90	88	89	86	87	85	85	83	80	77	76	74	74	72	73	72	72
7	93	88	93	89	92	87	91	87	89	85	84	80	81	77	79	76	79	75	78
8	95	91	95	91	94	89	92	87	90	85	85	80	81	77	80	75	79	74	78
9	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
10	92	89	93	86	90	85	89	84	86	83	73	74	75	71	73	69	72	68	70
11	79	74	79	75	80	75	79	73	77	74	73	67	70	65	69	65	70	65	70
12	85	83	85	82	83	81	82	79	79	78	71	69	69	68	67	66	65	65	66
13	88	84	88	85	87	83	86	82	84	81	78	74	72	71	73	70	73	71	73
AVG	88	84	88	85	87	83	86	82	84	81	78	74	72	71	73	70	73	71	73
<u>Slave Camera</u>	85	83	85	84	84	82	83	78	80	76	74	68	68	64	65	62	64	61	62
3	85	79	84	80	84	80	83	77	82	76	77	69	71	66	69	64	68	63	66
4	85	82	86	83	86	82	85	82	83	81	80	74	75	71	73	71	72	71	71
5	80	78	81	80	81	79	80	79	78	78	74	71	70	68	68	67	69	67	
6	84	82	84	83	84	82	83	80	81	79	76	72	72	70	70	69	70	68	
7	82	78	83	80	83	79	82	79	81	78	77	71	72	68	71	67	70	67	
8	78	77	81	78	80	78	81	77	79	77	76	71	73	69	71	68	71	69	
9	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-	
10	74	72	74	77	76	73	76	73	73	72	71	64	63	61	62	60	63	60	62
11	88	81	89	83	87	81	87	81	84	79	79	72	75	69	72	67	71	66	69
12	75	73	77	77	76	75	75	74	73	73	69	64	64	63	62	61	62	61	
13	82	78	82	80	82	79	81	76	79	77	75	70	71	67	68	66	68	66	
AVG	82	78	82	80	82	79	81	76	79	77	75	70	71	67	68	66	68	66	67
<u>Supply Spool</u>	66	68	70	70	72	72	72	72	72	72	71	68	65	64	62	62	61	62	61
1	87	75	79	77	81	77	81	77	79	75	75	69	69	67	68	65	67	65	66

NOTE: All data corrected for self-heating, except injection.

TABLE 3-4
J-25 TEMPERATURE SUMMARY

		<u>ORBITS ACQUIRED</u>																			
<u>Fair ("A")</u>																					
<u>Barrel #1 ("B")</u>		9	16	16	31	40	47	56	63	72	78	88	94	103	110	119	126	135	142	15	
1		-	-	-	-	-	-	-	-	-	-	82	85	82	82	85	85	82	85	8	
2		-5	-12	-2	-15	-5	-12	-2	-15	-2	-17	56	44	60	44	60	48	64	48	6	
3		-13	-13	-13	-11	-13	-13	-11	-13	-13	-13	79	96	76	93	79	93	76	82	7	
4		-	OBL	OBL	OBL	-	OBL	OBL	OBL	-	-	51	67	46	59	43	54	43	46	3	
5		71	68	68	60	60	52	68	48	56	44	40	47	33	37	33	33	26	30	1	
6		64	79	64	76	56	64	64	56	52	45	-	-	-	-	-	-	-	-		
<u>Barrel #2</u>																					
1		58	73	54	65	46	65	50	58	42	50	30	34	26	26	23	23	15	15	1	
2		61	92	61	92	58	80	61	75	55	69	49	66	46	61	46	52	41	43	3	
3		69	88	69	96	69	88	73	88	73	86	69	91	69	86	69	88	69	88	3	
4		62	53	66	53	66	56	72	59	69	56	66	56	66	53	69	59	72	59	3	
5		52	56	60	60	52	64	64	64	64	64	41	41	37	37	37	41	37	41	3	
<u>Conic Adapter</u>																					
1		57	62	54	57	51	51	54	46	49	43	35	30	33	25	30	22	25	22	2	
<u>Clock</u>																					
1		68	64	70	64	68	66	68	64	68	64	57	51	51	51	51	55	49	51	49	5
2		81	77	83	79	70	79	83	79	81	77	70	66	68	62	68	62	66	62	6	
<u>Thrust Cone "A" to "B" SRV</u>																					
1		49	42	45	41	44	40	44	39	43	39	65	63	63	60	60	59	59	59	5	
2		69	61	64	58	61	56	59	55	57	52	74	70	69	67	67	65	65	64	6	
<u>Stellar/Index "A" to "B"</u>																					
1		67	60	63	57	57	57	60	54	57	54	54	51	51	47	51	47	44	47	4	
2		62	59	59	55	55	55	55	52	52	49	59	55	55	47	51	47	47	47	4	
<u>Recovery Battery "B" SRV</u>																					
1		75	75	75	75	74	75	74	74	73	72	80	83	81	79	83	78	80	78	7	
<u>Master Cassette "A" SRV</u>																					
2		79	76	77	77	77	75	75	75	75	75	76	-	-	-	-	-	-	-		

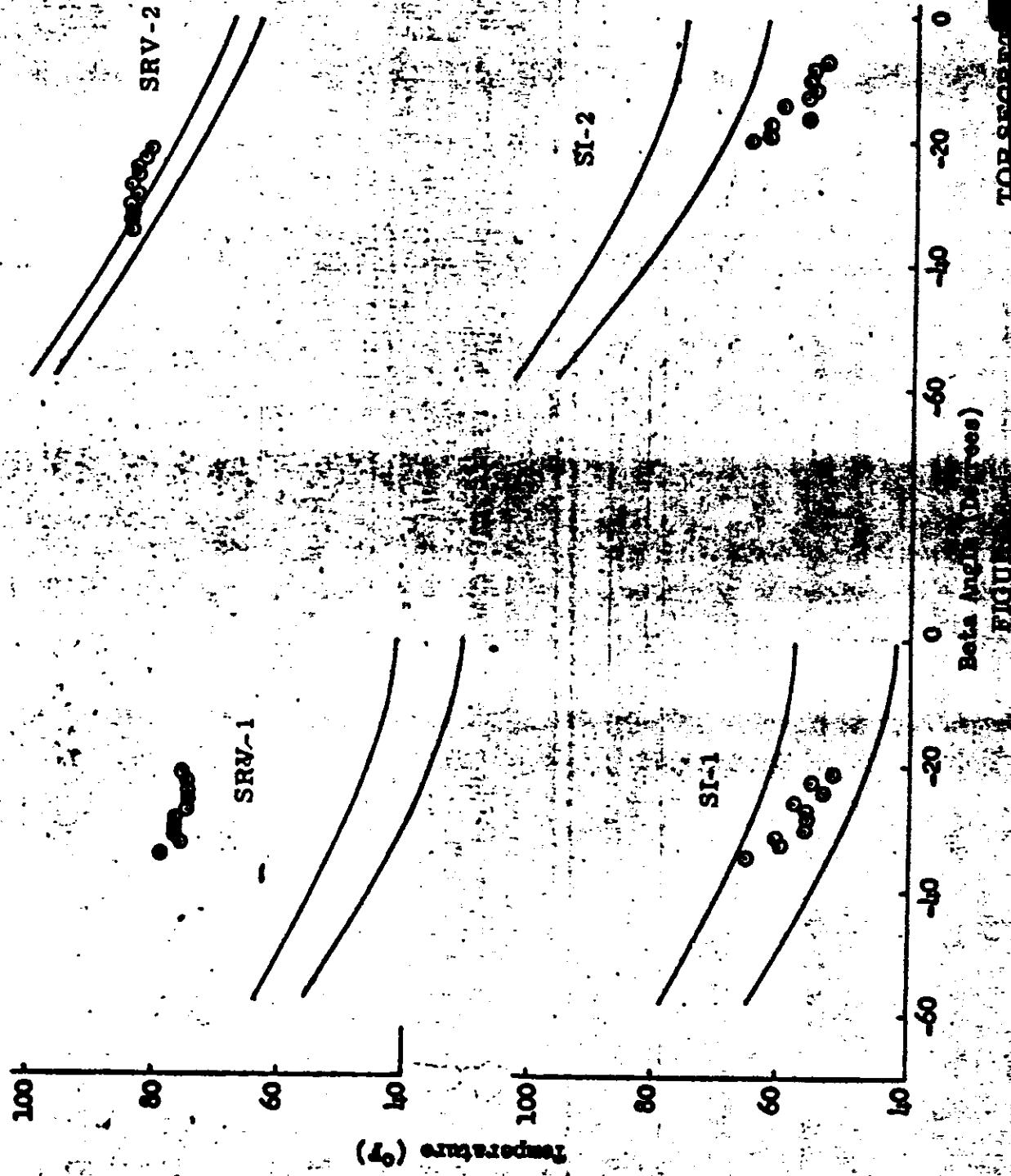
NOTE: Only Thrust Cone Data corrected for Self-heating.

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TOP SECRET

No. [REDACTED]

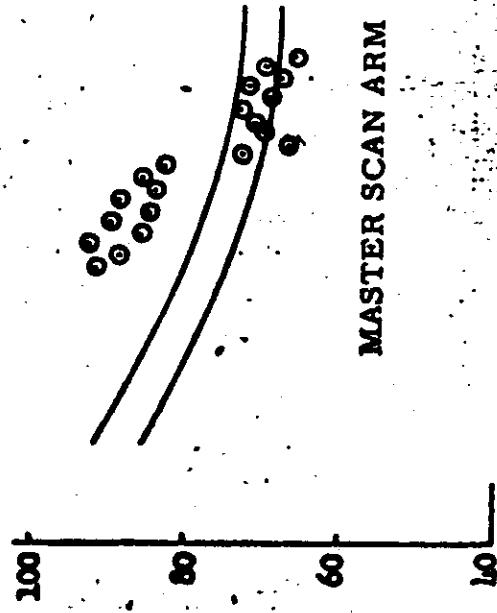
J-25 FLIGHT
10/28/65 - 11/7/65



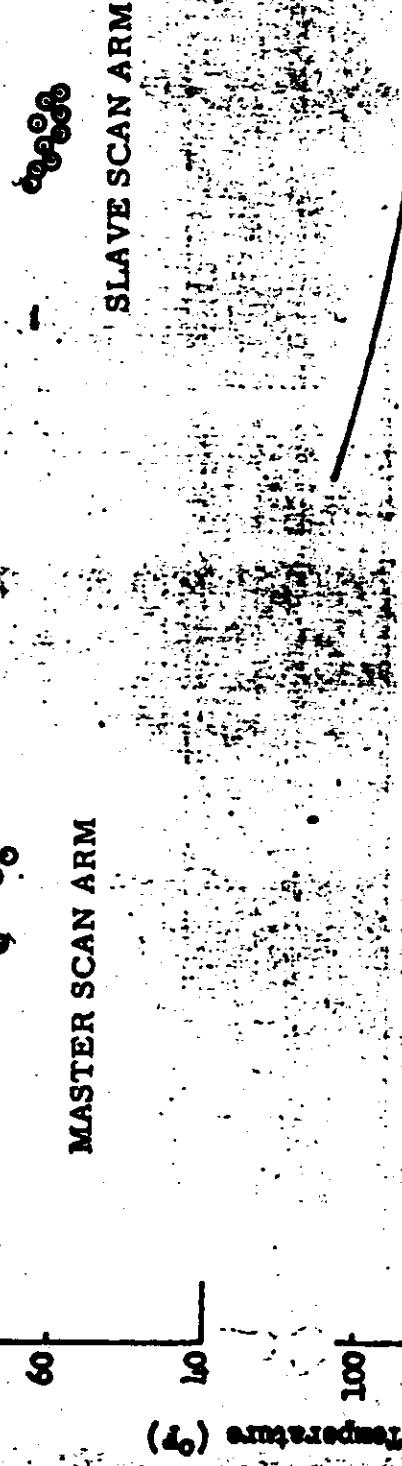
TOP SECRET

No.

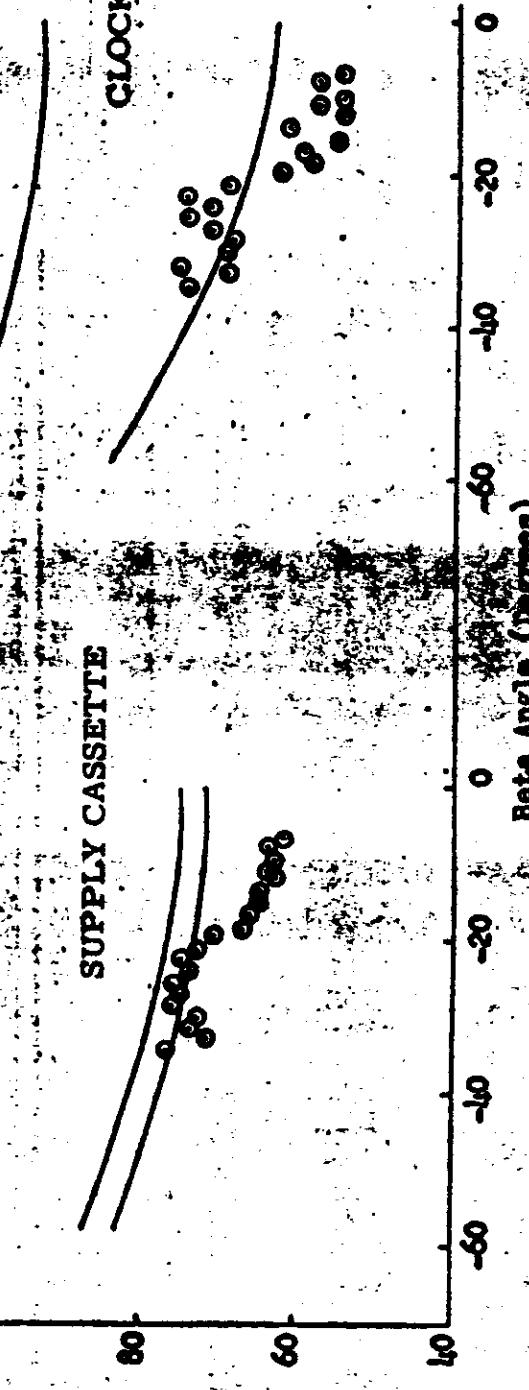
J-25 FLIGHT
10/28/65 - 11/7/65



SLAVE SCAN ARM



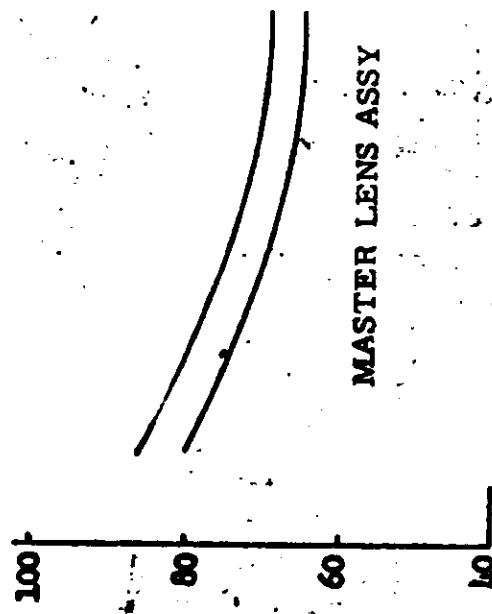
CLOCK
SUPPLY CASSETTE



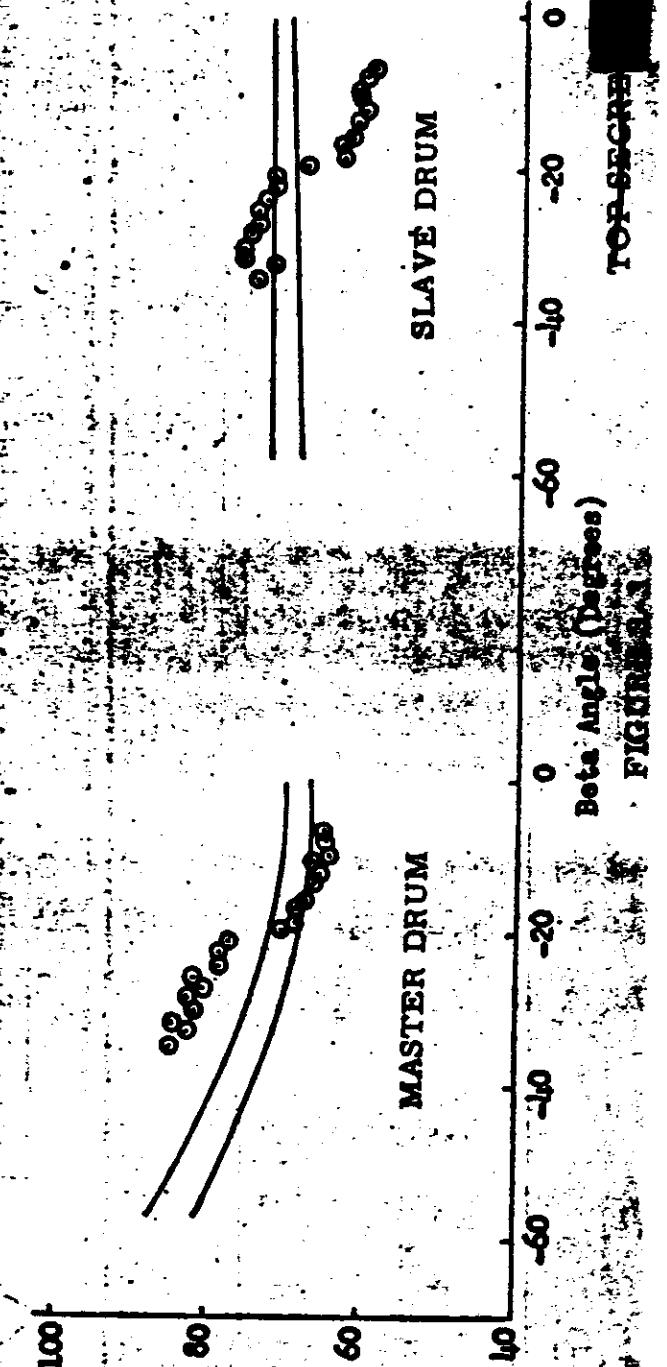
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No

J-25 FLIGHT
10/28/65 - 11/7/65



(%) measured



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FIGURE 3

SECTION 4**MISSION 1026-1 RECOVERY SYSTEM**

SRV #701 was received at A/P on 7 May 1965. The receiving weight was 149.7 pounds. After modifications and incorporation of outstanding E.O.'s, the SRV was delivered to Systems Test for incorporation into the J-25 system.

The capsule was shipped to VAFB on 1 October 1965.

The -1 recovery system was successfully recovered by air catch from orbit 81 at 16:03 PST on 2 November 1965. The impact point was as follows:

Predicted Impact $24^{\circ}01.5'N/168^{\circ}48'W$

Actual Impact $23^{\circ}53'N/168^{\circ}37'W$

The condition of the recovered capsule was satisfactory with no damage other than normal paint blistering due to the re-entry environment.

Event times are shown in Table 4-1. There were no chute events recorded at orbit 81 [REDACTED] or T/M ship.

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No. [REDACTED]

MISSION 1026-1

RECOVERY SEQUENCE OF EVENTS

<u>Event</u>	<u>Delta Time (Seconds)</u>		
	<u>Actual</u>	<u>Nominal</u>	<u>Events + △ T</u>
*Arm	76.99	77.0	+ 1.0
*Transfer	2.3	2.0	+ 0.25
Electrical Disconnect	1.0	0.900	+ 0.430 - 0.400
Separation	---	---	---
**Spin	3.43	3.4	+ 0.30
Retro	7.53	7.55	+ 0.45
Despin	10.70	10.75	+ 0.59
T/C Separation	1.50	1.5	+ 0.15
***"G" Switch Open	N/A	469.7	
Parachute Cover Off	N/A	34.0	+ 1.5
Drogue Chute Deployed	N/A	0.63	+ 0.08
Main Chute Bag Separate	N/A	10.25	+ 1.5
Main Chute Deployed	N/A	0.52	+ 0.13
Main Chute Disreef	N/A	4.5	+ 0.80

* From Separation

** From Electrical Disconnect

*** From Retro

TABLE 4-1

~~TOP SECRET~~

SECTION 5

MISSION 1026-2 RECOVERY SYSTEM

SRV #702 was received at A/P on 7 May 1965. The receiving weight was 147.8 pounds. After modifications and incorporation of outstanding E.O.'s the unit was delivered to Systems Test for mating to the J-25 system.

The capsule was shipped to VAFB on 1 October 1965.

The -2 recovery system was successfully recovered by air catch from orbit 160 at 15:04 PST on 7 November 1965. The impact point was as follows:

Predicted Impact $24^{\circ}02'N/189^{\circ}49'W$

Actual Impact $23^{\circ}53'N/189^{\circ}42'W$

Event times are shown in Table 5-1.

The condition of the recovered capsule indicated no abnormal re-entry effects.

MISSION 1026-2

RECOVERY SEQUENCE OF EVENTS

<u>Event</u>	<u>Delta Time (Seconds)</u>		
	<u>Actual</u>	<u>Nominal</u>	<u>Events + △ T</u>
* Arm	76.84	77.0 <u>±</u> 1.0	
* Transfer	2.03	2.0 <u>±</u> 0.25	
Electrical Disconnect	0.89	0.900 ^{+0.430} _{-0.400}	
Separation	---	----	
** Spin	3.39	3.4 <u>±</u> 0.30	
Retro	7.59	7.55 <u>±</u> 0.45	
Despin	10.71	10.75 <u>±</u> 0.50	
T/C Separation	1.51	1.5 <u>±</u> 0.15	
*** "G" Switch Open	469.07	469.7	
Parachute Cover Off	33.88	34.0 <u>±</u> 1.5	
Drogue Chute Deployed	0.59	0.63 <u>±</u> 0.08	
Main Chute Bay Separate	11.05	10.25 <u>±</u> 1.5	
Main Chute Deployed	0.55	0.52 <u>±</u> 0.13	
Main Chute Disreef	4.45	4.45 <u>±</u> 0.80	

* From Separation
 ** From Electrical Disconnect
 *** From Retro

TABLE 5-1

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No.

SECTION 6

MASTER PANORAMIC CAMERA

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Main Camera	174
Main Camera Lens	1712435
Supply Horizon Camera	276-G2
Supply Horizon Camera Lens	816646
Take-up Horizon Camera	276-G1
Take-up Horizon Camera Lens	816625
Supply Cassette	SC-35

B. CAMERA DATA AND FLIGHT SETTINGS

Main Camera:

Lens	24" f/3.5
Slit Width	0.225"
Filter Type	Wratten 25
Film Type	Eastman Type 4404

Supply (Port) Horizon Camera:

Lens	55 mm f/6.8
Aperture Setting	f/6.8
Exposure Time	1/100 second
Filter Type	Wratten 25

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Take-up (Starboard) Horizon Camera:

Lens	55 mm f/6.8
Aperture Setting	f/8.0
Exposure Time	1/100 second
Filter Type	Wratten 25

C. POST FLIGHT PERFORMANCE EVALUATION (Master Instrument)

The image quality of the photography produced by the Master camera was rated good but not equal to the quality of the previous two missions (1024 and 1025). Mission 1026-1 and 1026-2 were extensively affected by atmospheric conditions in combination with low solar elevations. Cloud cover ran up to 70% over some primary target areas and averaged approximately 50% for Mission 1026-1 and 1026-2.

The electro-mechanical operation of the camera system was good during both missions.

Light leaks fogged part of the first and third frames of most passes and the sixth frame from the end of most passes. Light leak fog was generally low density in the original negative. Light leak fog had a minor degrading effect on terrain imagery.

Fog on the first and third frames is attributed to a minor light leak in the vicinity of the master camera drum area. Fog on the sixth frame from the end of pass is attributed to a minor light leak in the vicinity of SRV #1. The main camera drum light leak is expected to be suppressed by application of a silicon elastomer on the suspect light leak area. Opaque film chutes will be added to eliminate the light leak fog occurring at the SRV #1 fairing interface area beginning approximately J-30 and up.

Auxiliary data recording such as the 200 pps track, blanked pulse, S/I slur pulse, instrument serial number, and start of pass mark were consistently good throughout Mission 1026-1 and 1026-2.

Horizon fiducials were consistently good. The imagery of the take-up horizon camera was veiled on passes D-03 through D-14. The cause of veiled horizon imagery is under investigation. Take-up horizon imagery was extremely overexposed in most passes after D-98, however, horizon arcs appear usable. Take-up horizon light fog is attributed to excessive light from high reflectance scenes on the sun side of the vehicle.

Traces of dendritic static are present intermittently along both edges of the film in passes D-40, D-70, D-72, D-78, D-79, D-84, D-85, D-86, and D-116 and along the binary edge in passes D-05 and D-51.

Minor scratches are present under the camera serial number, along the edge opposite the camera number and at the take-up end of each frame. Minor scratches are attributed to normal scan head roller action.

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No.

SECTION 7

SLAVE PANORAMIC CAMERA

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Main Camera	175
Main Camera Lens	1722435
Supply Horizon Camera	277-G2
Supply Horizon Camera Lens	816626
Take-up Horizon Camera	277-G1
Take-up Horizon Camera Lens	816631
Supply Cassette	SC-55

B. CAMERA DATA AND FLIGHT SETTINGS

Main Camera:

Lens	24" f/3.5
Slit Width	0.150
Filter Type	Wratten 21
Film Type	Eastman Type 4404

Supply (Starboard) Horizon Camera:

Lens	55 mm f/6.8
Aperture Setting	f/8.0
Exposure Time	1/100 second
Filter Type	Wratten 25

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~~TOP SECRET~~

No.

Take-up (Port) Horizon Camera

Lens	55 mm f/6.8
Aperture Setting	f/6.8
Exposure Time	1/100 second
Filter Type	Wratten 25

C. POST FLIGHT PERFORMANCE EVALUATION (Slave Instrument)

The photographic quality and information content of the imagery produced by the slave camera was rated slightly higher on the average than the master camera for Missions 1026-1 and 1026-2.

The binary index lamp, binary work, and camera number were absent on 57 frames and the horizon fiducials were absent on 27 frames of Mission 1026-1. Failure occurred randomly throughout Mission 1026-1 beginning with pass D-24 and ending with pass D-70. Failure of the subject auxiliary data recording is attributed to a faulty center of format switch. Center of format switch operation was acceptable throughout Mission 1026-2 as evidenced by the presence of good data recording on all frames of recorded material.

The horizon cameras were operational throughout Mission 1026-1 and 1026-2. However, the imagery recorded by the supply horizon camera of the slave instrument appeared out of focus or veiled throughout Mission 1026-1. This condition continues in Mission 1026-2 but gradually disappears. After pass D-110 the supply horizon photography is sharp. The cause of this phenomenon, although unknown at this time, is under investigation.

Minor light leak fog marks were observed on the first frame and the last three frames of most passes. Fogging is attributed to minor light leaks in the vicinity of the slave instrument drum area.

Minor dendritic static was present at random intervals along both edges of the slave camera film in several passes of photography.

Fine scratches were present under the camera number and at the take-up end of each frame. Scratches are attributed to normal scan head roller action.

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SECTION 8

PANORAMIC CAMERA EXPOSURE

The Master camera contained a 0.225 inch slit and a Wratten 25 filter. The Slave camera had a 0.150 inch slit and a Wratten 21 filter. These conditions placed the nominal exposure on the full processing curve.

The frequency distributions of the solar elevations and solar azimuths encountered during the photographic operations are shown in Figures 8-1 to 8-4.

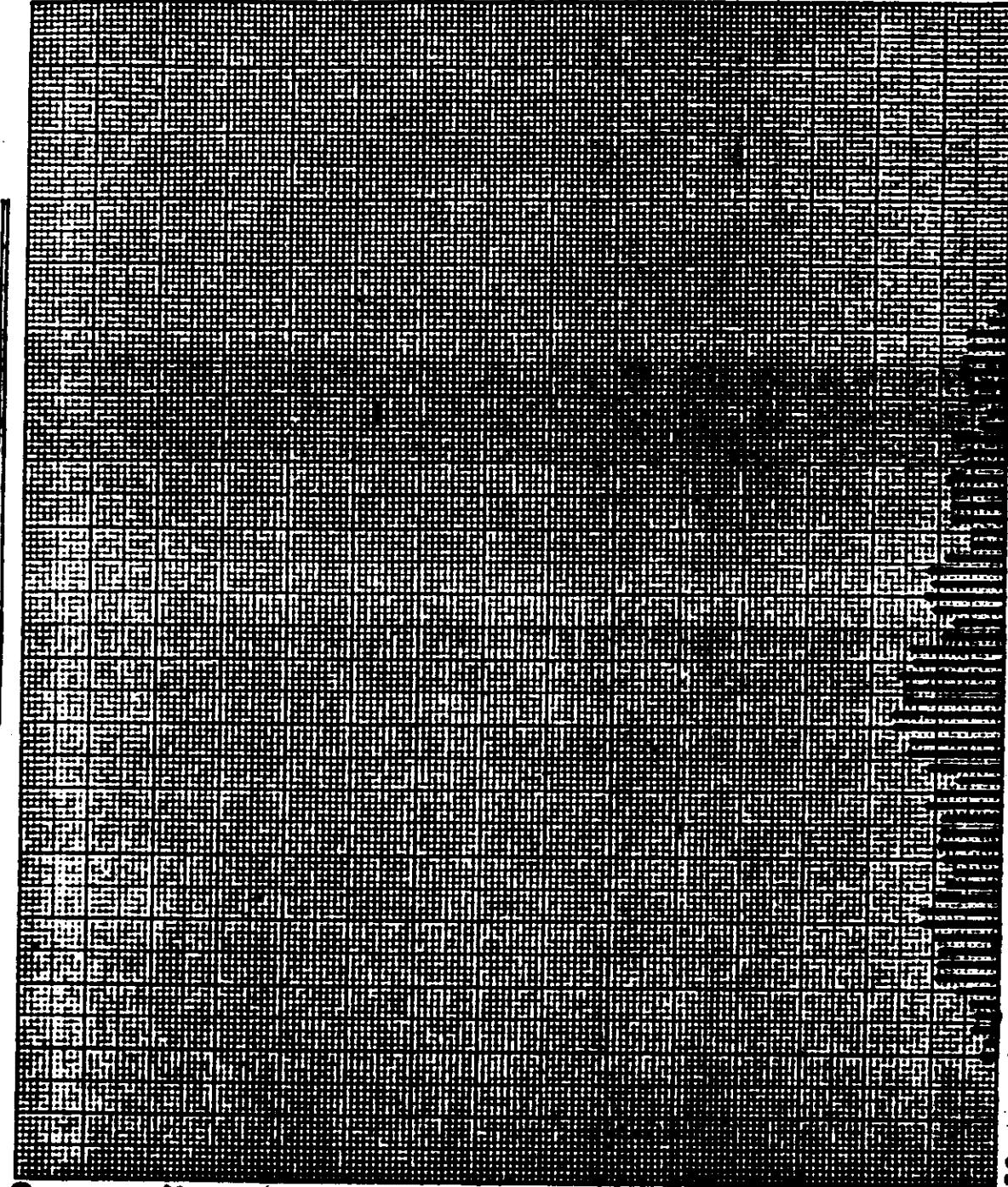
The nominal exposure times of the Master and Slave cameras are shown as a function of latitude for passes D-8, D-58, D-104 and D-152 in Figures 8-5 to 8-12. The predicted level of processing for the original negative is based on the in-flight performance estimate and is tabulated below with the processing levels reported by [REDACTED]

Mission	Camera		Primary	Intermediate	Full
1026-1	FWD	Predicted	0	4	96
		Reported	0	21	79
1026-1	AFT	Predicted	0	5	95
		Reported	0	4	96
1026-2	FWD	Predicted	1	0	99
		Reported	2	5	93
1026-2	AFT	Predicted	1	0	99
		Reported	2	5	93

On pass D-87 the V/h programmer did not operate due to a command error. The imagery was taken at a slow exposure and primary processing was predicted; [REDACTED] reported Full processing.

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No. [REDACTED]
SOLAR ELEVATION FREQUENCY DISTRIBUTION



30

20

-20

-10

10

20

30

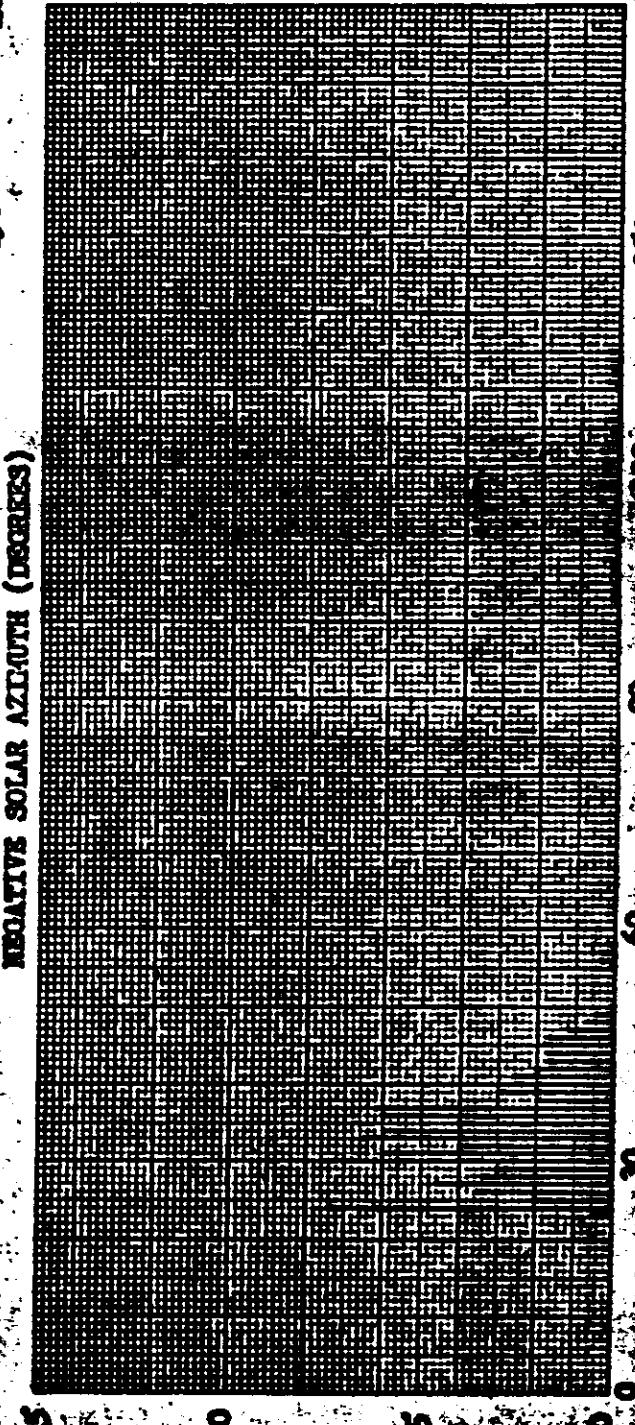
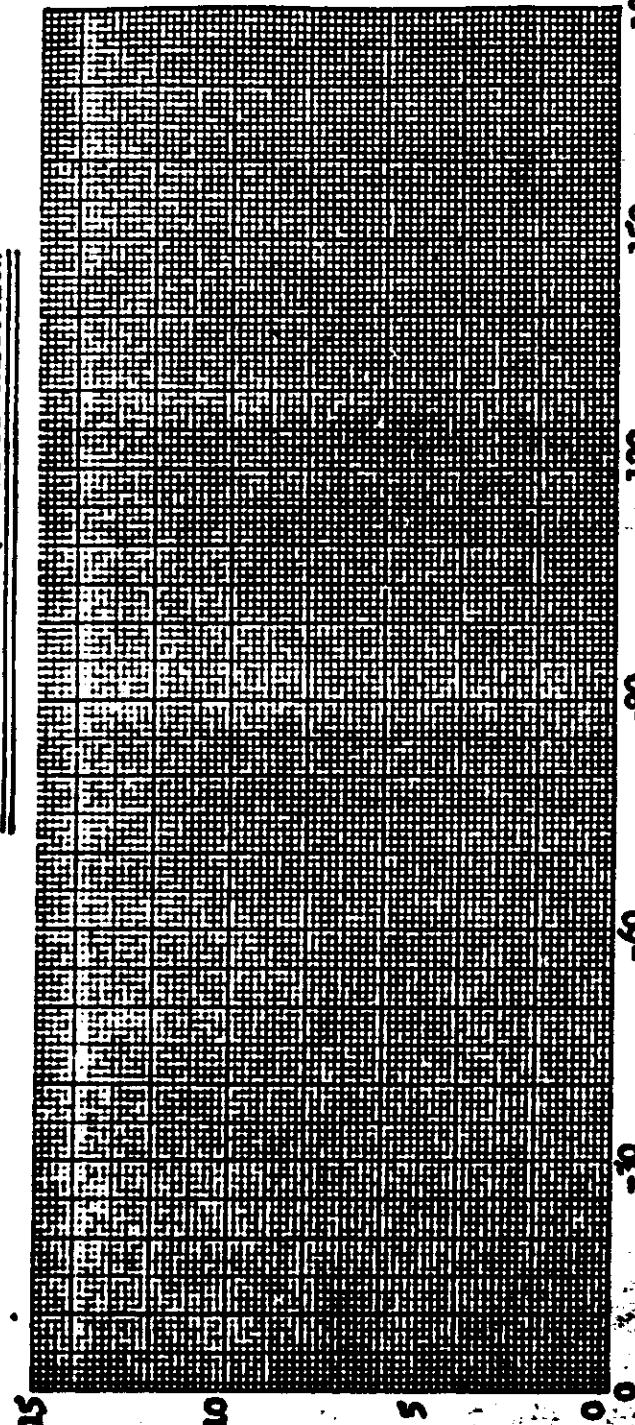
PERCENT (PERCENT)

~~TOP SECRET~~
FIGURE
[REDACTED]

~~TOP SECRET~~

No.

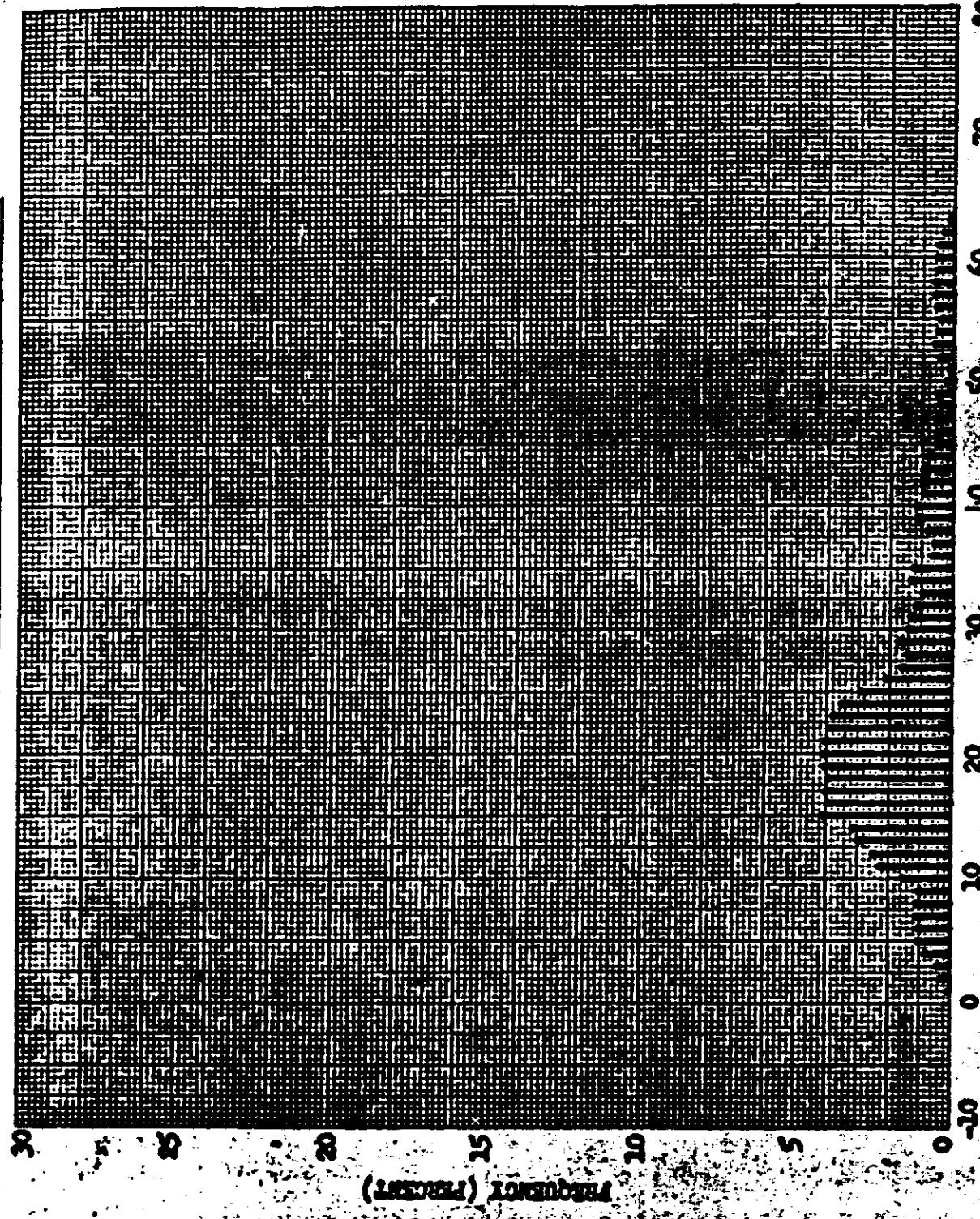
SOLAR AZIMUTH FREQUENCY DISTRIBUTION



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Transmitter No. [REDACTED]

SOLAR ELEVATION FREQUENCY DISTRIBUTION

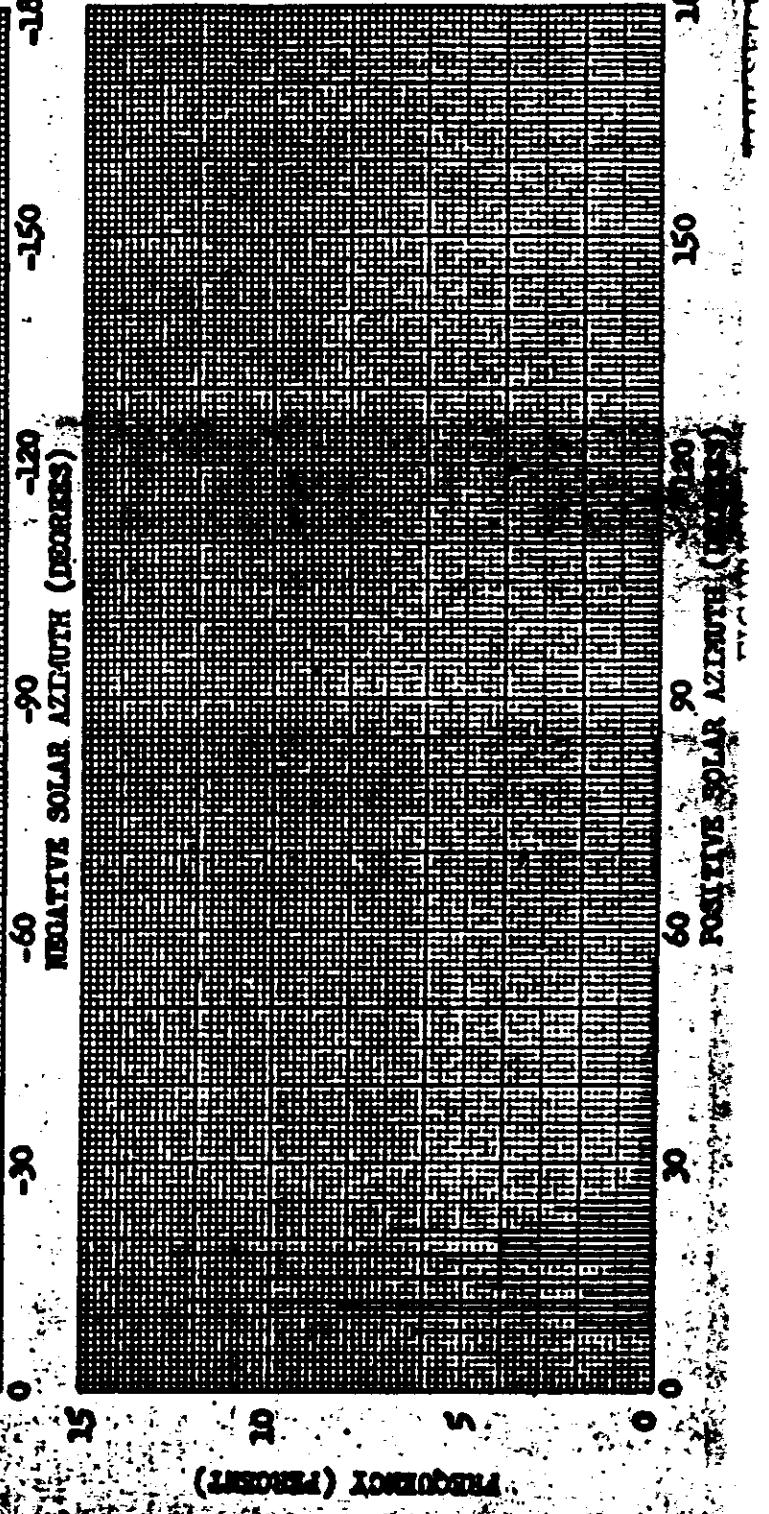
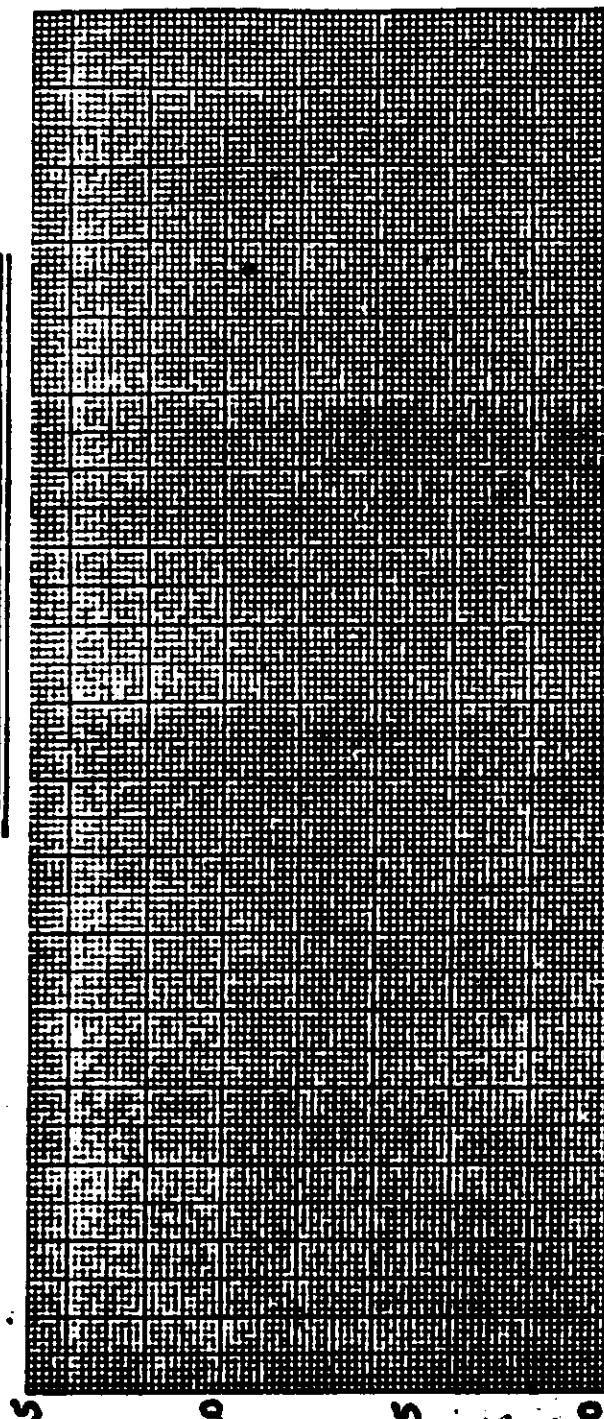


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TOP SECRET

No. [REDACTED]

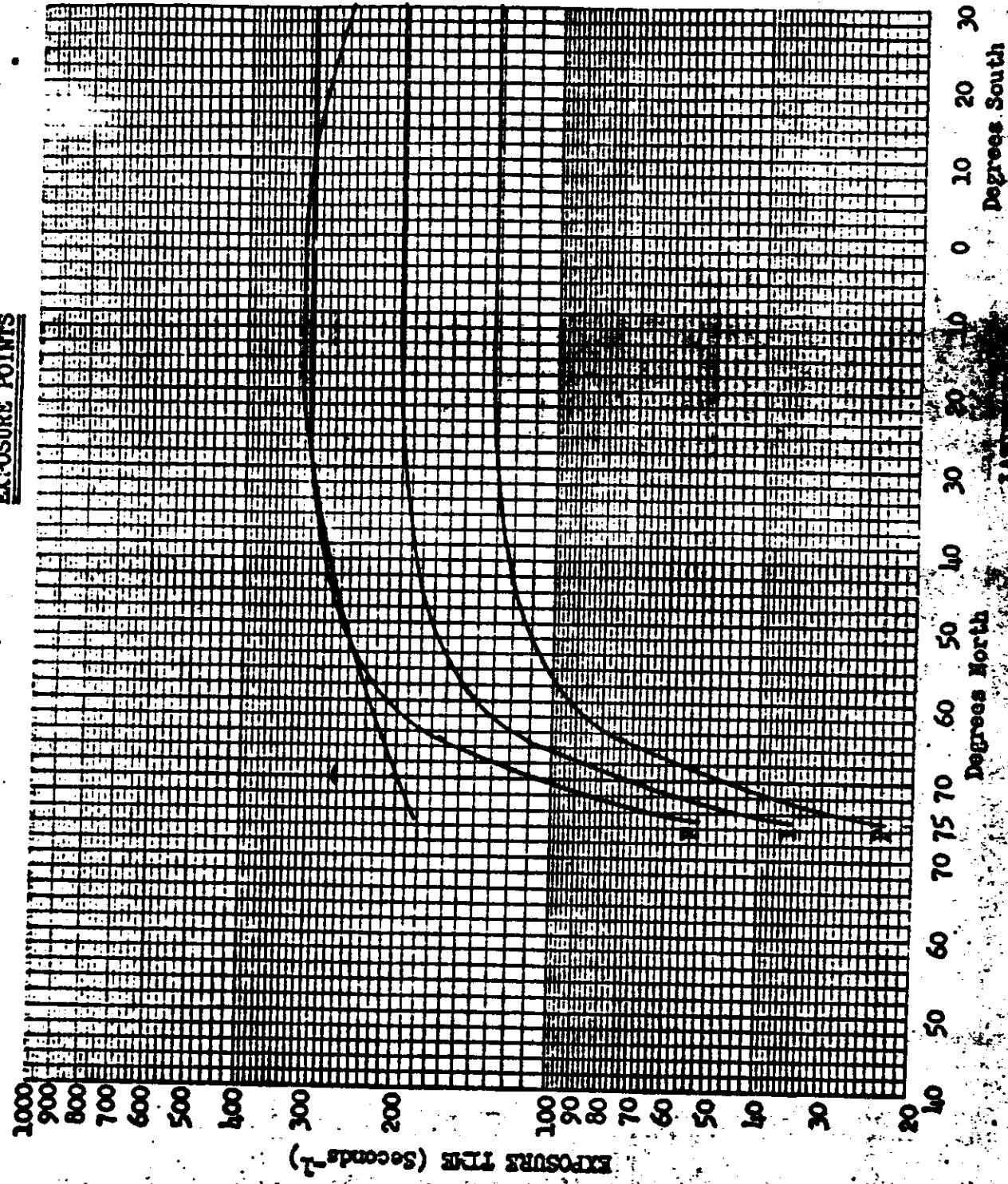
SOLAR AZIMUTH FREQUENCY DISTRIBUTION



RESC-SECRET

No. [REDACTED]

EXPOSURE POINTS



Degrees South

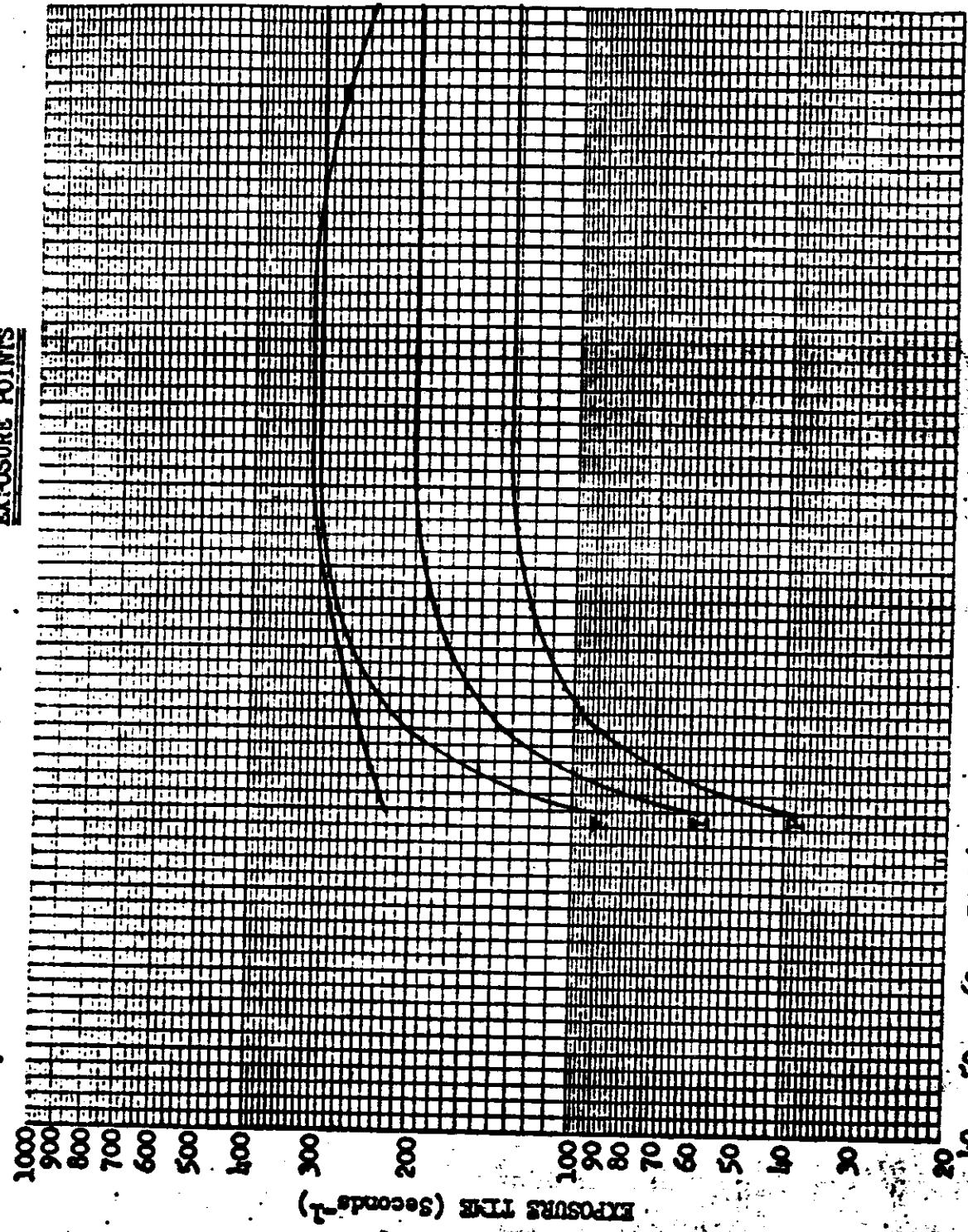
30 20 10 0 -10 -20 -30

Degrees North

-20 -10 0 10 20 30

LIMITED

EXPOSURE POINTS



No. [REDACTED] [REDACTED]

Mission No: 1026

Payload No: 125

Camera No: 124

Pass No: 56

Launch Date: 10/28/65

Launch Time: 2117 Z

Sat: Width: • 22

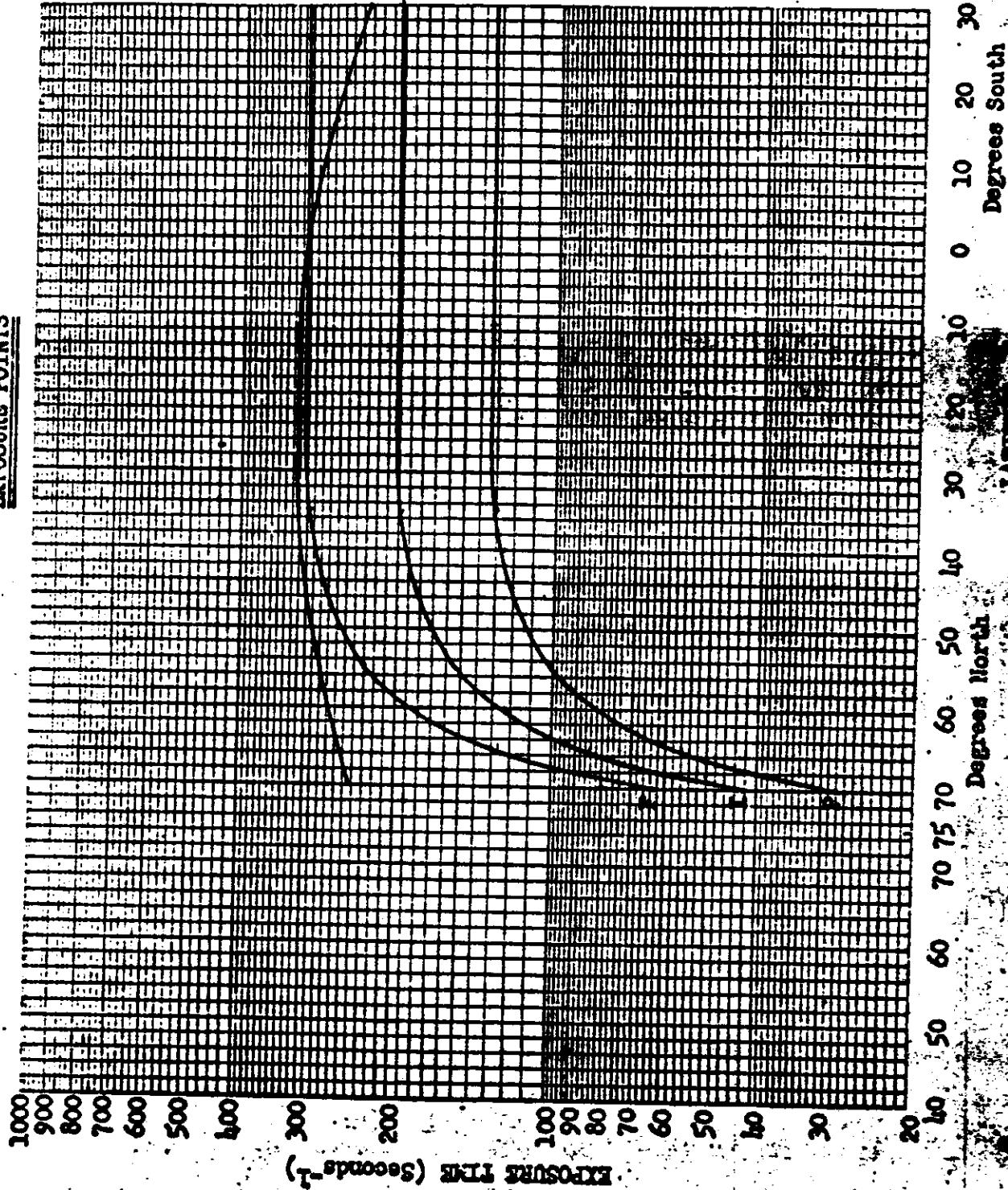
Filter Type: Watten

Film type: 3500

TOP SECRET

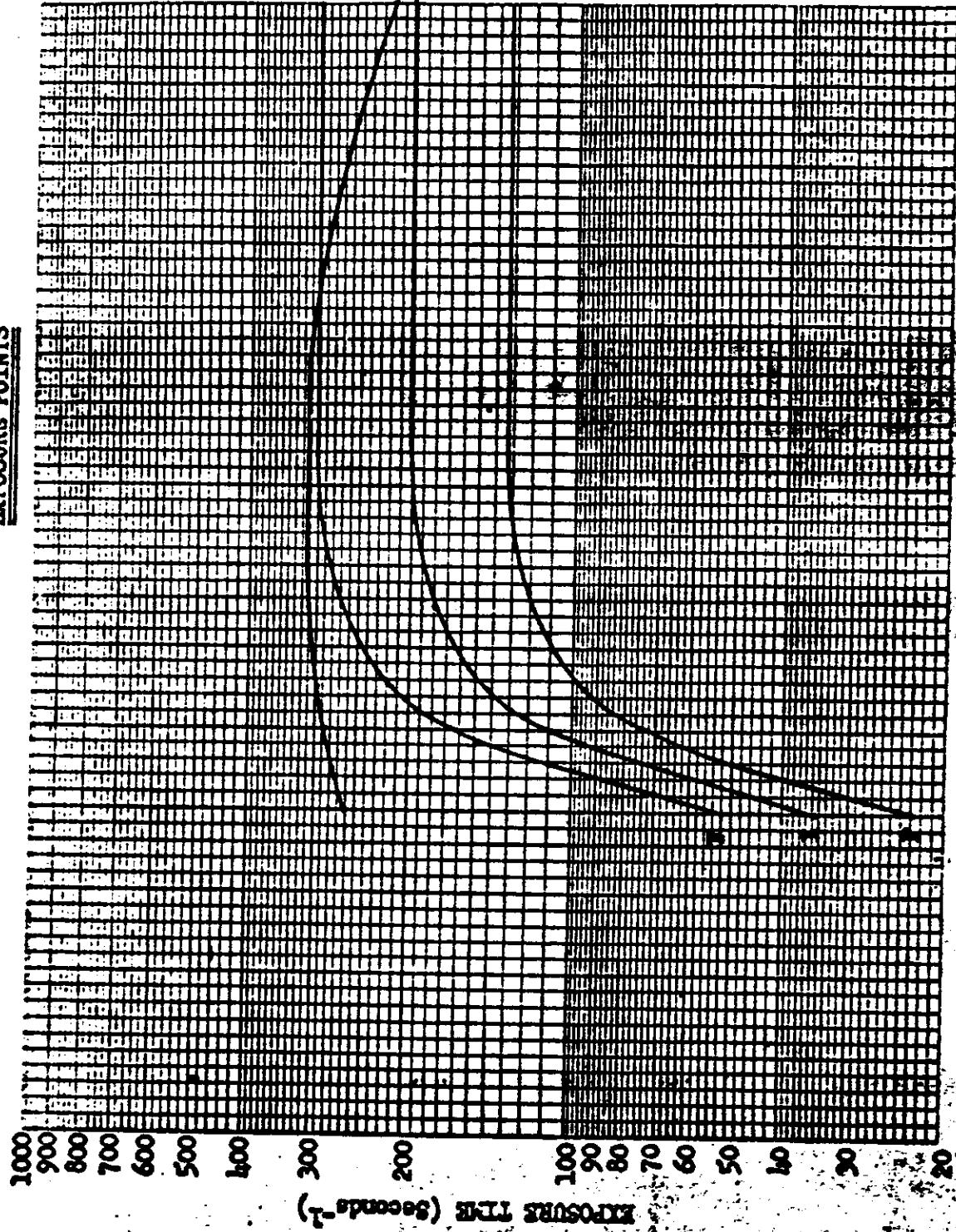
No. [REDACTED]

EXPOSURE POINTS



No. [REDACTED]

EXPOSURE POINTS



TOP SHEET
FIGURE 1
100
90
80
70
60
50
40
30
20
10
0
Degrees North
Degrees South

PRINTED ON

BETWEEN 80TH

DEGREES NORTH

AND 30TH

DEGREES SOUTH

ON

10TH

20TH

30TH

40TH

50TH

60TH

70TH

80TH

90TH

100TH

110TH

120TH

130TH

140TH

150TH

160TH

170TH

180TH

190TH

200TH

210TH

220TH

230TH

240TH

250TH

260TH

270TH

280TH

290TH

300TH

310TH

320TH

330TH

340TH

350TH

360TH

370TH

380TH

390TH

400TH

410TH

420TH

430TH

EXPOSURE POINTS

NO. 1

EXPOSURE TIME (Seconds)

七

EXPOSURE POINTS

Mission No: 1026

Payload No: J-25

Camera No: 175

Pass No: 56

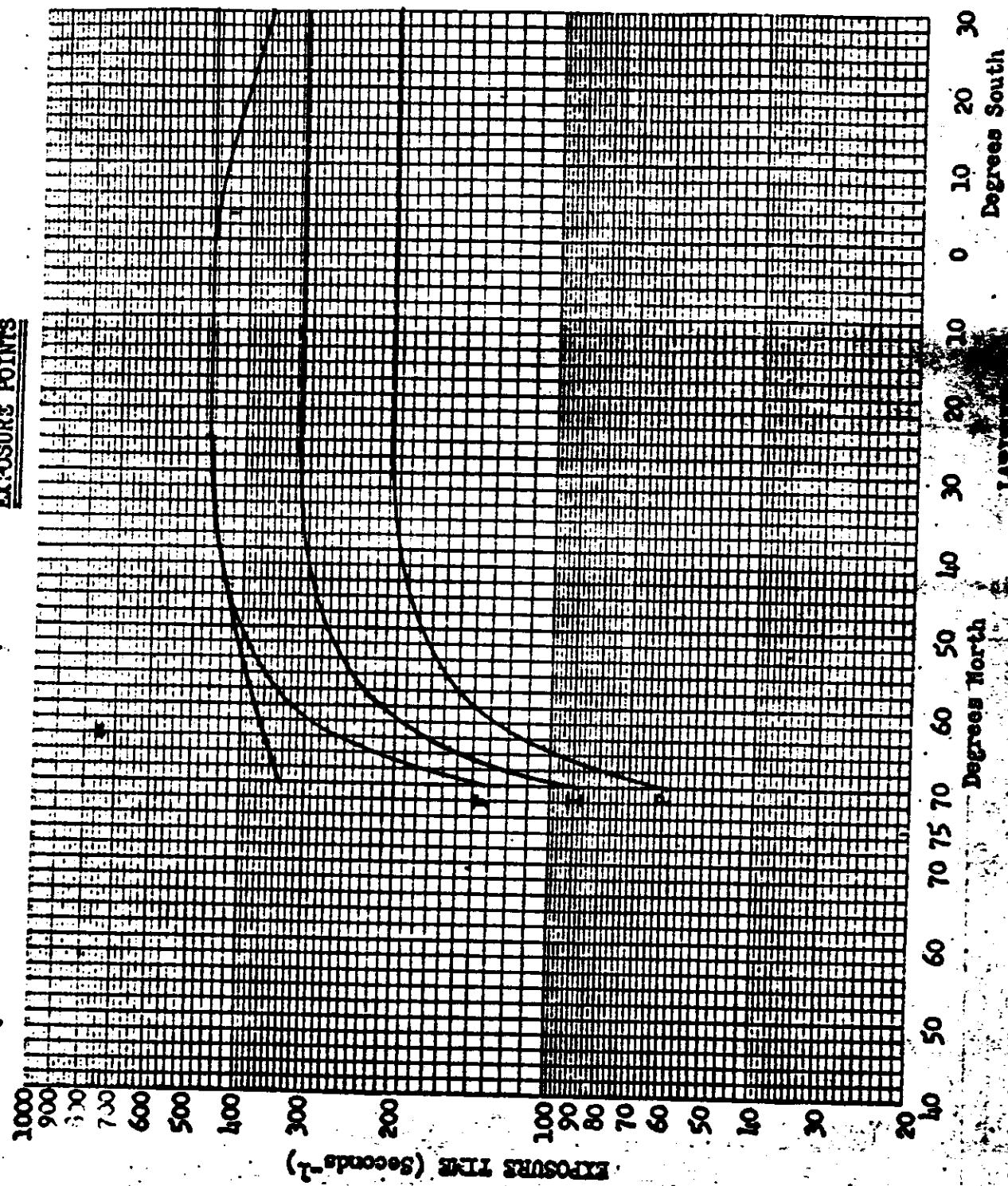
Launch Date: 10/28/65

Launch Time: 2347Z

•150

Filter Type: Watten

סמלים וזמן



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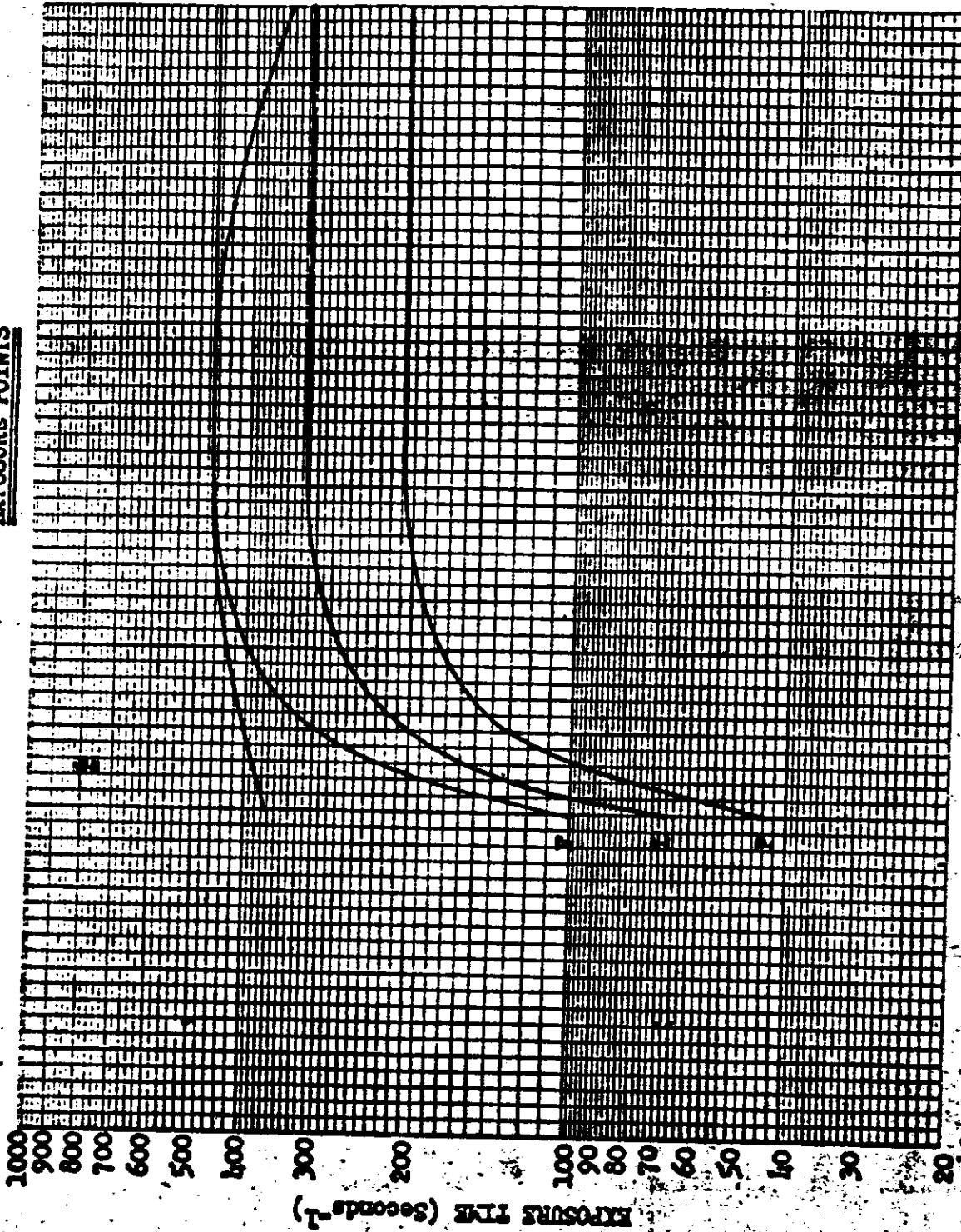
Degrees South

TOP SECRET

Degrees North

TOP SECRET

Exposures Time (seconds) 200 180 160 140 120 100 80 60 40 20 0



No.

TOP SECRET

EXPOSURE POINTS

TOP SECRET//
REF ID: A6512

TOP SECRET//
REF ID: A6512

Degrees South

0 10 20 30 40 50 60 70 80 90

Degrees North

20 30 40 50 60 70 80 90 100 110 120 130 140 150 160 170 180 190 200 210 220 230 240 250 260 270 280 290 300 310 320 330 340 350 360 370 380 390 400 410 420 430 440 450 460 470 480 490 500 510 520 530 540 550 560 570 580 590 600 610 620 630 640 650 660 670 680 690 700 710 720 730 740 750 760 770 780 790 800 810 820 830 840 850 860 870 880 890 900 910 920 930 940 950 960 970 980 990 1000

EXPOSURE TIME (Seconds⁻¹)

EXPOSURE POINTS

TOP SECRET//
REF ID: A6512

Mission No: 1026

Payload No: J-25

Camera No: 175

Pass No: 152

Launch Time: 2017 2 10/20/69

Launch Date: 10/20/69

Slit Width: .150

Filter type: Wratten

Filter No: 310L

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No.

SECTION 9

DIFFUSE DENSITY MEASUREMENTS

The diffuse density measurements made by AFSPFF were computer sorted at A/P to permit analysis of the density ranges encountered at the three processing levels. A study of sorting techniques showed that no absolute method was available to separate the density values as the accuracy of the Processing History published by [REDACTED] appears rather low and processing transition phases are not accounted for. The sorting technique selected uses the base plus fog density values where measurements up to 0.09 density are considered as having received Primary processing, 0.10 to 0.17 as Intermediate and above 0.17 density as Full. The percentage of original negative that was processed at each level, based on the computer sort, is tabulated below with the predicted and reported processing percentages.

Mission	Camera		Primary	Intermediate	Full
1026-1	FWD	Predicted	0	4	98
		Reported	0	21	79
		Computed	0	24	76
1026-1	AFT	Predicted	0	5	95
		Reported	0	4	96
		Computed	0	37	63
1026-2	FWD	Predicted	1	0	99
		Reported	2	5	93
		Computed	0	5	95
1026-2	AFT	Predicted	1	0	99
		Reported	2	5	93
		Computed	0	15	85

The tabulations of density frequency distributions for Missions 1026-1 and 1026-2 are included in Appendix A, Table A-1 thru A-4. The graphical presentation of the density distribution are computer plotted in Appendix A Figures A-1 thru A-36.

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A summary of the processing and exposure analysis is shown in Table 9-1. The terrain D_{Min} criteria, (range) for proper exposure and processing is 0.40 to 0.90 density units. The area measured for D_{Min} is selected subjectively and is not necessarily the absolute D_{Min} in the photography.

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- CONTROL NO.

MISSION 1026-1 INSTR - FRHD 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	59	3 PC	53 PC	41 PC	44 PC	40 PC
FULL	154	74 PC	100 PC	24 PC	28 PC	20 PC
ALL LEVELS	243	57 PC	13 PC	28 PC	28 PC	20 PC

MISSION 1026-1 INSTR - AFT 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	90	0 PC	71 PC	0 PC	0 PC	0 PC
FULL	150	0 PC	107 PC	0 PC	0 PC	0 PC
ALL LEVELS	240	69 PC	27 PC	0 PC	0 PC	0 PC

MISSION 1026-2 INSTR - FRHD 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	124	0 PC	30 PC	0 PC	0 PC	0 PC
FULL	226	978 PC	302 PC	119 PC	119 PC	119 PC
ALL LEVELS	330	978 PC	302 PC	119 PC	119 PC	119 PC

MISSION 1026-2 INSTR - AFT 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	124	0 PC	79 PC	0 PC	0 PC	0 PC
FULL	226	70 PC	12 PC	25 PC	25 PC	25 PC
ALL LEVELS	330	70 PC	12 PC	25 PC	25 PC	25 PC

MISSION 1026-2 BASE FOG 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	BASE EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	124	164 PC	70 PC	12 PC	12 PC	12 PC
FULL	226	60 PC	12 PC	25 PC	25 PC	25 PC
ALL LEVELS	330	60 PC	12 PC	25 PC	25 PC	25 PC

MISSION 1026-2 INSTR - AFT 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	BASE EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	124	0 PC	0 PC	0 PC	0 PC	0 PC
FULL	226	0 PC	0 PC	0 PC	0 PC	0 PC
ALL LEVELS	330	0 PC	0 PC	0 PC	0 PC	0 PC

MISSION 1026-2 INSTR - FRHD 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	BASE EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	124	0 PC	0 PC	0 PC	0 PC	0 PC
FULL	226	0 PC	0 PC	0 PC	0 PC	0 PC
ALL LEVELS	330	0 PC	0 PC	0 PC	0 PC	0 PC

MISSION 1026-2 INSTR - AFT 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	BASE EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	124	0 PC	0 PC	0 PC	0 PC	0 PC
FULL	226	0 PC	0 PC	0 PC	0 PC	0 PC
ALL LEVELS	330	0 PC	0 PC	0 PC	0 PC	0 PC

MISSION 1026-2 INSTR - FRHD 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	BASE EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	124	0 PC	0 PC	0 PC	0 PC	0 PC
FULL	226	0 PC	0 PC	0 PC	0 PC	0 PC
ALL LEVELS	330	0 PC	0 PC	0 PC	0 PC	0 PC

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SECTION 10

PERFORMANCE MEASUREMENTS

The photography acquired by both panoramic cameras during Missions 1026-1 and 1026-2 received a MIP rating of 85. A summary is tabulated below of the MTF/AIM resolution values measured by AFSPPF [REDACTED]

[REDACTED] The microdensitometer slit used by AFSPPF [REDACTED]
[REDACTED] was 1 micron by 80 microns.

<u>Mission</u>	<u>Camera</u>	<u>AFSPPF</u>
1026-1	FWD	76
1026-2	AFT	88
1026-2	FWD	85
1026-2	AFT	93

The details of the measurement and computing techniques, targets measured and target locations are fully reported in the evaluation report published by AFSPPF and are not normally included in this report.

SECTION 11

OBSERVED DATA

Mission 1026-1 provided photographic coverage of 3 fixed and 2 mobile ground resolution targets. The best measurable performance was demonstrated on Pass D-47 where the Indian Springs hight contrast CORN target was resolved to 7.7 feet by both Master and Slave cameras. Two frames later, (45 miles) Southwest of Indian Springs near Pahrump, Nevada, both cameras resolved 12 feet on a mobile medium contrast T-Bar target.

This indicates that on-orbit system resolution capability was at least 130 lines/mm for high contrast targets. The dynamic system resolution test at A/P showed a capability of 194 lines /mm. The on-orbit resolution limit, as computed from vehicle attitude rates and IMC error, was less than one foot along track and 5.8 feet cross track due, principally to earth rotation. This would limit on-orbit resolution to approximately 160 lines/mm. It is highly probable that the loss of resolution, from the limiting 160 lines/mm to the observed 130 lines/mm, is due to atmospheric attenuation and its attendant lowering of contrast.

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No.

SECTION 12

MISSION 1026-1 STELLAR-INDEX CAMERA

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Camera	D-75
Index Reseau	92
Stellar Reseau	93

B. CAMERA DATA AND FLIGHT SETTINGS

Stellar Camera:

Lens	85 mm f/1.8
Exposure Time	2 seconds
Filter Type	None
Film Type	Eastman Type 3401

Index Camera:

Lens	38 mm f/4.5
Exposure Time	1/500 second
Filter Type	Wratten 21
Film Type	Eastman Type 3400

C. POST FLIGHT EVALUATION

Approximately 20 star images were recorded on each frame of stellar photography throughout mission 1026-1 by S/I #D75/92/93. Although some star imagery was elongated due to vehicle motion during shutter open time, star imagery was adequate to determine vehicle attitude.

Four consecutive stellar frames out of 404 recovered frames from mission 1026-1 were affected by faulty shutter operation. Stellar frames 182 and 183 contained no star imagery and no flare fog imagery of any kind. It is assumed that the stellar shutter failed completely closed during the

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normal exposure time of frames 182 and 183. Frame 184 and 185 contain gross fogging that reduced the star count in frame 184 to 50% of normal; frame 185 to 0% stars recorded. It is assumed that the stellar shutter failed either partly open or fully open during the normal exposure time of frames 184 and 185. Also, the frame line is exposed between frames 184 and 185 suggesting that the stellar shutter was partly open during film meter. Investigation of the stellar camera shutter anomaly has been initiated to determine the precise cause of the failure.

Minor edge static and no corona discharge marks were observed in stellar photography.

Stellar baffle flare fog was low on all but a few frames. The first 18 frames of stellar photography contain streaks of what is presumed to be particles of jettisoned fuel.

Emulsion cracking parallel to the minor axis of the film begins in frame 264 and continues until the tail end of the roll.

The index camera operated well throughout the mission producing 416 frames of good photography. No index camera malfunctions occurred. Dendritic static and corona discharge marks were absent from all index camera photography.

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No.

SECTION 13

MISSION 1026-2 STELLAR-INDEX CAMERA

A. COMPONENT ASSIGNMENT

<u>Component</u>	<u>Serial Number</u>
Camera	D-72
Index Reseau	89
Stellar Reseau	85

B. CAMERA DATA AND FLIGHT SETTINGS

Stellar Camera:

Lens	85 mm f/1.8
Exposure Time	2 seconds
Filter Type	None
Film Type	Eastman Type 3401

Index Camera:

Lens	38 mm f/4.5
Exposure Time	1/500 second
Filter Type	Wratten 21
Film Type	Eastman Type 3400

C. POST FLIGHT EVALUATION

S/I #D 72/89/85 operated normally throughout mission 1026-2. Stellar film was expended in frame 55 of pass D 148 after producing 388 frames of acceptable stellar photography containing approximately 20 stars per frame. Dendritic static and corona discharge were absent from stellar and index camera photography. The index camera operation was good throughout mission 1026-2 and produced 434 frames of recovered photography. Stellar/Index camera operation is rated excellent throughout mission 1026-2.

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SECTION 14

VEHICLE ATTITUDE

The vehicle attitude errors for both Mission 1026-1 and 1026-2 were derived from the reduction of the Stellar camera photography. This attitude date is supplied to A/P by NPIC.

The attitude errors for each frame and the attitude control rates are calculated at the A/P computer facility. The computer also plots the frequency distribution of the rates and errors. Figures 14-1 through 14-6 show these distributions for Mission 1026-1 and Figures 14-7 through 14-12 for Mission 1026-2.

The summary table below lists the maximum attitude errors and rates that were experienced during 90% of the FWD camera photographic operations, excluding the first six frames of each operation, and the total range of the errors and rates.

<u>Value</u>	Mission 1026-1		Mission 1026-2	
	<u>90%</u>	<u>Range</u>	<u>90%</u>	<u>Range</u>
Pitch Error ($^{\circ}$)	0.65	-0.92 to +0.28	0.55	-0.88 to +0.14
Roll Error ($^{\circ}$)	0.24	-0.40 to +0.60	0.56	-0.56 to +0.98
Yaw Error ($^{\circ}$)	0.70	-0.56 to 1.45	0.87	-1.10 to +0.65
Pitch Rate ($^{\circ}/hr.$)	37.90	-80 to +100	41.08	-85 to +85
Roll Rate ($^{\circ}/hr.$)	33.15	-75 to +70	46.52	-95 to +95
Yaw Rate ($^{\circ}/hr.$)	28.48	-58 to +54	30.75	-38 to +62

The performance of the attitude control system is comparable to the control systems used on recent missions. The panoramic photography was not degraded by the attitude control system.

-25A-BUCKET FORWARD INSTRUME FRAMES 1-6 ON EACH OF THE LIFTED 90 PERCENT

Y PITCH ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY - PERCENT (%)

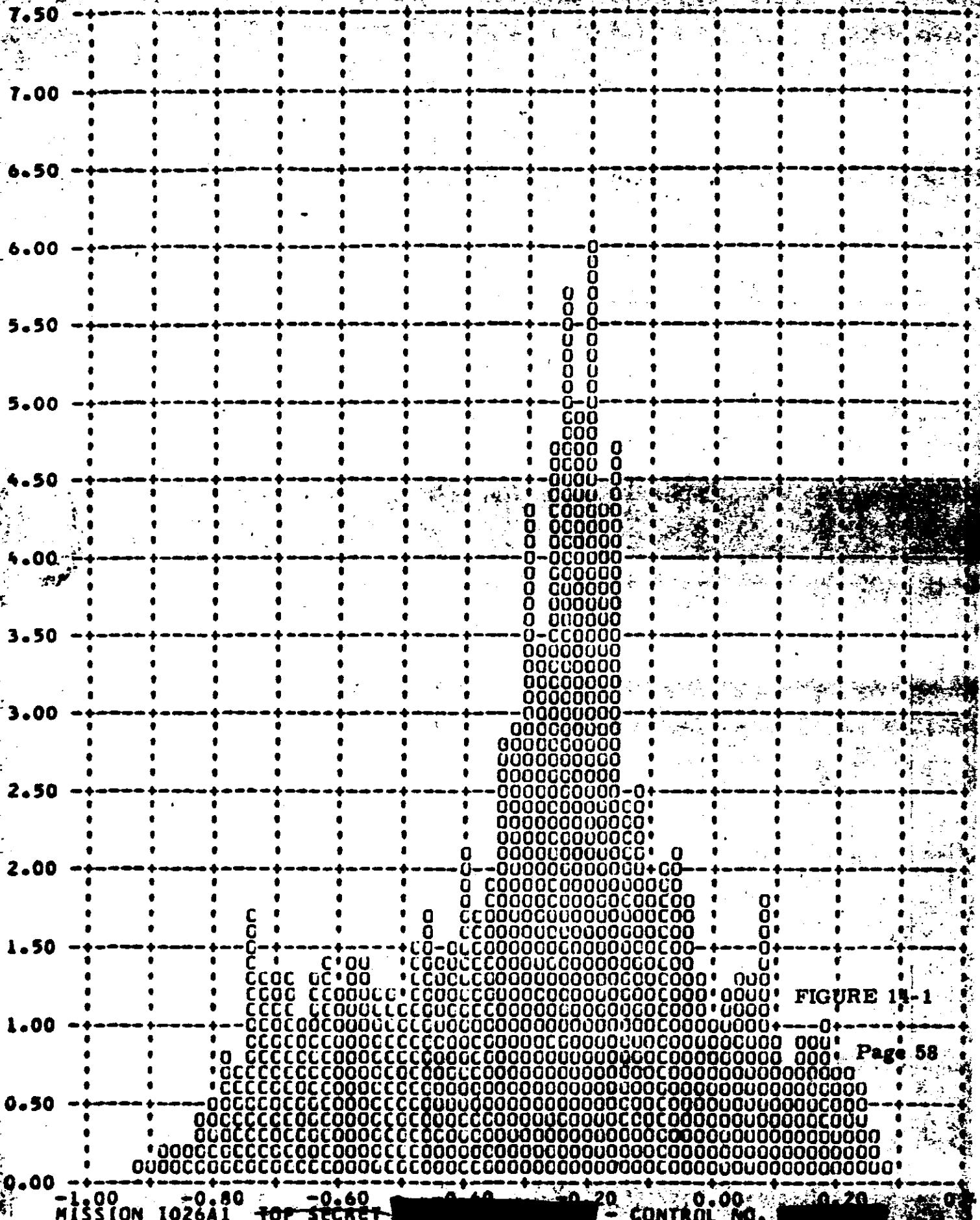


FIGURE 14-1

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ROLL ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY (PERCENT)

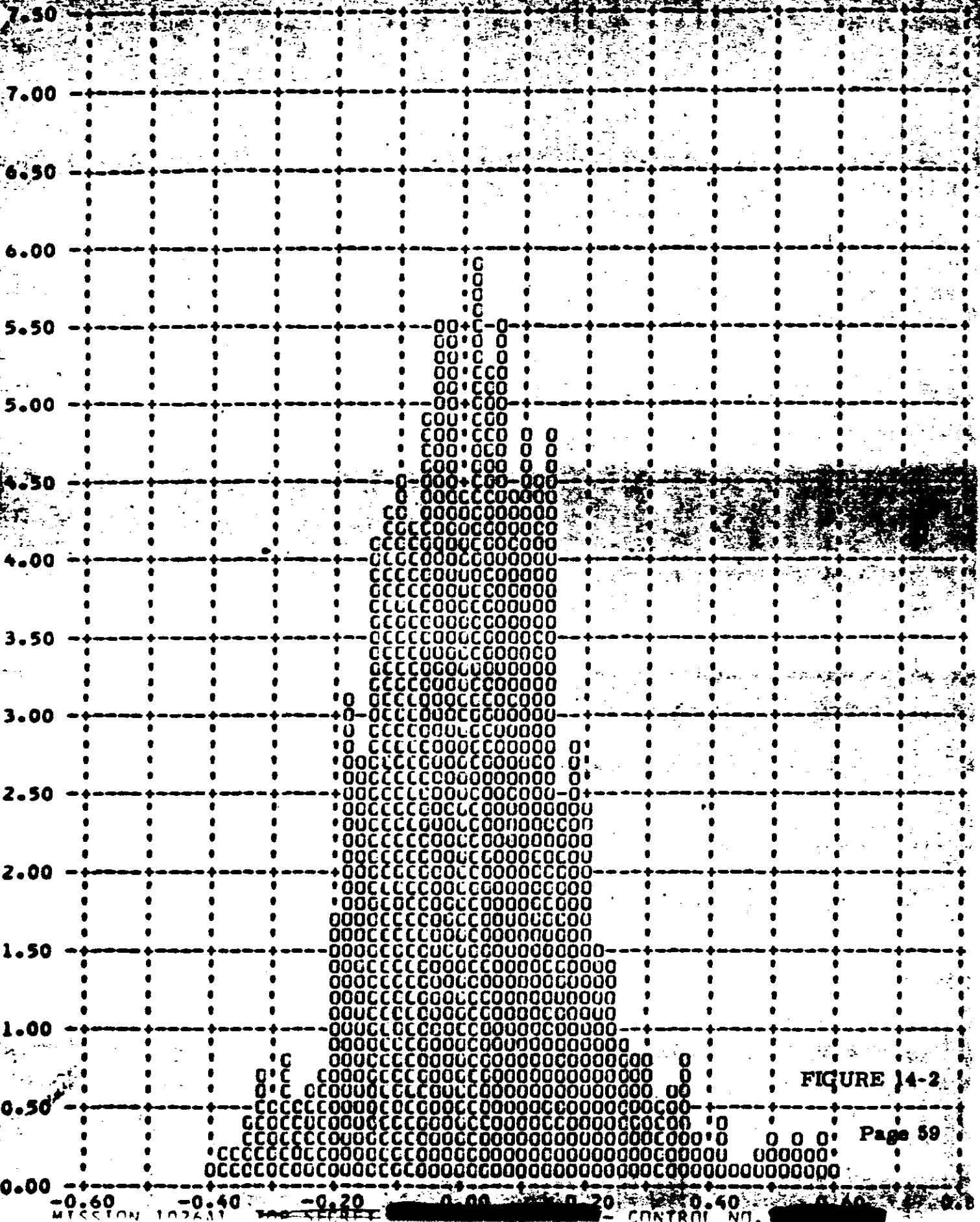


FIGURE 14-2

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MISSION 102641

CONTROL NO.

YAW ANGLE ERROR - DEGREES (XT) VERSUS FREQUENCY - PERCENT (Y)

3.00

14.00

13.00

12.00

11.00

10.00

9.00

8.00

7.00

6.00

5.00

4.00

3.00

2.00

1.00

0.00

-1.00

-0.50

0.00

0.50

1.00

1.50

2.00

MISSION 102641

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CONTROL NO.

FIGURE 14-3

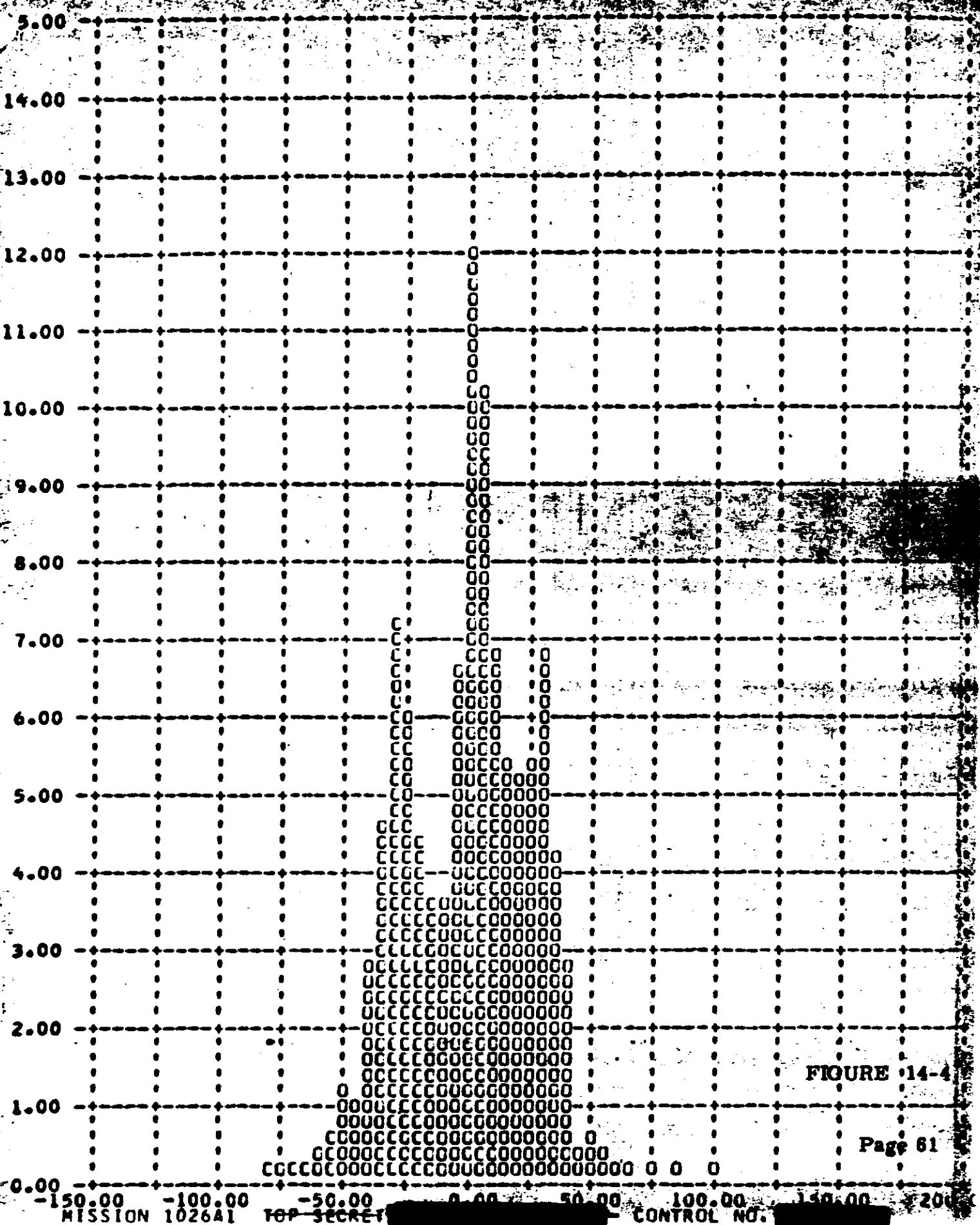
Page 60

MISSION 102641

CONTROL NO.

NUCLEAR WEAPONS SYSTEMS - SEARCH

Y PITCH RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)



Y - ROLL RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY - PERCENT (Y)

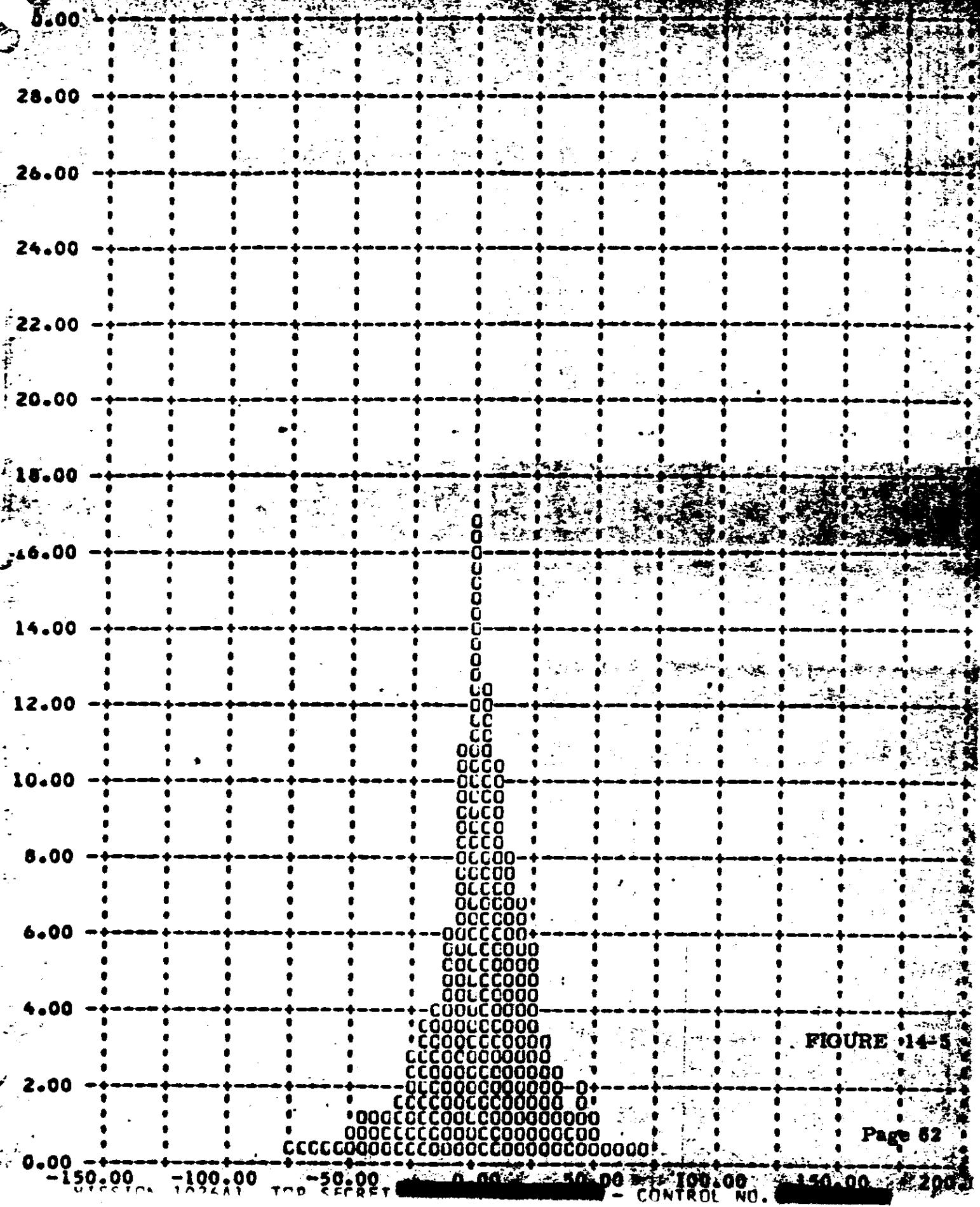


FIGURE 14-5

Y = TAN RATE ENRRCR - DEG/HCUR (X) VERSUS FREQUENCY - PERCENT (%)

7.50

7.00

6.50

6.00

5.50

5.00

4.50

4.00

3.50

3.00

2.50

2.00

1.50

1.00

0.50

0.00

-60.00

-40.00

-20.00

0.00

20.00

40.00

60.00

80.00

100.00

MISSION 102681

CONTROL NO.

PITCH ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY (%)

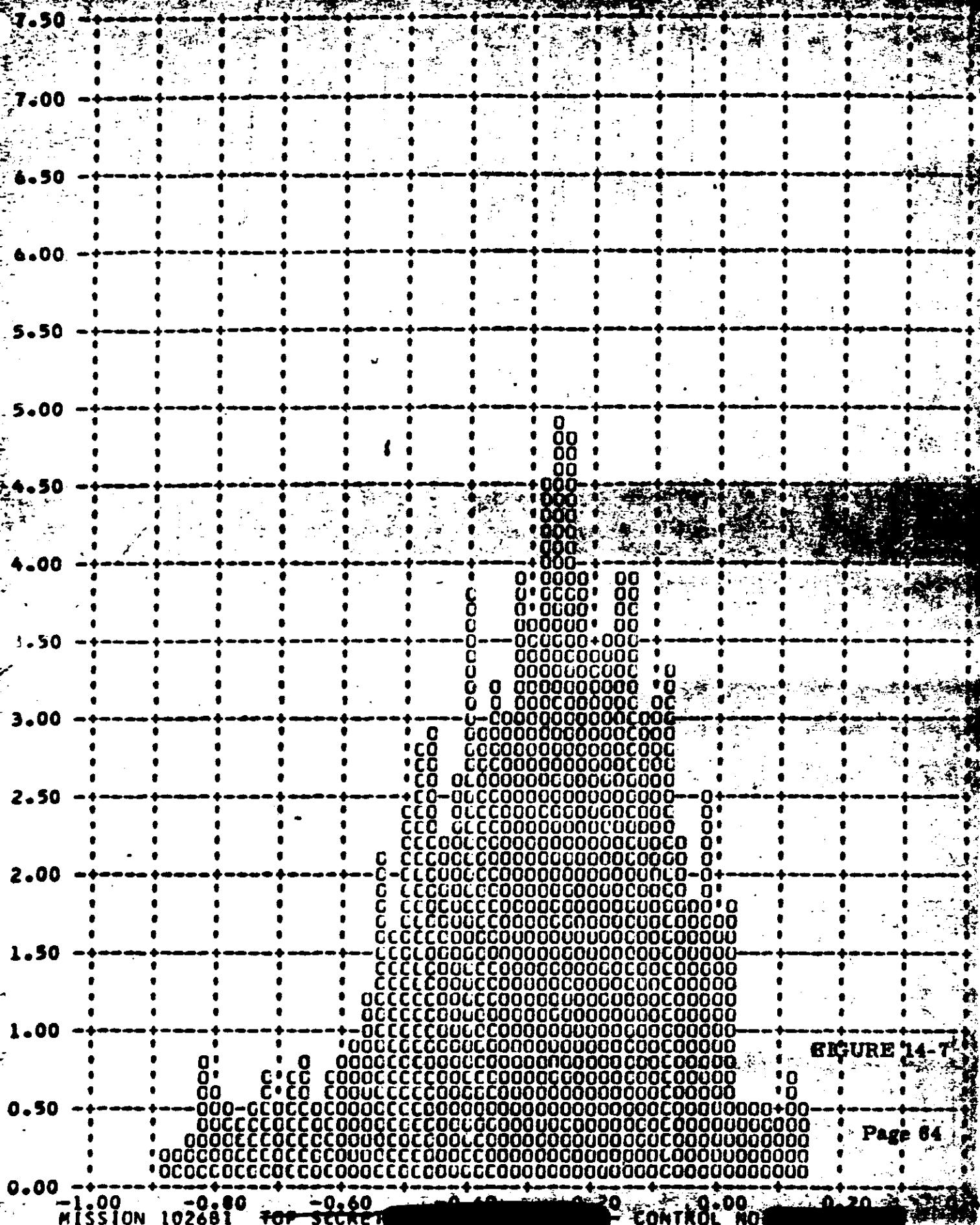


FIGURE 14-7

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ROLL ANGLE ERROR - DEGREES AT VERSUS FREQUENCY - PERCENT (%)

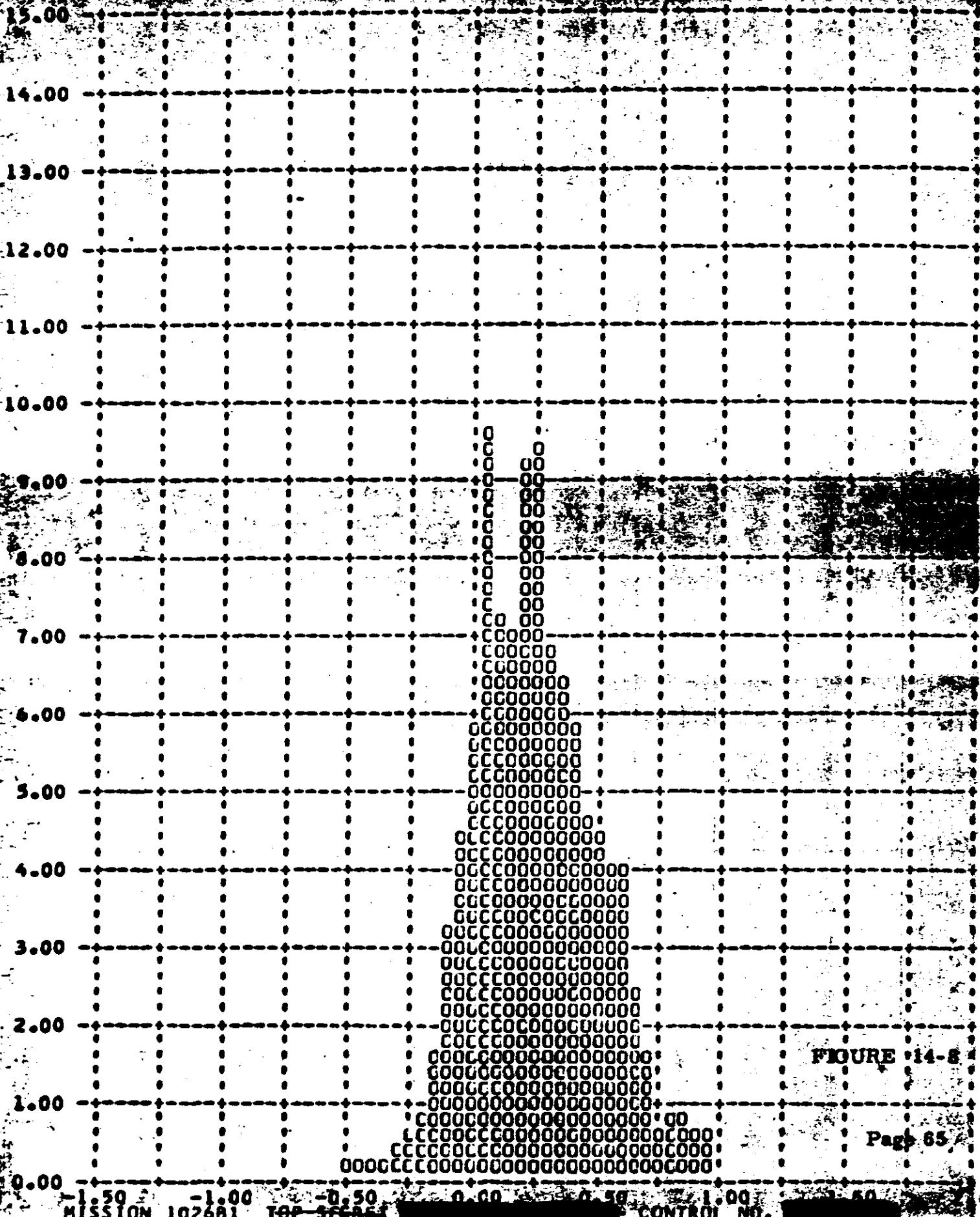


FIGURE 14-8

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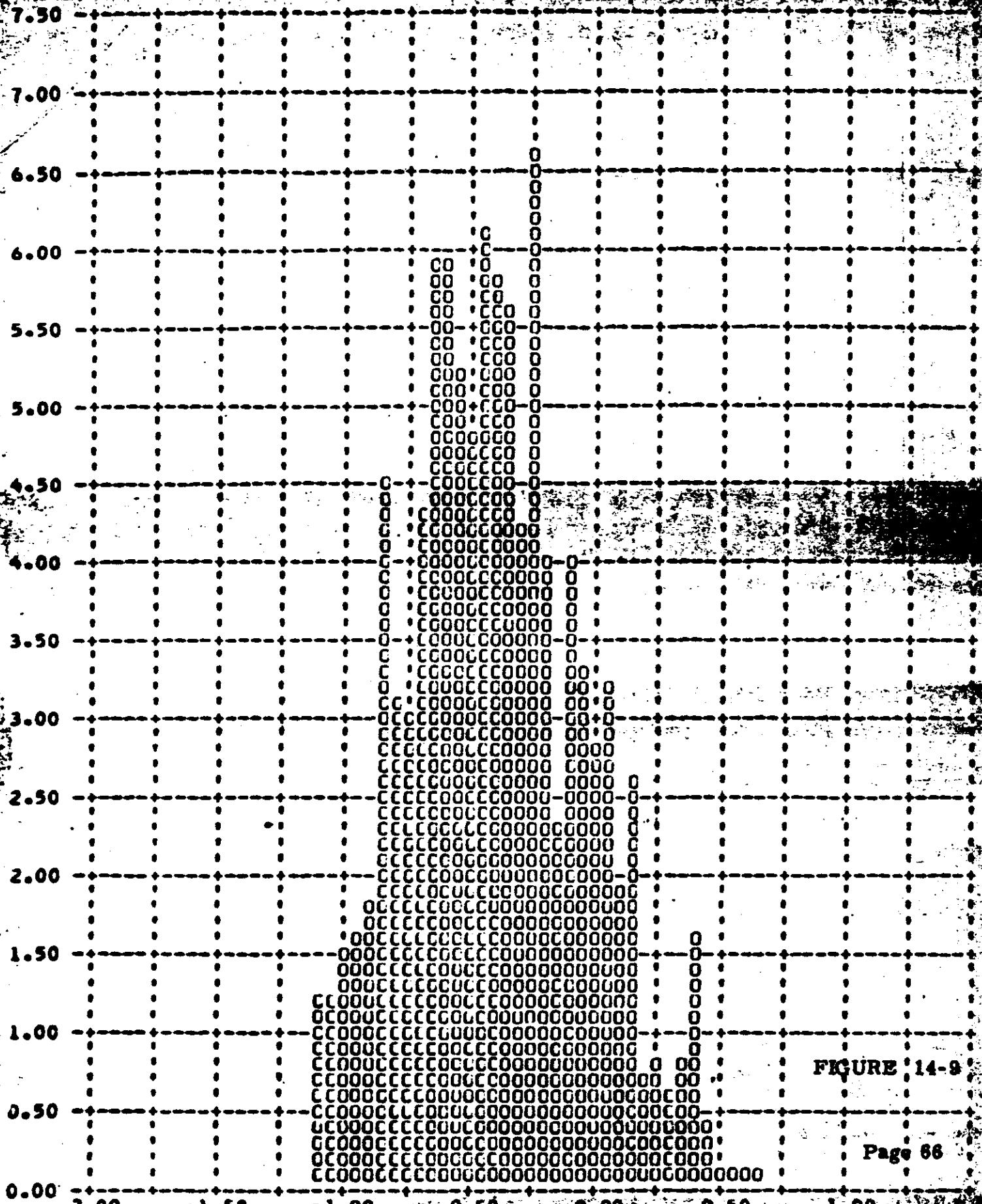
-25 E-BUCKET FORWARD INSTRUMENT PAN AND TILT TESTS
YAW ANGLE ERROR - DEGREES (X) VERSUS FREQUENCY (%)

FIGURE 14-8

PITCH RATE ERROR - DEG/HOUR (XT) VERSUS FREQUENCY - PERCENT (Y)

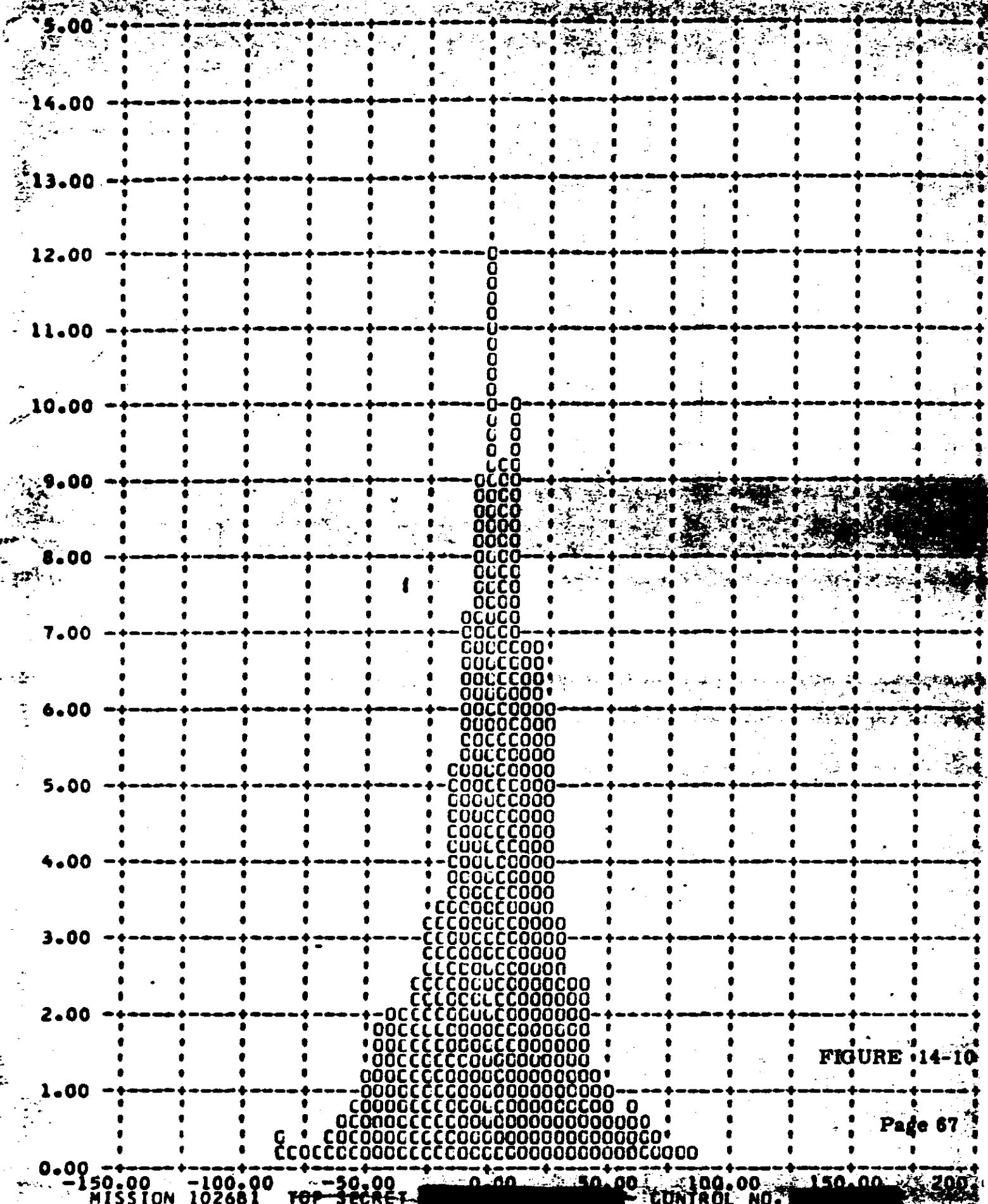


FIGURE 14-10

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ROLL RATE ERROR - DEG/HOUR (X) VERSUS FREQUENCY (PERCENT)

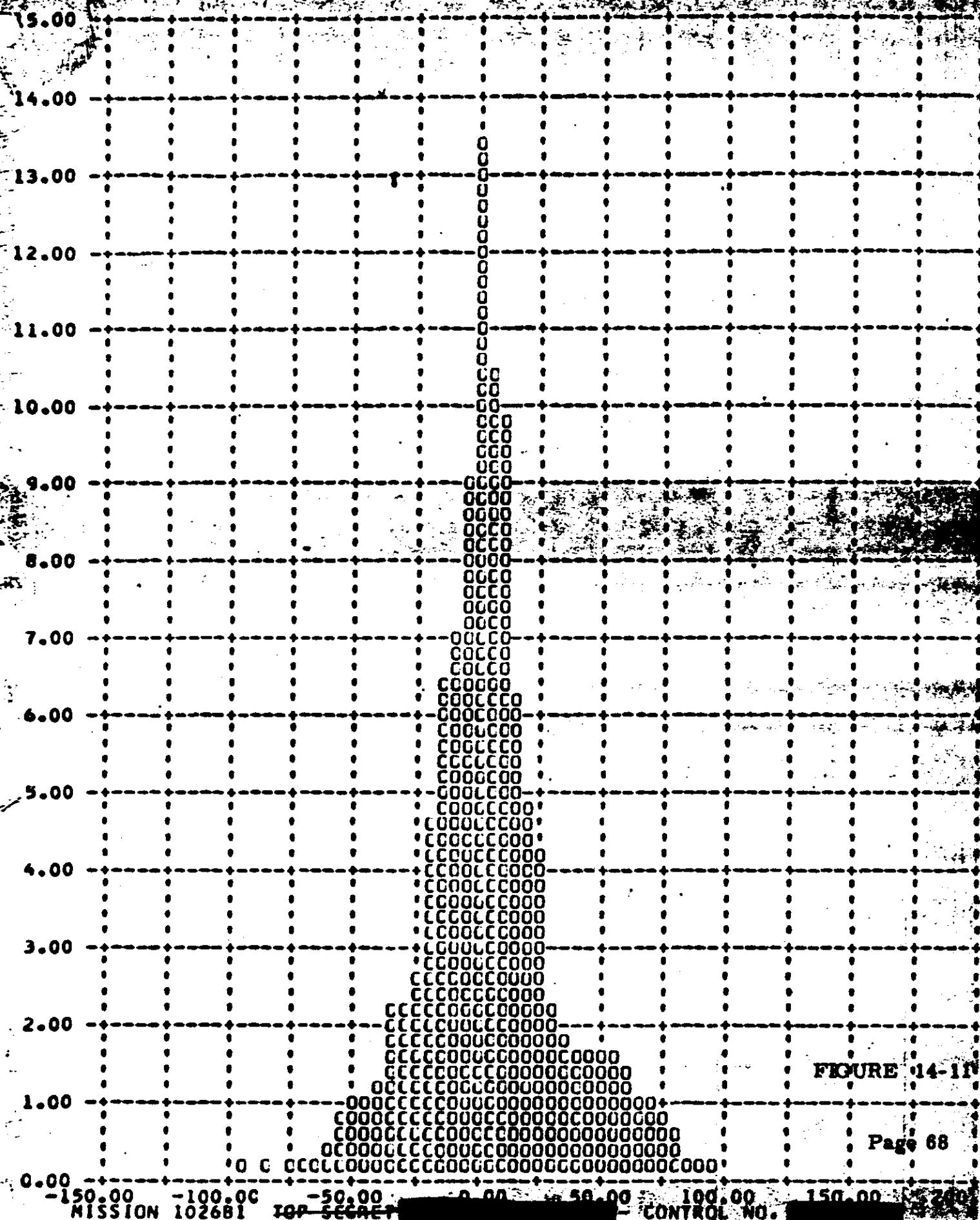


FIGURE 14-1B

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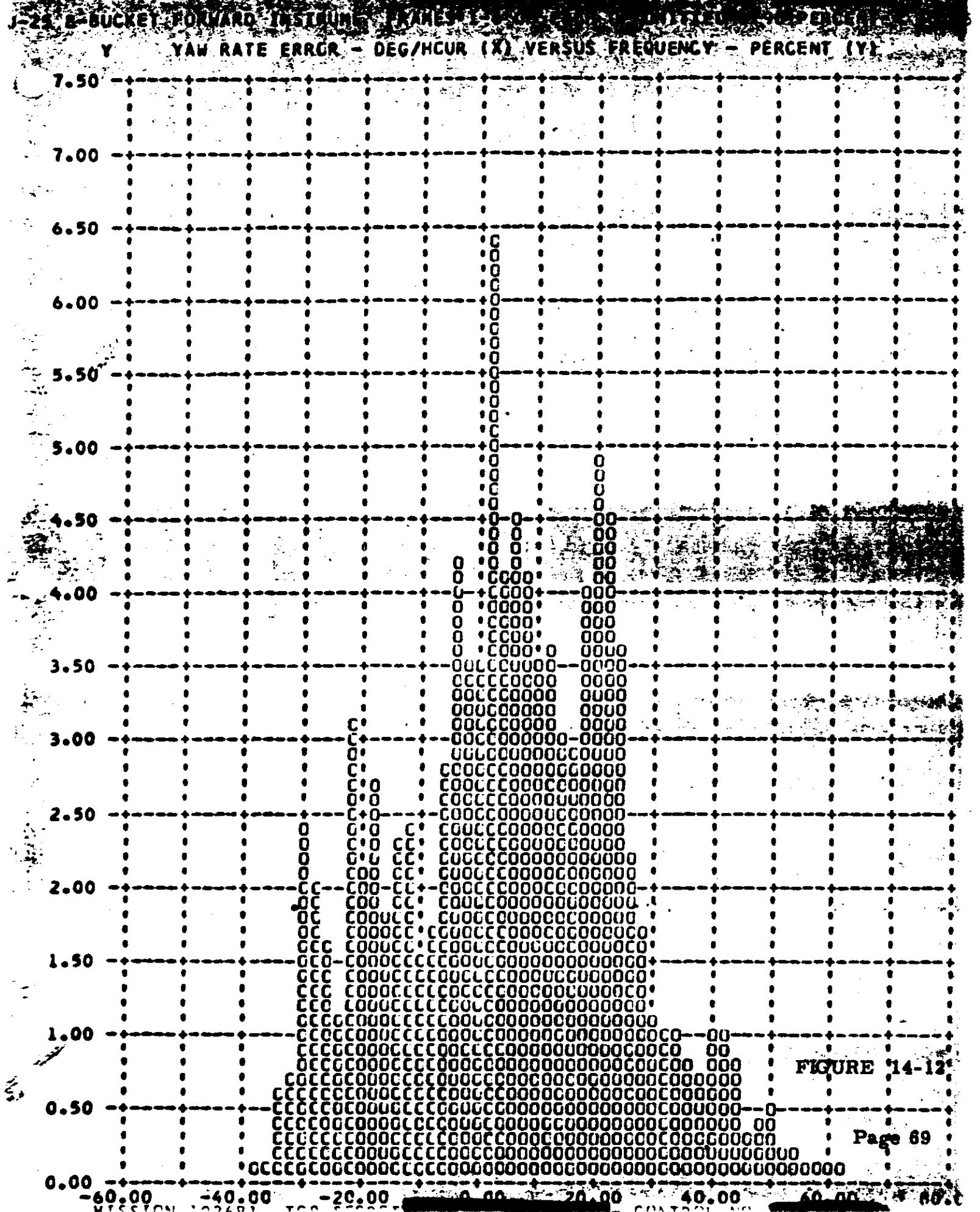


FIGURE 14-12

SECTION 15**IMAGE SMEAR ANALYSIS**

The frame correlation tape supplied to A/P by NPIC contains the binary time word of each frame of photography. A computer program has been assembled at A/P which calculates the exposure time of each frame and compares the camera cycle rate with the ephemeris to calculate the V/h mismatch. This data is combined with the vehicle attitude error and rate values of each frame and the crab error caused by earth rotation at the latitude of each frame. The program outputs the total along track and cross track IMC error and the limit of ground resolution that can be acquired by a camera regardless of focal length and system capabilities.

The computer rejects the first six frames of all operations as the large V/h error induced by camera start-up is not representative of the overall system operations. The frequency distribution of the V/h errors and resolution limits are computer plotted and are shown in Figures 15-1 through 15-12.

The summary table 15-1 presents the maximum V/h ratio errors and resolution limits that existed during 90% of the photographic operations and the total range of values during all operations that were computed.

The high "Along Track Resolution Limit" on the A mission was due to a low perigee. From latitudes 230 to 205 the Pan Camera cycle rate limiter prevented appropriate FMC match. Some operations on the first day also encountered error rates higher than normal, previous to adequate orbit definition and ramp settings.

Passes 34M (Median) and 35M (Median) had along track resolution limits as high as 17.5 feet. This type of pass near orbit tangency at this time of year exceeds the systems capability of matching IMC and the required exposure time. These two passes involved 356 frames. The vehicle H-timer reset error on orbit 33 caused all events on these 2 orbits to occur ten seconds early thereby displacing operations northward.

~~TOP SECRET~~

No.

MISSION 1026
V/h RATIO AND RESOLUTION LIMITS

UNITS	CAMERA	MISSION 1026-1		MISSION 1026-2	
		80%	Range	90%	Range
V/h Ratio Error	%	6.13	-11.5 to +8.5	6.08	-8.8 to +2.2
	FWD				
	AFT	6.13	-16.5 to +9.0	6.67	-8.2 to +1.4
Along Track Resolution Limit	Feet	13.50	0.5 to 17.5	5.47	0.2 to 7.4
Cross Track Resolution Limit	Feet	9.08	0.2 to 13.8	3.26	0.2 to 4.6

TABLE

~~TOP SECRET~~

MISSION 1026A

CONTROL NO.

Y V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)

10.00

28.00

26.00

24.00

22.00

20.00

18.00

16.00

14.00

12.00

10.00

8.00

6.00

4.00

2.00

0.00

-20.00

-15.00

-10.00

-5.00

0.00

5.00

10.00

15.00

MISSION 1026A1 TOP SECRET

CONTROL NO.

FIGURE 15-1

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ALONG TRACK RESOLUTION LENGTH PERCENT VERT. FREQUENCY PERCENT

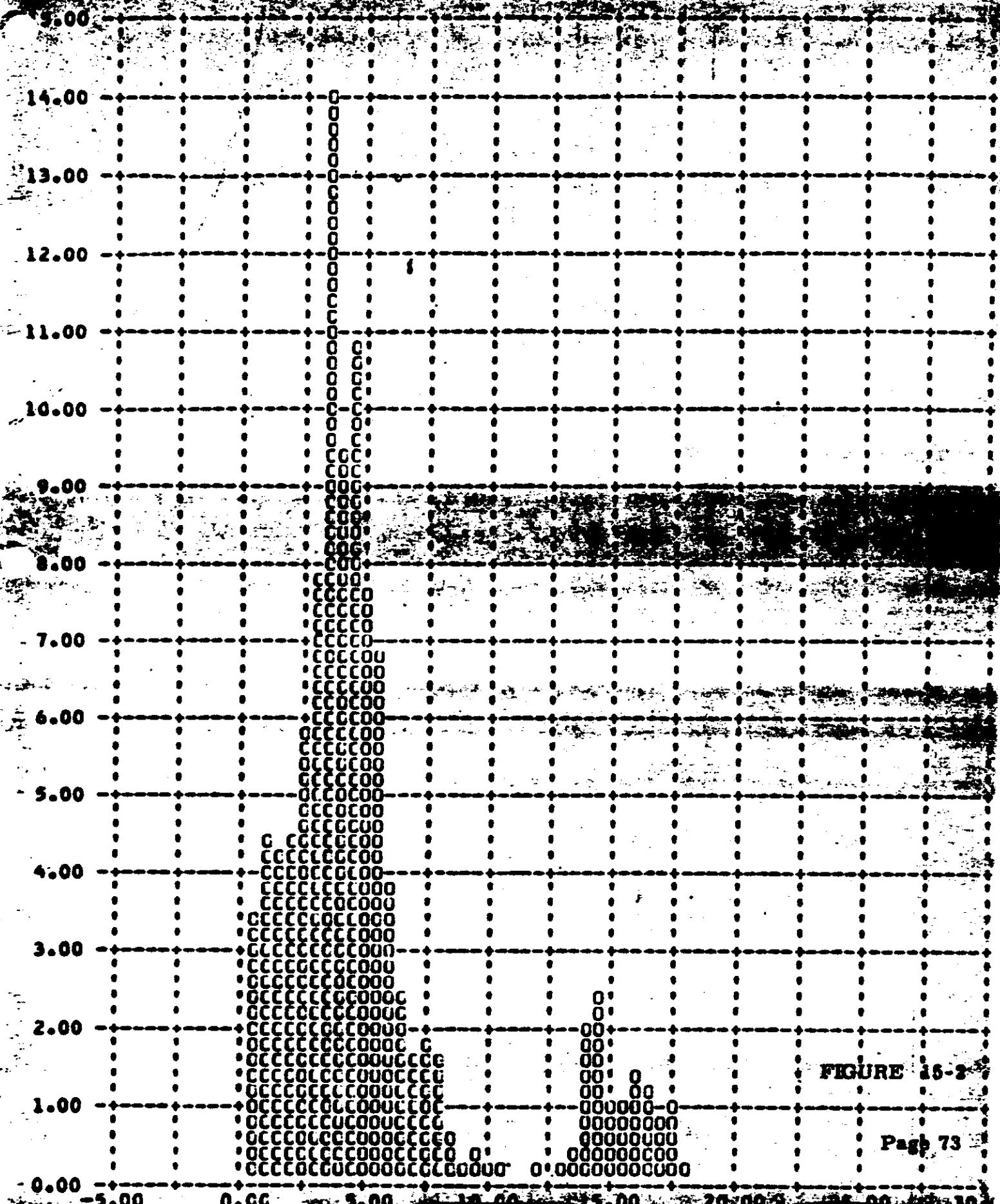


FIGURE 16-2

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V - CROSS TRACK RESOLUTION LIMIT - PEAK TO VERSUS FREQUENCY - PERCENT

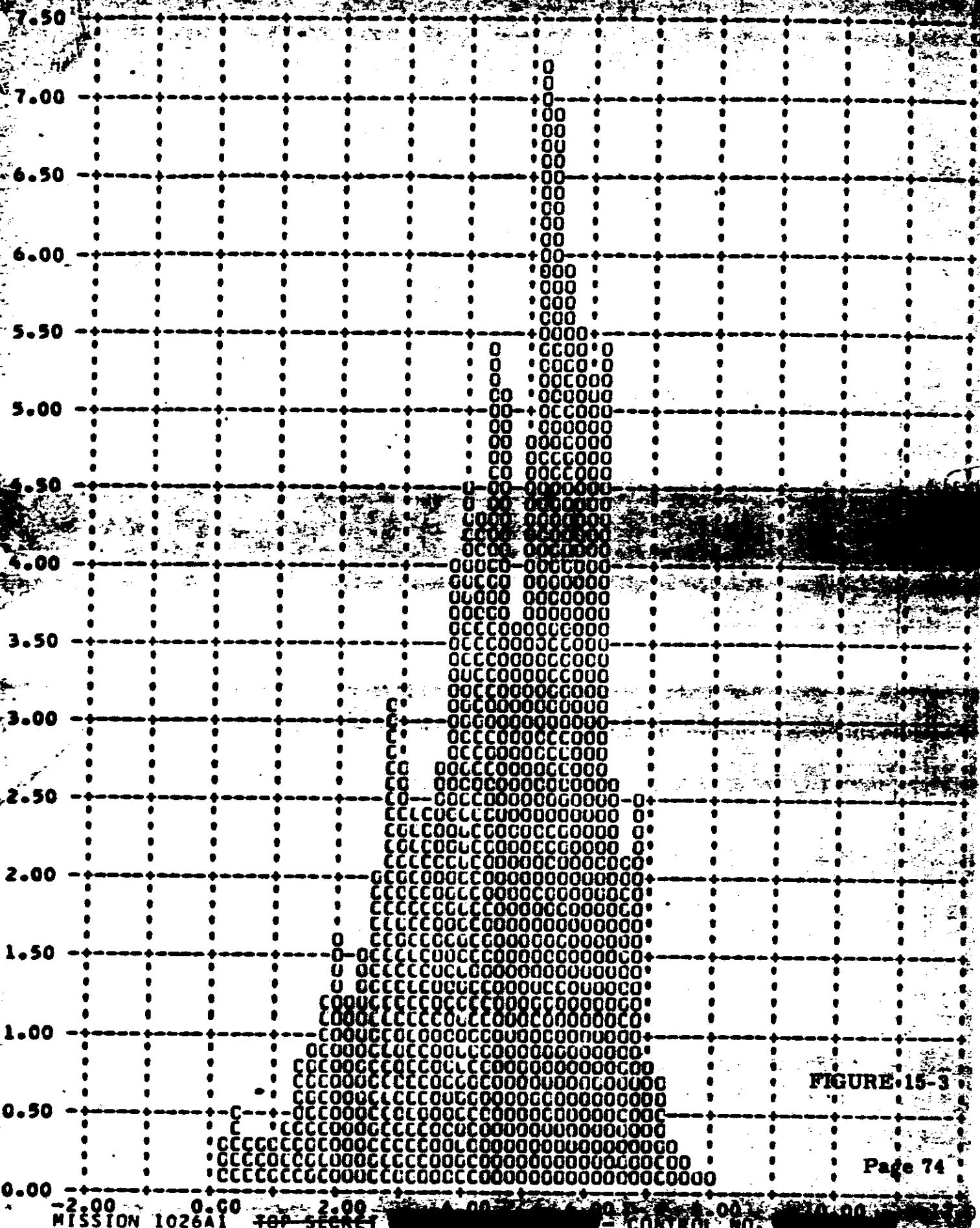


FIGURE 15-3

J-25 A-BUCKET-AFT INSTRUMENT FRAMES 1-6 OF EACH OF OMITTED 90 PERCENT
Y V/H RATIC ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)

30.00

28.00

26.00

24.00

22.00

20.00

18.00

16.00

14.00

12.00

10.00

8.00

6.00

4.00

2.00

0.00

-20.00

-15.00

-10.00

-5.00

0.00

5.00

10.00

15.00

J-25-A-BUCKET AFT INSTRUMENT FRAMES 1-6 OF EACH DFM OMITTED 5% 90 PERCENT
ALONG TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (%)

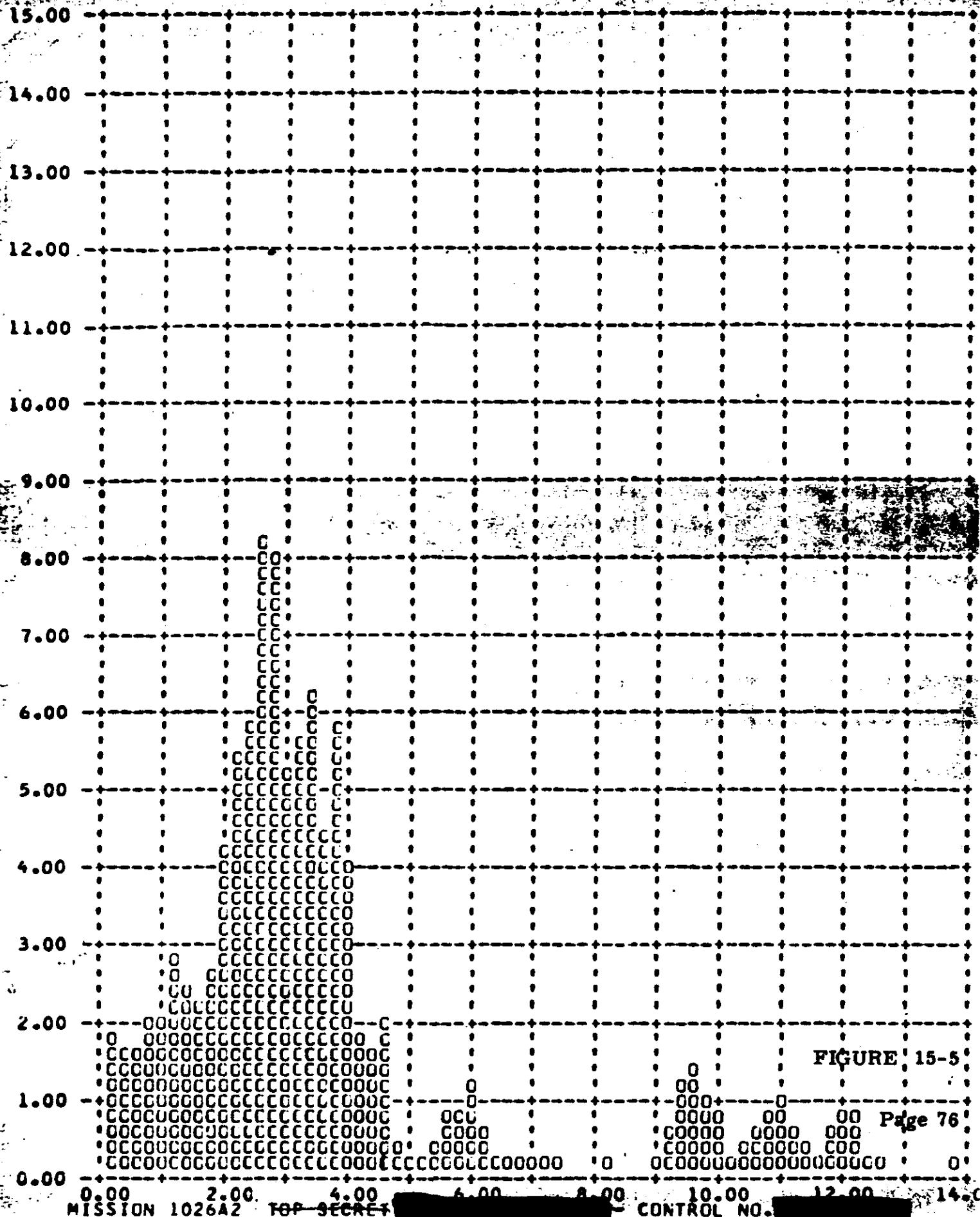


FIGURE 15-5

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CROSS TRACK RESOLUTION ETHTT - FEET/TAXI VERSUS FREQUENCY - PERCENT

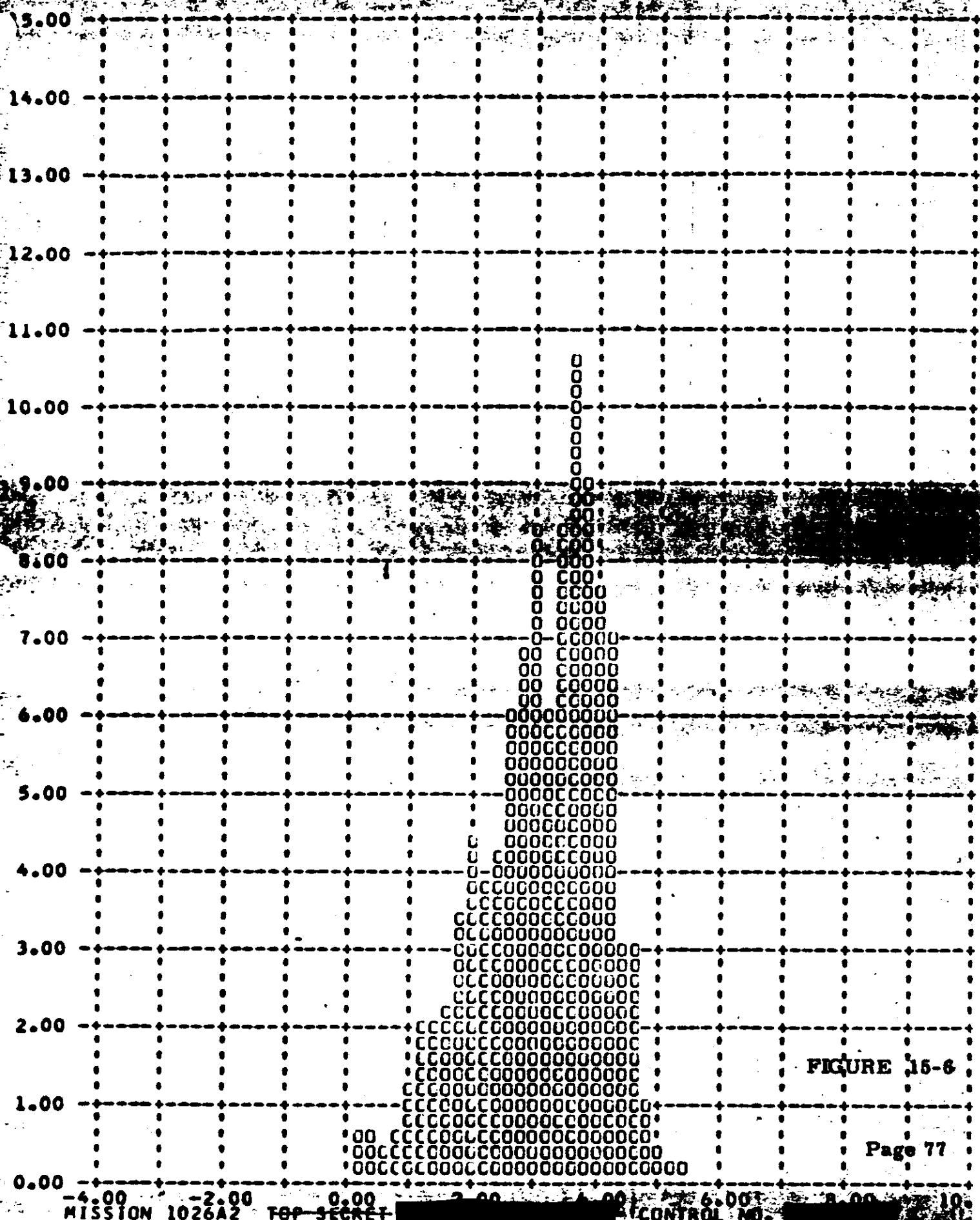


FIGURE 15-6

Page 77

-250-B BUCKET FORWARD INSTRUME~~N~~TE SCALES X & C FOR EACH OP POSITIONED 90 PERCENT
Y V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)

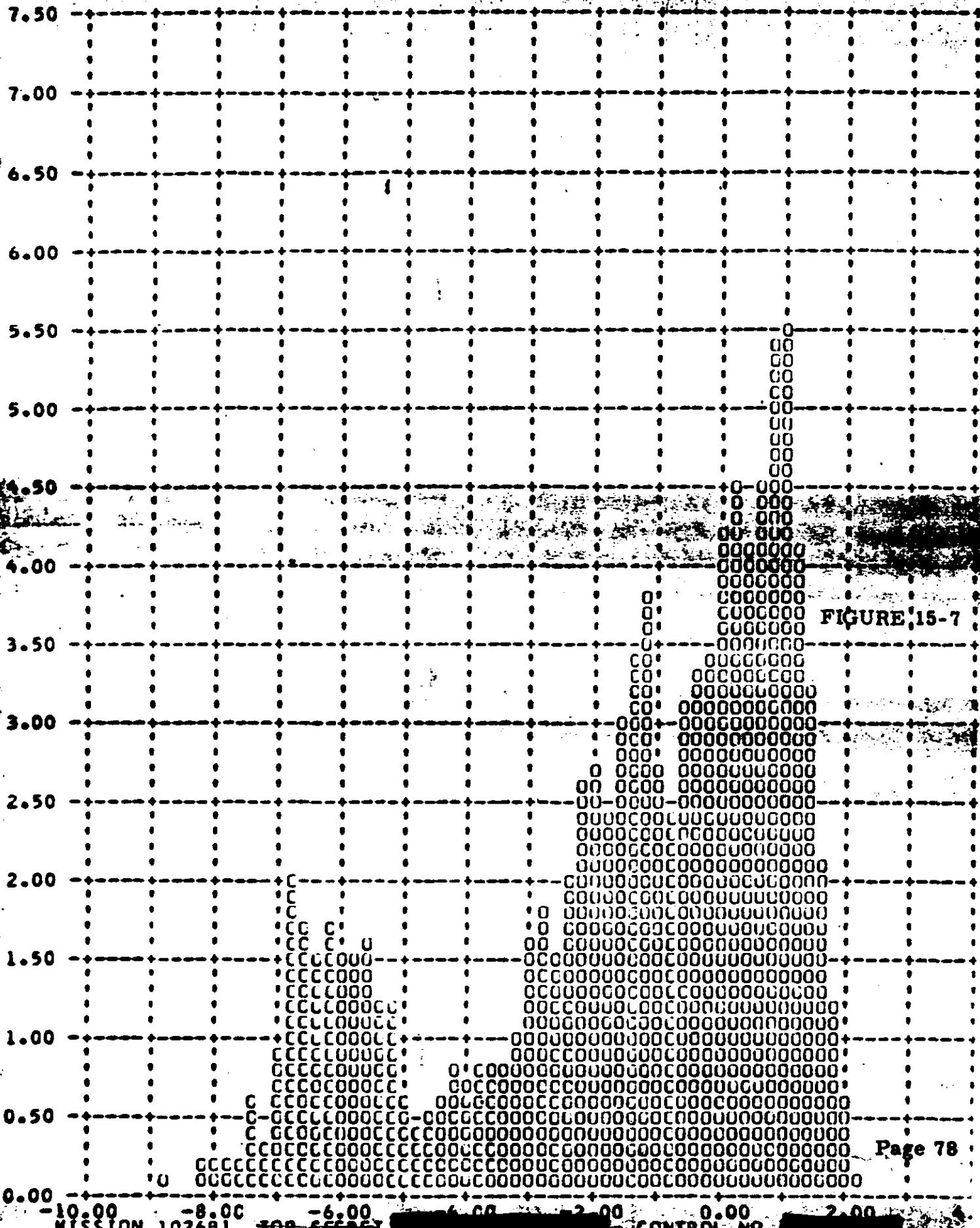


FIGURE 15-7

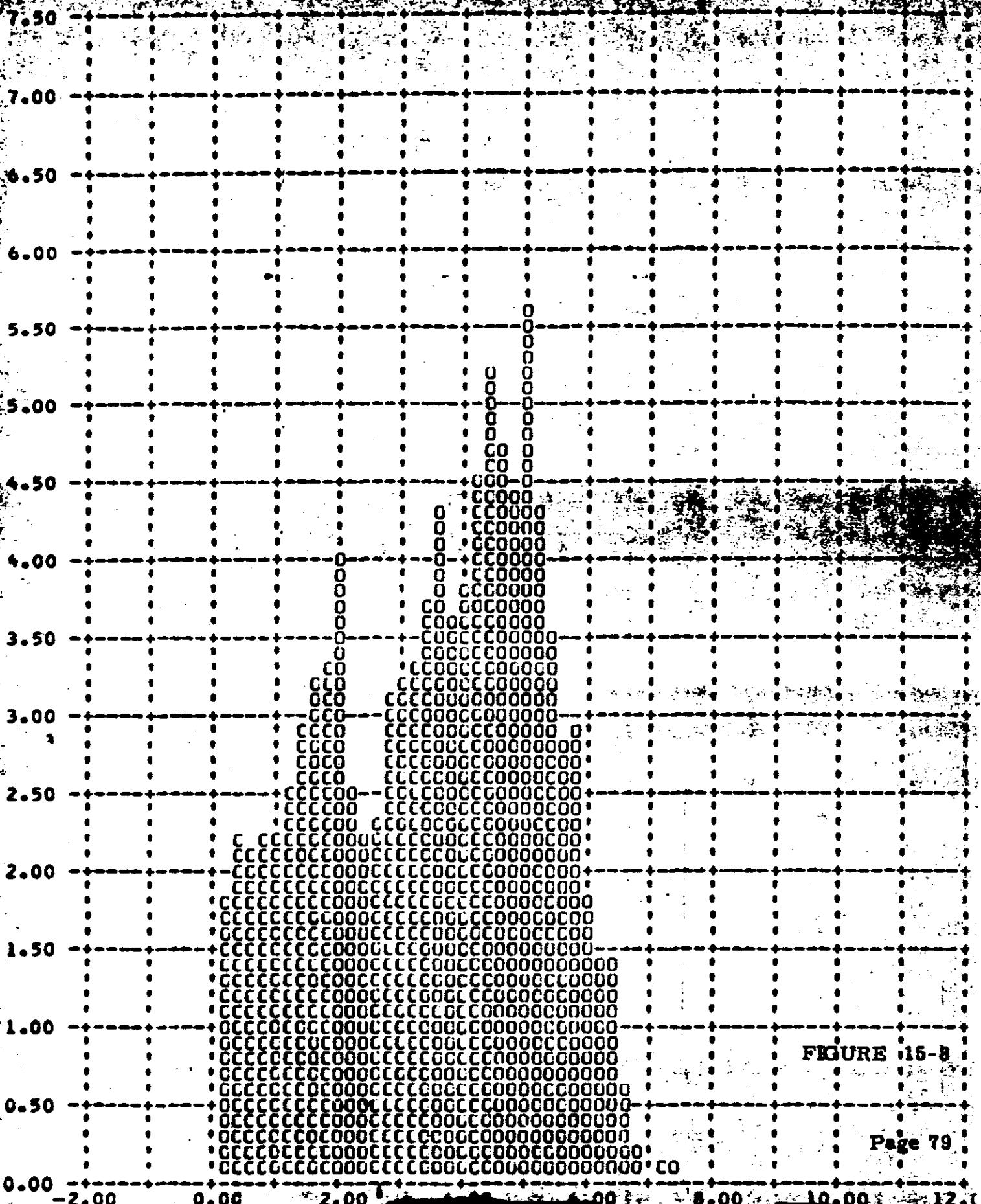
12 BUCKET FORWARD INSTRUMENTS - MEASUREMENT OF OMITTED - 90 PERCENT
ALONG TRACK RESOLUTION (MM) - PERTAIN VERSUS FREQUENCY - PERCENTAGE

FIGURE 15-8

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MISSION 102681 10 DEGREES

CONTROL NO.

BUCKET FORWARD INSTANT

BUCKET FORWARD

V. CROSS TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT

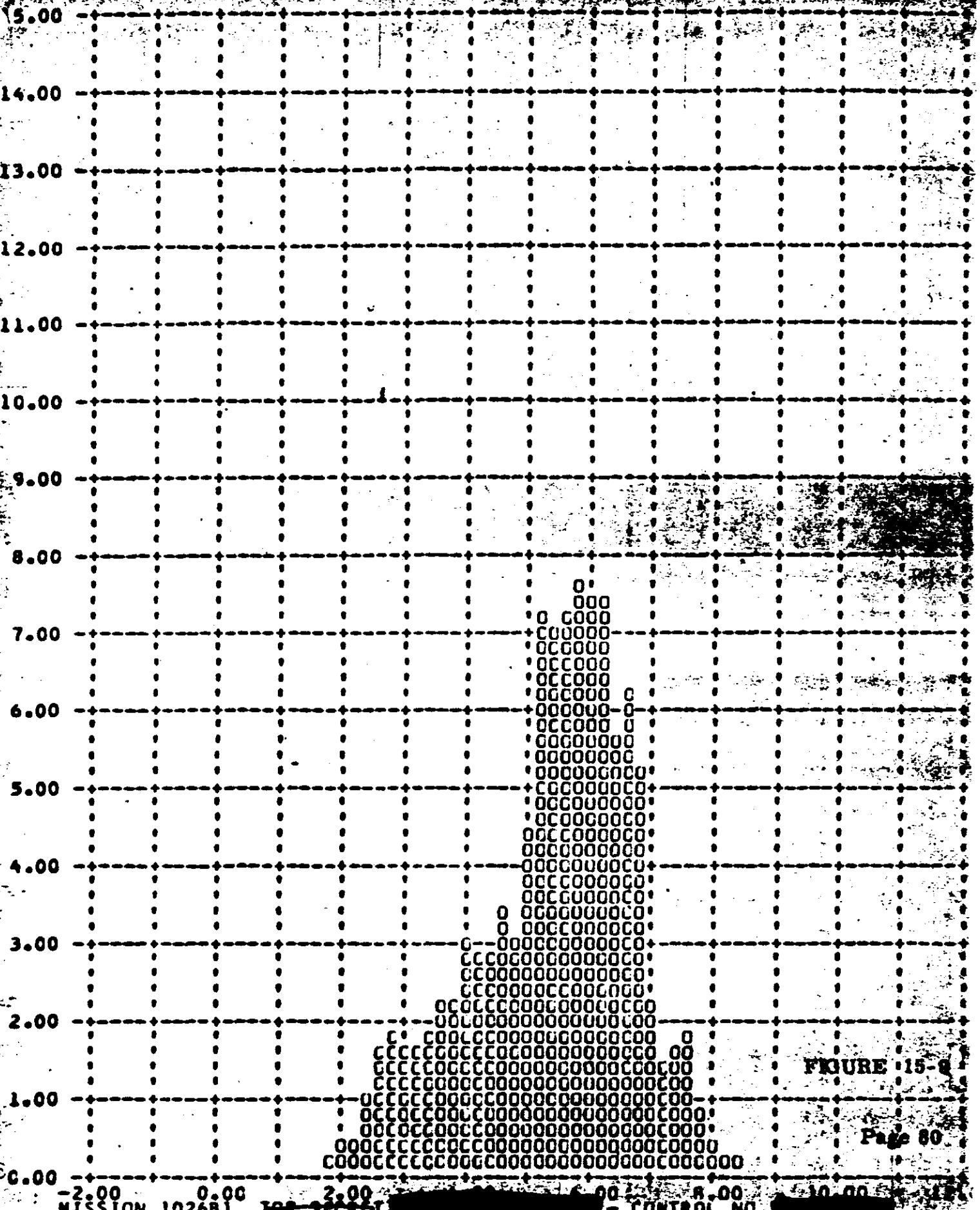


FIGURE 15-8

Page 50

MISSION 102681

TOP SECRET

- CONTROL NO.

BLOCK 25 OF THE INSTRUMENTATION TRAILER TESTED 100 PERCENT

Y = V/H RATIO ERROR - PERCENT (X) VERSUS FREQUENCY - PERCENT (Y)

7.50

7.00

6.50

6.00

5.50

5.00

4.50

4.00

3.50

3.00

2.50

2.00

1.50

1.00

0.50

0.00

-10.00

-8.00

-6.00

-4.00

-2.00

0.00

2.00

MISSION 102682 TOP SECRET

CONTROL NO.

FIGURE 15-10

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Y ALONG TRACK RESOLUTION LIMIT - FEET TAX VERSUS FREQUENCY - PERCENT

5.00

14.00

13.00

12.00

11.00

10.00

9.00

8.00

7.00

6.00

5.00

4.00

3.00

2.00

1.00

0.00

-4.00

-2.00

0.00

2.00

4.00

6.00

8.00

10

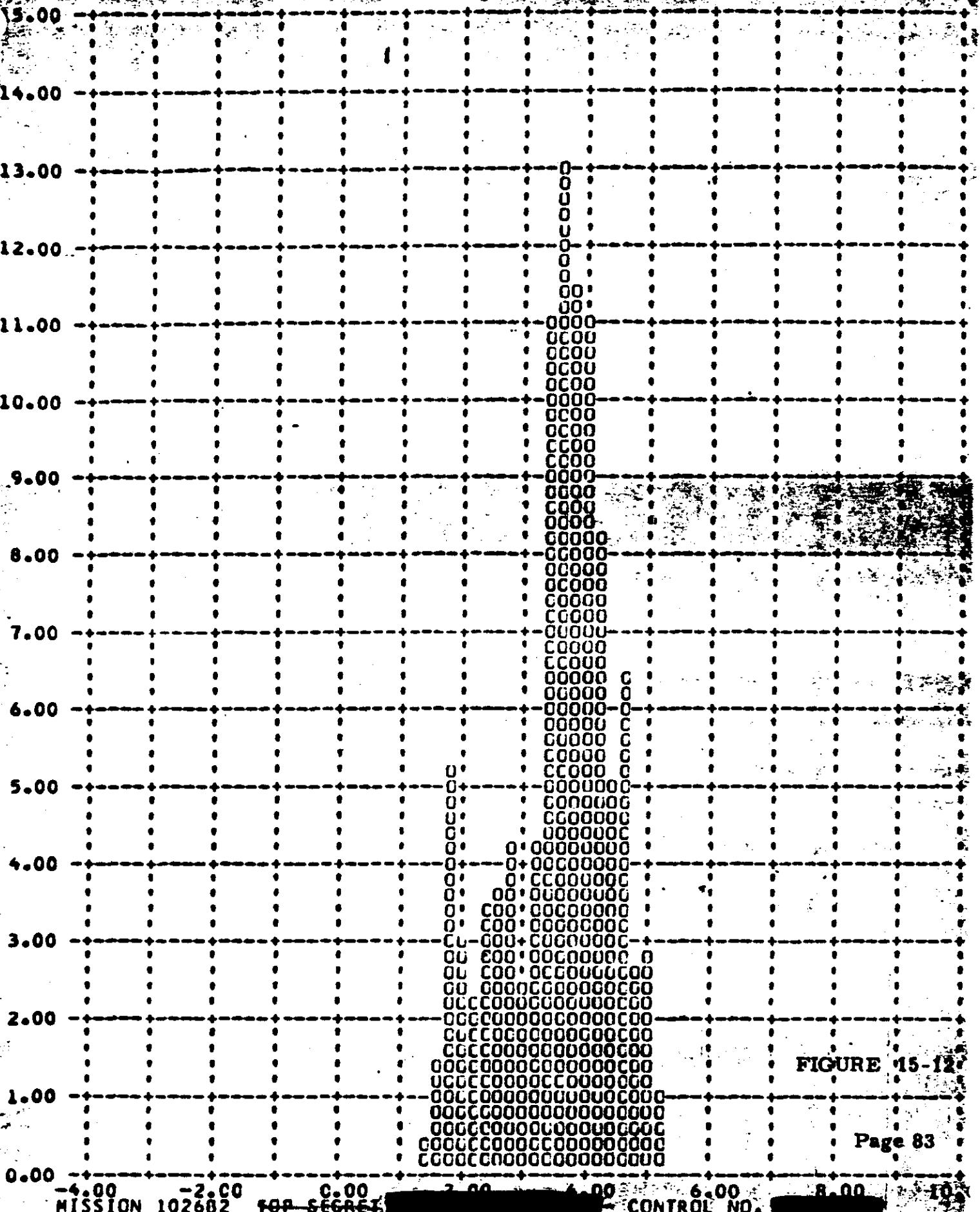
FIGURE 15-11

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MISSION 102682 FOR SOURCE CONTROL NO.

-2000-BUCKET AND INSTRUMENTS WORKS 1-NOV-1968 UNLIMITED 90 PERCENT

CROSS TRACK RESOLUTION LIMIT - FEET (X) VERSUS FREQUENCY - PERCENT (%)



SECTION 16

RADIATION DOSAGE

Each recovery system flown on a Corona mission contains a sealed packet of Eastman Type 3401 and Royal X Pan emulsions to determine the total radiation received at the take-up cassette. Both film types have been irradiated by LMSC at various levels and the base plus fog densities recorded after controlled processing.

Following recovery the film dosimeter packets are removed at A/P and processed with a pre-flight sample of the same film type and sensitometric control film. The resulting base plus fog density measurement of the dosimeter strips is used to ascertain the total radiation level. The table below presents the base plus fog readings for the dosimeter strips and the radiation level equivalents.

<u>Emulsion</u>	Mission 1026-1		Mission 1026-2	
	<u>B + F</u>	<u>Density</u>	<u>B + F</u>	<u>Density</u>
Type 3401	0.15	0.3 R	0.20	0.7 R
Royal X Pan	0.22	0.3 R	0.30	0.6 R

The radiation level on mission -1 was one half of that experienced on mission -2 material. These levels are below that which will degrade the photography.

SECTION 17.**SYSTEM RELIABILITY**

Reliability calculations for the payload are based on a sample beginning with M-7. Hence both the major part of the Mural Program and the "J" Program are covered in the calculation. For certain auxiliaries, i.e., the stellar-index camera and the horizon cameras, the sample size is changed to recognize incorporation of modified equipment or new designs where reliability was one of the principal reasons for the modification. However, for primary mission function, the sample size is consistent with reliability reporting for the vehicle.

The reliability estimates of this section deal exclusively with the payload. Failures to achieve orbit or vehicle induced failures are thereby excluded. Recoveries before a complete mission has been completed are considered as full missions providing that early termination was caused by reasons not connected with payload operation. Film quality is not considered in the reliability estimate calculation. Hence, only electrical and mechanical functioning are considered.

The reliability estimate is also divided into primary and secondary functions. The primary functions are operation of the panoramic cameras, main camera door operation, operation of the payload clock, and recovery operations. The secondary mission functions are horizon camera operation excluding catastrophic open shutter failure mode, auxiliary data recording, and stellar-index camera operation. A summary of estimated reliability is shown in Tables 17-1.

Panoramic Camera Reliability

Sample Size - 126 opportunities to operate.

One failure - capping shutter on slave instrument on system M-7.

Assume - 3000 cycles per camera per mission.

Estimated Reliability = 98.7% at 50% confidence level.

Main Camera Door Reliability

Sample Size - 44 vehicles x 2 doors = 88 opportunities to operate

Estimated Reliability = 99.2% at 50% confidence level

~~TOP SECRET~~

No.

Payload Command and Control

Sample Size - 6720 hours operation in sample

Two failures

Estimated Reliability = 96.3% at 50% confidence level

Payload Clock Reliability

Sample Size - 6720 hours operation in sample

No failures

Estimated Reliability = 99.0% at 50% confidence level

Estimated Reliability of Payload Functioning on orbit = 96.5% at
50% confidence level

Recovery System Reliability

53 opportunities to recover

1 failure - improper separation due to water seal cutter failure

Estimated Reliability = 96.8% at 50% confidence level

Stellar-Index Camera Reliability

Sample begins with J5

Sample size = 14,740 cycles

Two failures

Estimated Reliability = 92.6 at 50% confidence level

Horizon Camera Reliability

Sample begins with J5 - 63,000 cycles

Estimated Reliability of Single Camera = 98.4% at 50% confidence level

Estimated Reliability of Four Horizon Cameras at a Parallel

Redundant System = 99.9% at 50% confidence level.

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TOP SECRET

NO.

ESTIMATED RELIABILITY SUMMARY

(AT 50% CONFIDENCE LEVEL)

MISSION NUMBER	PRIMARY FUNCTIONS	SECONDARY FUNCTIONS	
		RECOVERY SYSTEM	STELLAR - IMAGE GAMMA RAY
1000	PANORAMIC CAMERA COMMAND & CONTROL SYSTEM	PAYOUT CLOCK SAMPLE	HORIZON GAMMA RAY
1001	RELAYED RELAY	RELAY	SAMPLE
1002	RELAYED RELAY	RELAY	FALLOUT
1003	RELAYED RELAY	RELAY	HORIZON GAMMA RAY
1004	RELAYED RELAY	RELAY	RELAY
1005	RELAYED RELAY	RELAY	RELAY
1006	RELAYED RELAY	RELAY	RELAY
1007	RELAYED RELAY	RELAY	RELAY
1008	RELAYED RELAY	RELAY	RELAY
1009	RELAYED RELAY	RELAY	RELAY
1010	RELAYED RELAY	RELAY	RELAY
1011	RELAYED RELAY	RELAY	RELAY
1012	RELAYED RELAY	RELAY	RELAY
1013	RELAYED RELAY	RELAY	RELAY
1014	RELAYED RELAY	RELAY	RELAY
1015	RELAYED RELAY	RELAY	RELAY
1016	RELAYED RELAY	RELAY	RELAY
1017	RELAYED RELAY	RELAY	RELAY
1018	RELAYED RELAY	RELAY	RELAY
1019	RELAYED RELAY	RELAY	RELAY

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ESTIMATED RELIABILITY SUMMARY

(AT 50% CONFIDENCE LEVEL)

MISSION NUMBER	PRIMARY FUNCTIONS	SECONDARY FUNCTIONS		RECOVERY SYSTEM	ON - ORBIT FUNCTIONS	STELLAR - INDEX CAMERAS	HORIZON GAMMA RAY DETECTORS	
		SAMPLE	IMAGE					
1020	PANORAMIC CAMERA DOORS	COMMAND & CONTROL SYSTEM	PAYOUT CLOCK	SAMPLE	FALLOWS	FALLOWS	SAMPLE	FALLOWS
	SAMPLE	FALLOWS	FALLOWS	R544	96.9	96.9	R545	96.9
	FALLOWS	FALLOWS	FALLOWS	70	97.1	97.1	70	97.1
	RELIABILITY	RELIABILITY	RELIABILITY	8544	97.1	97.1	8545	97.1
1021				8576	97.0	97.0	8577	97.0
				8576	97.0	97.0	8577	97.0
				8784	97.0	97.0	8785	97.0
				8784	97.0	97.0	8785	97.0
1022				8600	97.1	97.1	8601	97.1
				8600	97.1	97.1	8601	97.1
				8540	97.2	97.2	8541	97.2
				8540	97.2	97.2	8541	97.2
1023				8440	97.3	97.3	8441	97.3
				8440	97.3	97.3	8441	97.3
				8440	97.3	97.3	8441	97.3
1024				8440	97.4	97.4	8441	97.4
				8440	97.4	97.4	8441	97.4
				8440	97.4	97.4	8441	97.4
1025				8440	97.5	97.5	8441	97.5
				8440	97.5	97.5	8441	97.5
				8440	97.5	97.5	8441	97.5
1026				8440	97.6	97.6	8441	97.6
				8440	97.6	97.6	8441	97.6
				8440	97.6	97.6	8441	97.6
1027				8440	97.7	97.7	8441	97.7
				8440	97.7	97.7	8441	97.7
				8440	97.7	97.7	8441	97.7

SECTION 18

SUMMARY DATA

The comparison of the operating parameters and the performance achieved by previous missions has been difficult due to the large volume of data that results from each mission. Some of the pertinent characteristics from prior missions have been summarized in Tables 18-1 through 18-3.

The summary data was started with Mission 1004 as the J-05 camera system was the first to incorporate the major modifications of the titanium drum and scan arm, four roller scan head and Corona J capabilities. Only those missions that culminated in the recovery of some photography have been listed, therefore Missions 1003 and 1005 are deleted.

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MISSION SUMMARY

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TABLE

MISSION NUMBER	PAYLOAD NUMBER	Vehicle Number	Launch Date	Latitude	Longitude	PERIGEE ALTITUDE (NM)	APOGEE ALTITUDE (NM)	RECOVERY PASS	MASTER CAMERA NUMBER	CAMERA SLIT FILTER TYPE ('')	MASTER CAMERA NUMBER	SLAVE CAMERA NUMBER	SLAVE CAMERA FILTER TYPE ('')	STELLAR - 11 CAMERA IMAGE		
														DATE	TIME	
1004	J-08	1174	8/18/64	21° 4' 4	74° 9	93.3	29.0	48	124	0.250	W-21	123	0.200	W-21	020/02/20	020/02/20
1005	J-09	1175	8/19/64	2239 2	79.9	84.0	63.2	60	148	0.200	W-21	149	0.200	W-21	049/02/21	049/02/21
1007	J-07	1009	8/19/64	2216 2	85.0	90.2	41.8	60	144	0.250	W-23	145	0.200	W-21	045/02/23	045/02/23
1008	J-10	1177	7/10/64	2314 2	84.0	89.4	40.8	60	112	0.200	W-21	161	0.200	W-21	048/02/16	021/02/16
1009	J-12	1006	8/19/64	2316 2	80.1	89.0	39.9	60	154	0.200	W-21	155	0.200	W-21	055/02/15	055/02/15
1010	J-11	1179	8/14/64	2256 2	84.9	87.4	42.5	60	152	0.175	W-21	193	0.175	W-21	041/01/19	044/01/19
1011	J-20	1170	8/19/64	2150 2	79.9	89.3	20.9	60	—	—	W-21	160	0.175	W-21	030/02/20	030/02/20
1012	J-13	1179	8/17/64	2202 2	73.0	80.2	38.4	60	156	0.200	W-21	187	0.200	W-21	051/01/17	051/01/17
1013	J-10	1173	11/2/64	2120 2	80.0	100.0	29.0	60	61	0.225	W-21	159	0.225	W-21	021/02/21	021/02/21
1014	J-16	1000	11/18/64	2056 2	70.0	103.2	65.0	61	162	0.250	W-23	139	0.175	W-21	059/02/09	059/02/09
1015	J-17	1007	12/18/64	2110 2	74.9	98.7	21.9	61	175	0.250	W-23	141	0.175	W-21	061/01/17	061/01/17
1016	J-19	1008	1/18/65	2101 2	74.9	93.4	20.2	61	160	0.250	W-23	133	0.175	W-21	059/02/09	059/02/09
1017	J-14	1011	2/23/65	2144 2	79.0	97.2	21.9	60	161	0.250	W-23	169	0.175	W-21	081/02/12	081/02/12
1018	J-19	1012	3/23/65	2111 2	84.0	100.2	40.3	60	95	0.250	W-23	123	0.175	W-21	029/02/23	029/02/23
1019	J-04	1014	4/23/65	2144 2	80.0	93.1	27.1	60	—	—	W-23	119	0.175	W-21	039/02/31	039/02/31
1020	J-03	1013	6/19/65	2156 2	79.1	97.1	40.9	57	115	0.200	W-23	137	0.175	W-21	067/02/02	067/02/02
1021	J-31	1015	5/18/65	1003 2	73.0	102.1	24.3	61	161	0.175	W-21	167	0.200	W-21	043/02/03	043/02/03
1022	J-26	1017	5/20/65	2201 2	85.0	98.7	30.3	60	144	0.250	W-23	169	0.175	W-21	063/02/23	063/02/23
1023	J-23	1018	5/17/65	2100 2	70.0	97.8	29.0	61	144	0.250	W-23	171	0.175	W-21	017/02/23	017/02/23
1024	J-24	1019	5/22/65	2131 2	80.0	95.3	19.4	61	161	0.250	W-23	173	0.175	W-21	063/02/23	063/02/23
1025	J-28	1016	5/23/65	1745 2	73.0	115.9	44.3	61	161	0.175	W-21	187	0.175	W-21	075/02/23	075/02/23
1026	J-25	1020	5/28/65	1117 2	73.0	93.0	17.0	61	174	0.250	W-23	175	0.175	W-21	075/02/23	075/02/23

TOP SECRET

No.

PERFORMANCE SUMMARY

MISSION NUMBER	CAMERA NUMBER	M I P VALUE	VISUAL RES.	AVG/SLIT		SLIT/AVERAGE		SLIT HIGH		SLIT LOW		SLIT ATTITUDE ENCL. (°)			202 ALTITUDE RATES (DEG) ROLL			202 YAW ROLL			202 YAW ENCL. (°)			202 ALTITUDE UNITS (SEC)			
				Y	X	Y	X	Y	X	Y	X	PITCH	ROLL	YAW	PITCH	ROLL	YAW	PITCH	ROLL	YAW	PITCH	ROLL	YAW	PITCH	ROLL	YAW	
1004-1	FWD	124	89	78	86	350	80	109	96	113	105	117	127	0.48	0.42	1.00	30.0	25.0	21.0	8.1	7.7	6.8	6.8	6.8	6.8	6.8	
1004-1	APT	125	88	76	84	350	81	108	97	112	107	117	124	0.48	0.42	1.00	30.0	25.0	21.0	8.1	7.7	6.8	6.8	6.8	6.8	6.8	
1005-1	FWD	146	90	78	74	350	65	88	87	92	84	97	104	0.48	0.42	1.14	28.3	28.3	27.8	19.4	19.4	19.4	19.4	19.4	19.4	19.4	
1005-1	APT	147	90	65	65	350	64	87	87	92	84	97	104	0.48	0.42	1.00	31.1	27.9	20.0	11.8	11.8	11.8	11.8	11.8	11.8	11.8	
1007-1	FWD	149	89	80	83	350	72	90	87	93	85	97	104	0.48	0.42	1.43	37.6	23.9	23.9	3.6	3.6	3.6	3.6	3.6	3.6	3.6	
1007-1	APT	150	89	80	85	350	72	93	93	97	85	97	104	0.48	0.42	0.47	—	43.0	25.0	—	4.2	4.2	4.2	4.2	4.2	4.2	4.2
1008-1	FWD	150	85	80	78	350	60	90	85	95	85	95	104	0.48	0.42	0.50	43.8	23.9	23.9	2.8	2.8	2.8	2.8	2.8	2.8	2.8	
1008-1	APT	151	85	82	79	350	61	93	85	95	85	95	104	0.48	0.42	0.50	42.9	24.0	24.0	32.5	32.5	32.5	32.5	32.5	32.5	32.5	
1009-1	FWD	154	84	82	84	350	65	90	87	95	87	95	104	0.48	0.42	0.50	33.6	23.9	23.9	27.6	27.6	27.6	27.6	27.6	27.6	27.6	
1009-1	APT	155	85	87	87	350	65	90	87	95	87	95	104	0.48	0.42	0.50	33.6	23.9	23.9	27.6	27.6	27.6	27.6	27.6	27.6	27.6	
1010-1	FWD	162	85	80	85	350	80	90	87	95	87	95	104	0.48	0.42	0.50	30.1	23.6	20.6	4.8	4.8	4.8	4.8	4.8	4.8	4.8	
1010-1	APT	163	85	82	80	350	82	90	87	95	87	95	104	0.48	0.42	0.50	30.1	23.6	20.7	4.8	4.8	4.8	4.8	4.8	4.8	4.8	
1011-1	FWD	160	90	84	84	350	76	97	80	95	87	97	104	0.48	0.42	0.50	37.1	23.9	23.9	31.1	31.1	31.1	31.1	31.1	31.1	31.1	
1011-1	APT	161	90	84	84	350	77	97	80	95	87	97	104	0.48	0.42	0.50	37.1	23.9	23.9	31.1	31.1	31.1	31.1	31.1	31.1	31.1	
1012-1	FWD	165	84	82	82	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	47.1	33.2	—	1.8	1.8	1.8	1.8	1.8	1.8	1.8
1012-1	APT	166	84	82	82	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	47.1	33.2	—	1.8	1.8	1.8	1.8	1.8	1.8	1.8
1012-1	FWD	167	86	82	82	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	47.1	33.2	—	1.8	1.8	1.8	1.8	1.8	1.8	1.8
1013-1	FWD	159	85	82	82	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	47.1	33.2	—	1.8	1.8	1.8	1.8	1.8	1.8	1.8
1014-1	FWD	162	80	87	82	350	85	87	80	85	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8
1014-1	APT	163	80	87	82	350	85	87	80	85	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8
1015-1	FWD	158	85	87	87	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8
1015-1	APT	159	85	87	87	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8
1016-1	FWD	132	85	85	85	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8
1016-1	APT	133	85	85	85	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8
1017-1	FWD	160	85	85	85	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8
1017-1	APT	161	85	85	85	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8
1018-1	FWD	122	85	85	85	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8
1018-1	APT	123	85	85	85	—	—	80	82	82	87	85	90	0.48	0.42	0.50	—	46.9	29.2	26.2	6.8	6.8	6.8	6.8	6.8	6.8	6.8

DATA NOT PRESENTLY AVAILABLE

TOP SECRET

Perspectiva

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PERFORMANCE SUMMARY

DATA NOT PRESENTLY AVAILABLE
TOP SECRET

TOP SECRET

No.

EXPOSURE - PROCESSING SUMMARY

TOP SECRET

TABLE

MISSION NUMBER	CAMERA	SOLAR ELEVATION (DEG)	ALTITUDE (K)	REPORTED PROCESSING			COMPUTED PROCESSING			TERRAIN D-MIN			CLOUD D-MAX			UNDER EXPOSED			OVER EXPOSED			CLOUD COVER (%)		
				LOW	HIGH	CL	LOW	HIGH	CL	LOW	HIGH	MEAN	LOW	HIGH	MEAN	LOW	HIGH	MEAN	LOW	HIGH	MEAN			
6004-1	FWD	-1	61	15	124	5	76	19	479	21	0.26	1.89	0.81	0.78	0.43	2.41	1.97	2.02	1.00	2.43	2.04	2.08		
6004-1	ART	-1	61	15	124	5	76	19	479	20	0.22	1.86	0.81	0.76	0.43	2.41	1.94	1.96	1.04	2.43	1.96	2.03		
6004-2	FWD	-1	61	10	151	7	14	17	37	30	13	4	0.83	1.3	0.89	0.80	0.63	0.78	0.36	2.30	0.84	1.27	1.87	
6004-2	ART	-1	61	10	151	7	76	17	37	30	13	4	0.83	1.3	0.89	0.81	0.63	0.78	0.36	2.30	0.84	1.27	1.87	
6004-1	FWD	-1	61	10	140	-	91	0	-31	48	0	31	0.93	1.81	0.71	0.68	0.80	2.31	1.99	1.92	2.46	1.99	1.98	
6004-1	ART	-1	61	10	140	-	91	0	-30	47	0	30	0.92	1.80	0.71	0.67	0.80	2.30	1.98	1.91	2.45	1.98	1.97	
6005-2	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6005-2	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6007-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6007-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6007-2	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6008-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6008-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6008-2	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6008-2	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6009-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6009-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6009-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6009-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6010-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6010-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6010-2	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6010-2	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6011-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6011-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6012-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6012-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6013-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6013-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6014-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6014-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6015-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6015-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6016-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6016-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6017-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6017-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6018-1	FWD	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15
6018-1	ART	-1	62	32	167	2	86	0	35	49	25	21	15	14	0.51	0.50	0.56	0.52	2.21	2.14	2.14	2.20	2.14	2.15

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EXPOSURE - PROCESSING SUMMARY

DATA NOT PRESENTLY AVAILABLE

~~TOP SECRET~~

No.

SECTION A

APPENDIX

~~TOP SECRET~~

TOP SECRET

MISSION - 1026-1 - INSTRUMENT - PRMD - 12/28/65 - DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY		INTERMEDIATE		TURBIDITY		ALL LEVELS		
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM
0.01	0	0	0	0	0	0	0	0	0
0.02	0	0	0	0	0	0	0	0	0
0.03	0	0	0	0	0	0	0	0	0
0.04	0	0	0	0	0	0	0	0	0
0.05	0	0	0	0	0	0	0	0	0
0.06	0	0	0	0	0	0	0	0	0
0.07	0	0	0	0	0	0	0	0	0
0.08	0	0	0	0	0	0	0	0	0
0.09	0	0	0	0	0	0	0	0	0
0.10	0	0	0	0	0	0	0	0	0
0.11	0	0	0	0	0	0	0	0	0
0.12	0	0	0	0	0	0	0	0	0
0.13	0	0	0	0	0	0	0	0	0
0.14	0	0	0	0	0	0	0	0	0
0.15	0	0	0	0	0	0	0	0	0
0.16	0	0	0	0	0	0	0	0	0
0.17	0	0	0	0	0	0	0	0	0
0.18	0	0	0	0	0	0	0	0	0
0.19	0	0	0	0	0	0	0	0	0
0.20	0	0	0	0	0	0	0	0	0
0.21	0	0	0	0	0	0	0	0	0
0.22	0	0	0	0	0	0	0	0	0
0.23	0	0	0	0	0	0	0	0	0
0.24	0	0	0	0	0	0	0	0	0
0.25	0	0	0	0	0	0	0	0	0
0.26	0	0	0	0	0	0	0	0	0
0.27	0	0	0	0	0	0	0	0	0
0.28	0	0	0	0	0	0	0	0	0
0.29	0	0	0	0	0	0	0	0	0
0.30	0	0	0	0	0	0	0	0	0
0.31	0	0	0	0	0	0	0	0	0
0.32	0	0	0	0	0	0	0	0	0
0.33	0	0	0	0	0	0	0	0	0
0.34	0	0	0	0	0	0	0	0	0
0.35	0	0	0	0	0	0	0	0	0
0.36	0	0	0	0	0	0	0	0	0
0.37	0	0	0	0	0	0	0	0	0
0.38	0	0	0	0	0	0	0	0	0
0.39	0	0	0	0	0	0	0	0	0
0.40	0	0	0	0	0	0	0	0	0
0.41	0	0	0	0	0	0	0	0	0
0.42	0	0	0	0	0	0	0	0	0
0.43	0	0	0	0	0	0	0	0	0
0.44	0	0	0	0	0	0	0	0	0
0.45	0	0	0	0	0	0	0	0	0
0.46	0	0	0	0	0	0	0	0	0
0.47	0	0	0	0	0	0	0	0	0
0.48	0	0	0	0	0	0	0	0	0
0.49	0	0	0	0	0	0	0	0	0
0.50	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0	0	0
	47	4	167	10	1	214	14		

TOP SECRET

- CONTROL NO.

TABLE A-1

~~TOP SECRET~~

- CONTROL NO.

TABLE A-1

MISSION • 1026-1 • INSTRUMENT • FRWD • 12/28/69 • DENSITY-FREQ DISTR

DENSITY LEVEL NUMBER	PRIMARY				INTERMEDIATE				FULL				ALL LEVELS				
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX
01	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
02	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
03	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
04	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
05	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
06	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
07	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
08	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
09	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
10	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
11	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
12	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
13	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
14	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
15	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
16	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
17	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
18	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
19	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
20	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
21	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
22	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
23	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
24	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
25	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
26	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
27	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
28	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
29	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
30	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
31	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
32	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
33	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
34	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
35	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
36	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
37	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
38	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
39	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
40	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
41	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
42	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
43	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
44	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
45	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
46	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
47	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
48	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
49	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
50	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0	0	0	24	12	2	66	27	2	90	39

- TOP SECRET

- CONTROL NO.

TABLE A-1

MISSION • 1026-1 • INSTRUMENT • FRWD 12/28/65 DENSITY FREQ DIST
DENSITY PRIMARY INTERMEDIATE FULL ALL LEVELS
VALUE MIN MAX LIM MIN MAX LIM MIN MAX LIM MIN MAX LIM
1.51
1.52
1.53
1.54
1.55
1.56
1.57
1.58
1.59
1.60
1.61
1.62
1.63
1.64
1.65
1.66
1.67
1.68
1.69
1.70
1.71
1.72
1.73
1.74
1.75
1.76
1.77
1.78
1.79
1.80
1.81
1.82
1.83
1.84
1.85
1.86
1.87
1.88
1.89
1.90
1.91
1.92
1.93
1.94
1.95
1.96
1.97
1.98
1.99
2.00
SUBTOTAL

~~TOP SECRET~~

- CONTROL NO.

TABLE A-1

TOP SECRET

- CONTROL NO.

TABLE A-1

MISSION • 1026-1 • INSTRUMENT • FRWD • 12/28/65 DENSITY FREQ DSTA

DENSITY • PRIMARY AND INTERMEDIATE • FREQUENCY FOR ALL LEVELS

VALUE MIN MAX LIM MIN MAX LIM MIN MAX LIM MIN MAX LIM

2.51	0	0	0	0	0	0	0	0	0	0	0	0
2.52	0	0	0	0	0	0	0	0	0	0	0	0
2.53	0	0	0	0	0	0	0	0	0	0	0	0
2.54	0	0	0	0	0	0	0	0	0	0	0	0
2.55	0	0	0	0	0	0	0	0	0	0	0	0
2.56	0	0	0	0	0	0	0	0	0	0	0	0
2.57	0	0	0	0	0	0	0	0	0	0	0	0
2.58	0	0	0	0	0	0	0	0	0	0	0	0
2.59	0	0	0	0	0	0	0	0	0	0	0	0
2.60	0	0	0	0	0	0	0	0	0	0	0	0
2.61	0	0	0	0	0	0	0	0	0	0	0	0
2.62	0	0	0	0	0	0	0	0	0	0	0	0
2.63	0	0	0	0	0	0	0	0	0	0	0	0
2.64	0	0	0	0	0	0	0	0	0	0	0	0
2.65	0	0	0	0	0	0	0	0	0	0	0	0
2.66	0	0	0	0	0	0	0	0	0	0	0	0
2.67	0	0	0	0	0	0	0	0	0	0	0	0
2.68	0	0	0	0	0	0	0	0	0	0	0	0
2.69	0	0	0	0	0	0	0	0	0	0	0	0
2.70	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0	0	0	0	0	0
TOTAL	0	0	0	59	59	57	184	184	171	243	243	228

MISSION 1026-1 INSTR - FRWD 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	59	3 PC	53 PC	41 PC	3 PC	100 PC
FULL	184	74 PC	0 PC	24 PC	2 PC	100 PC
ALL LEVELS	243	57 PC	13 PC	28 PC	2 PC	100 PC
PROCESS LEVEL	BASE + FOG	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	0.91 AND UP
INTERMED	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND UP
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.90	0.91-1.69	1.70 AND UP

TOP SECRET

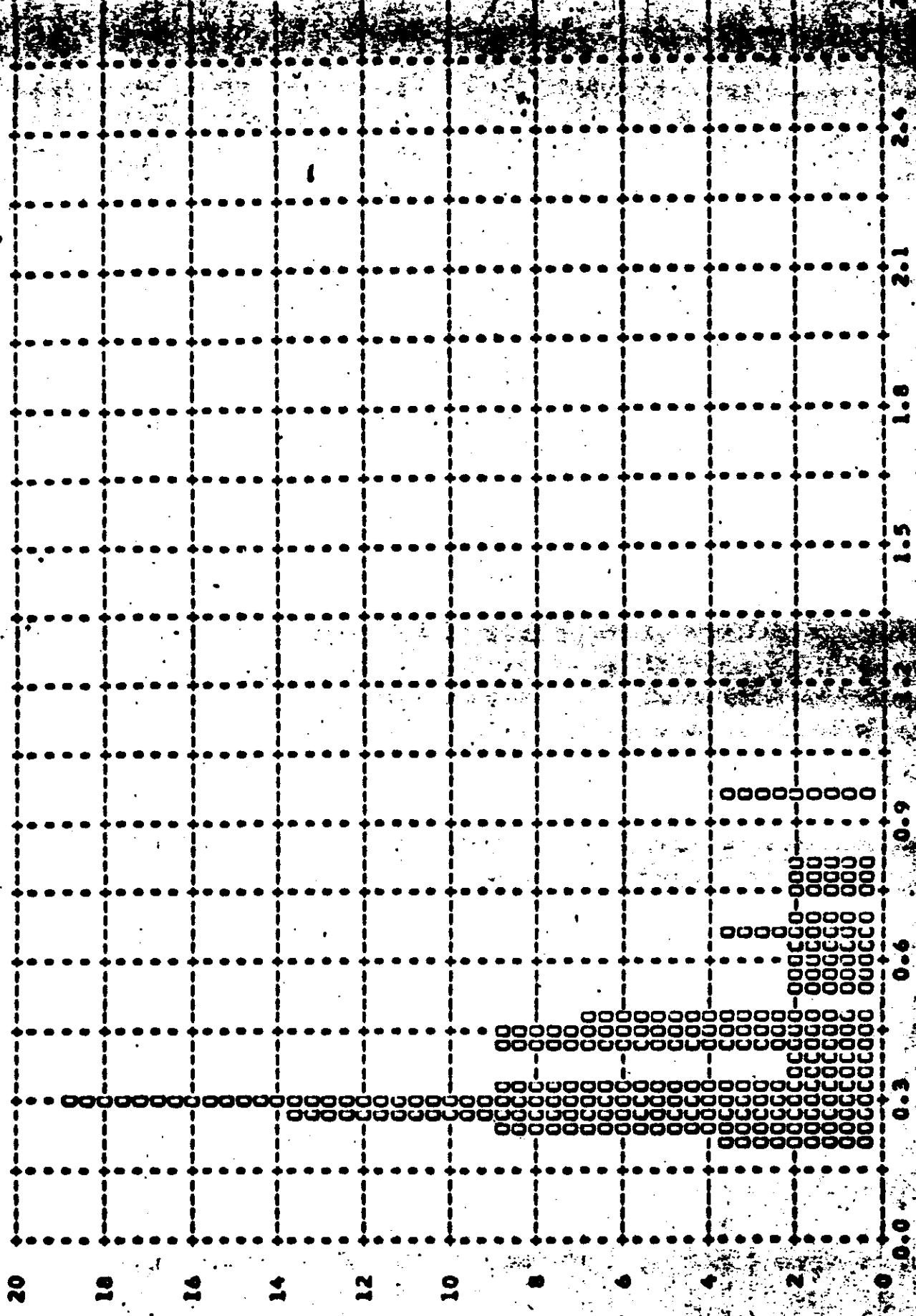
CONTROL NO. [REDACTED]

TABLE A-1

TOP SECRET

COPYRIGHT NO.

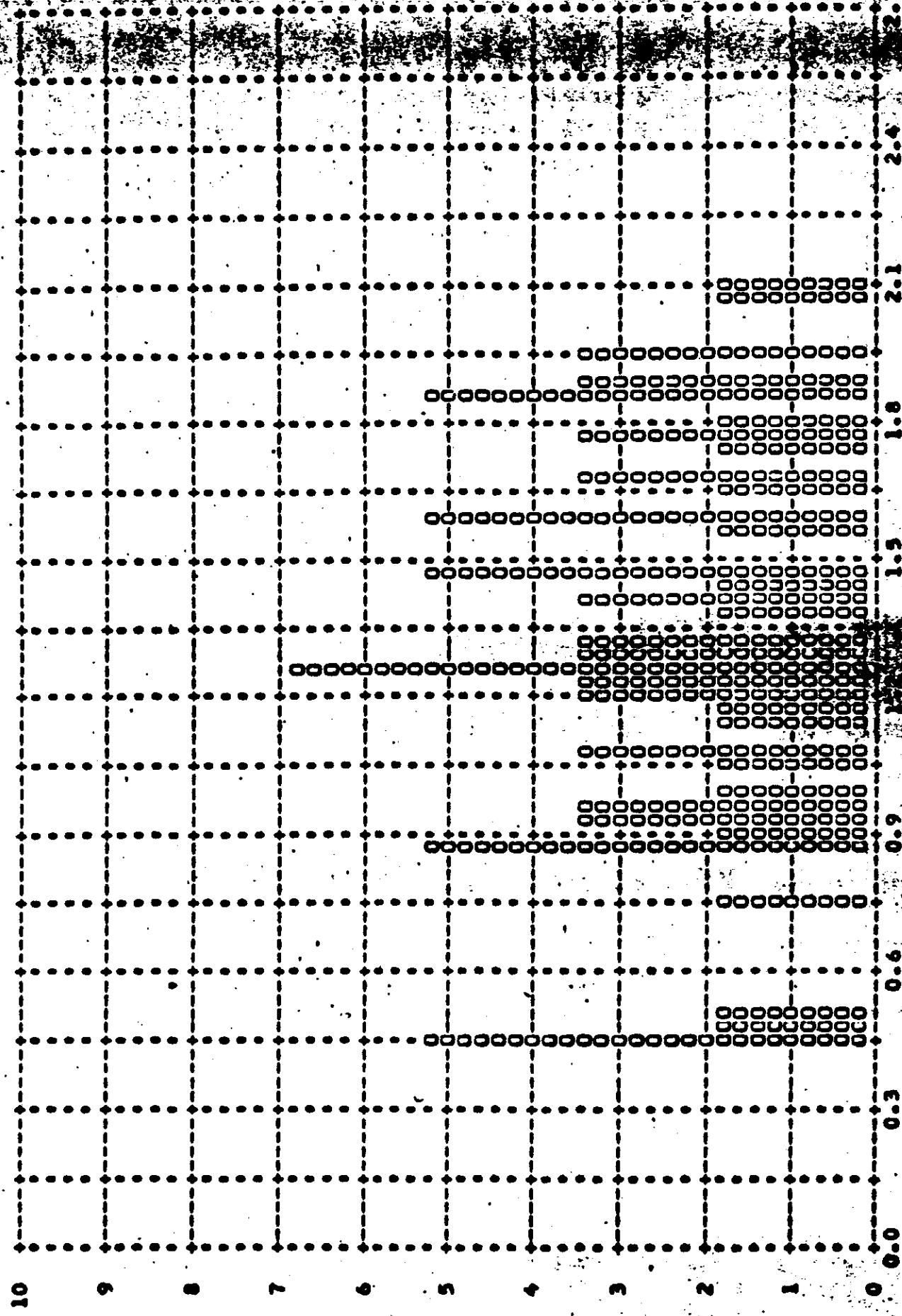
MISSION • 1026-1 • INSTR • FRND • 12/28/65 PLOT OF 0 MIN • TERRAIN • PROCESSING • INTERMEDIATE
ARITH MEAN • 0.40 • MEDIAN • 0.33 • STD DEV • 0.10 • RANGE • 0.20 TO 0.95 WITH 59 SAMPLES



TOP SECRET

- CONTROL NO.

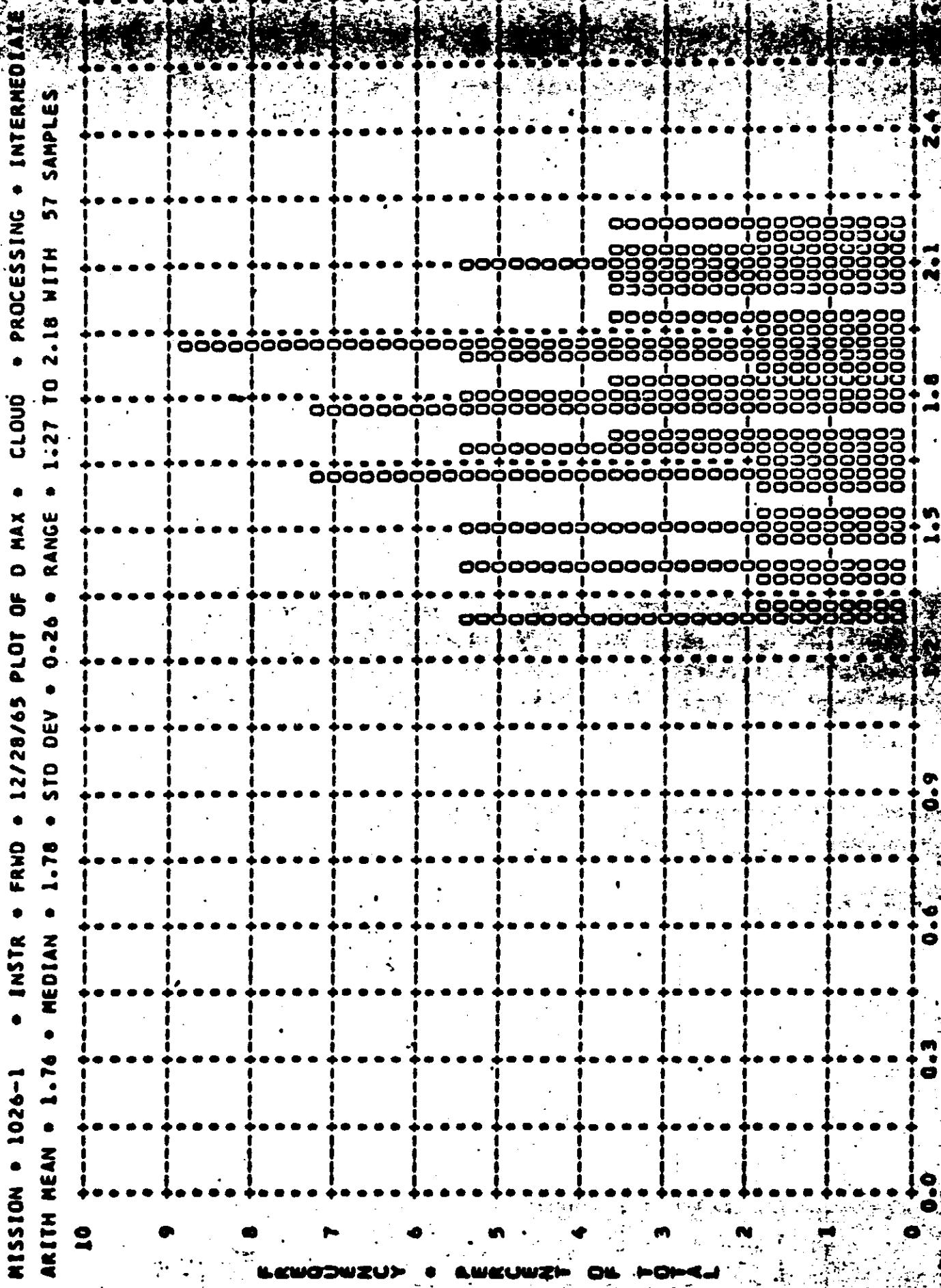
MISSION • 1026-1 • INSTR • FRWD • 12/28/65 PLOT OF D MAX • TERRAIN • PROCESSING • INTERMEDIATE
ARITH MEAN • 1.31 • MEDIAN • 1.28 • STD DEV • 0.43 • RANGE • 0.43 TO 2.08 WITH 59 SAMPLES



REMARKS: • REMARKS: ON FORM

TOP SECRET

CON (ROL NO.



TOP SECRET

- C4..ROL NO.

MISSION • 1026-1 • INSTK • FRWD • 12/28/65 PLOT OF D MIN • TERRAIN • PROCESSING • FULL
ARITH MEAN • 0.37 • MEDIAN • 0.33 • STD DEV • 0.14 • RANGE • 0.22 TO 1.26 WITH 184 SAMPLES

20

18

16

14

12

10

8

6

4

2

0

0.0

0.3

0.6

0.9

1.2

1.5

1.8

2.1

2.4

2.7

RECDWZUO • QWZUWZH OI, FOMED

TOP SECRET

- C. IROL NO. [REDACTED]

MISSION • 1026-1 • INSTR • FRWD • 12/28/65 PLOT OF D MAX • TERRAIN • PROCESSING • FULL
ARITH MEAN • 1.19 • MEDIAN • 1.18 • STD DEV • 0.46 • RANGE • 0.31 TO 2.27 WITH 184 SAMPLES

10

RECOMMEND • MEASURED ON POWER

A-11

0.0 0.3 0.6 0.9 1.2 1.5 1.8 2.1 2.4

INTENSITY

ACROSS

DOWN

TOP SECRET

FIGURE A-5

TOP SECRET

C. C. TROL NO.

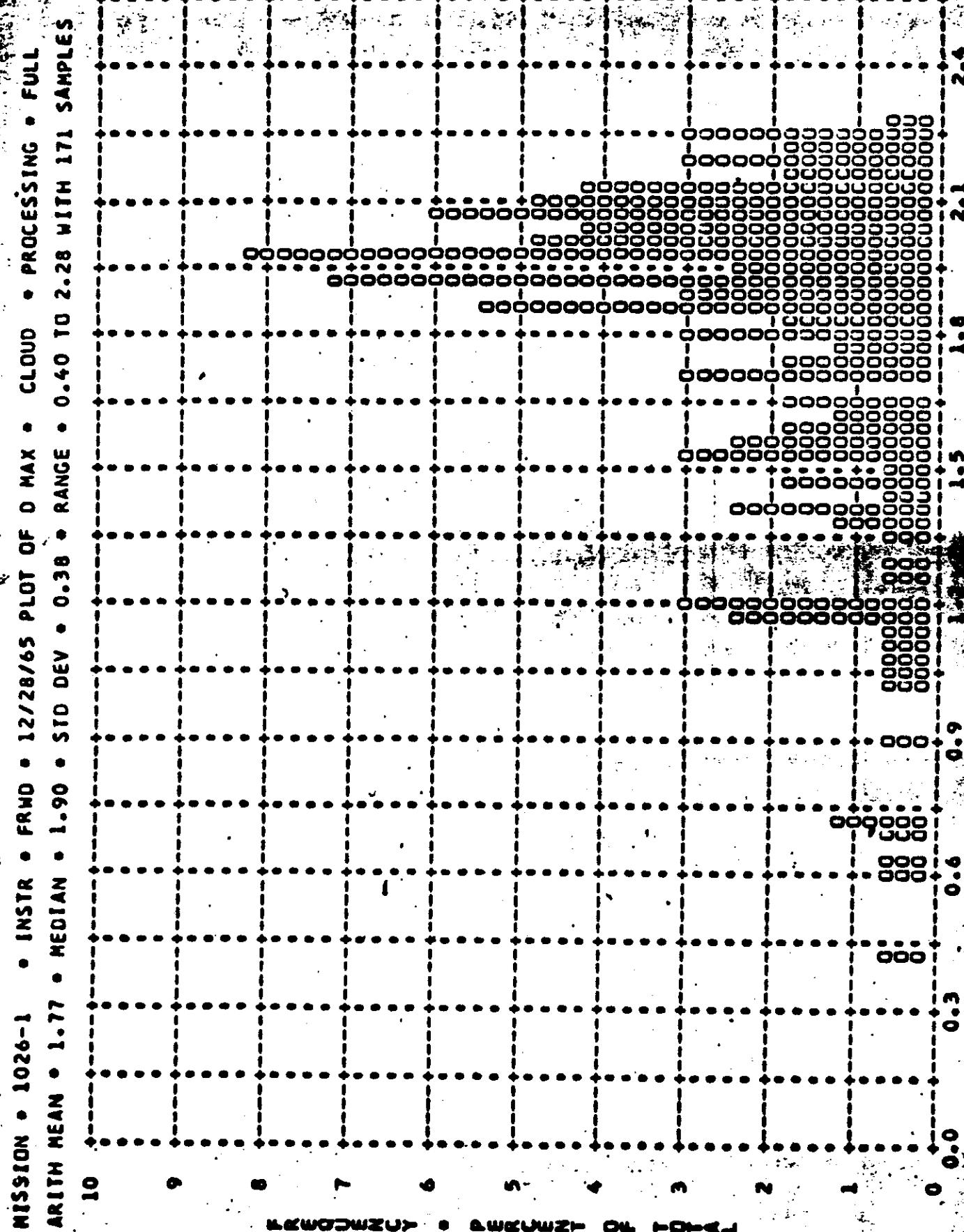


FIGURE A-6.

TOP SECRET

TOP SECRET

CC-401 MD.

MISSION • 1026-1 • INSTR • FRHO • 12/28/65 PLOT OF D MIN • TERRAIN • PROCESSING • ALL LEVELS
ARITH MEAN • 0.36 • MEDIAN • 0.33 • STD DEV • 0.15 • RANGE • 0.20 TO 1.26 WITH 243 SAMPLES

20

18

16

14

12

10

8

6

4

2

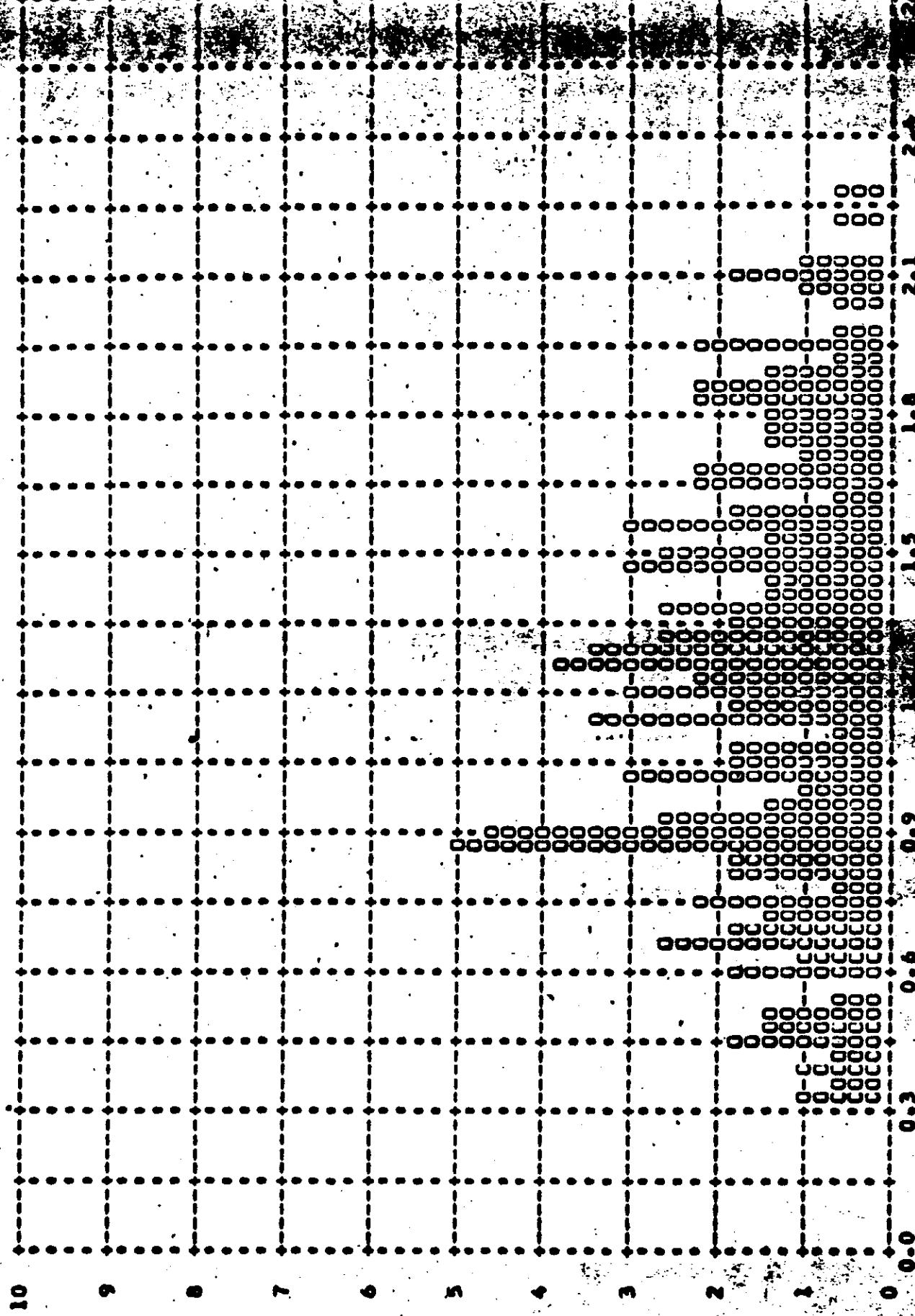
0

• 0.00000000 0.00000000 0.00000000

CONFIDENTIAL

TOP SECRET

MISSION • 1026-1 • INSTR • FRWD • 12/28/65 PLOT OF D MAX • TERRAIN • PROCESSING • ALL LEVELS
ARITH MEAN • 1.22 • MEDIAN • 1.22 • STD DEV • 0.45 • RANGE • 0.31 TO 2.27 WITH 243 SAMPLES



REF ID: A62427 - 001

-TOP SECRET

-CC-ROL NO.

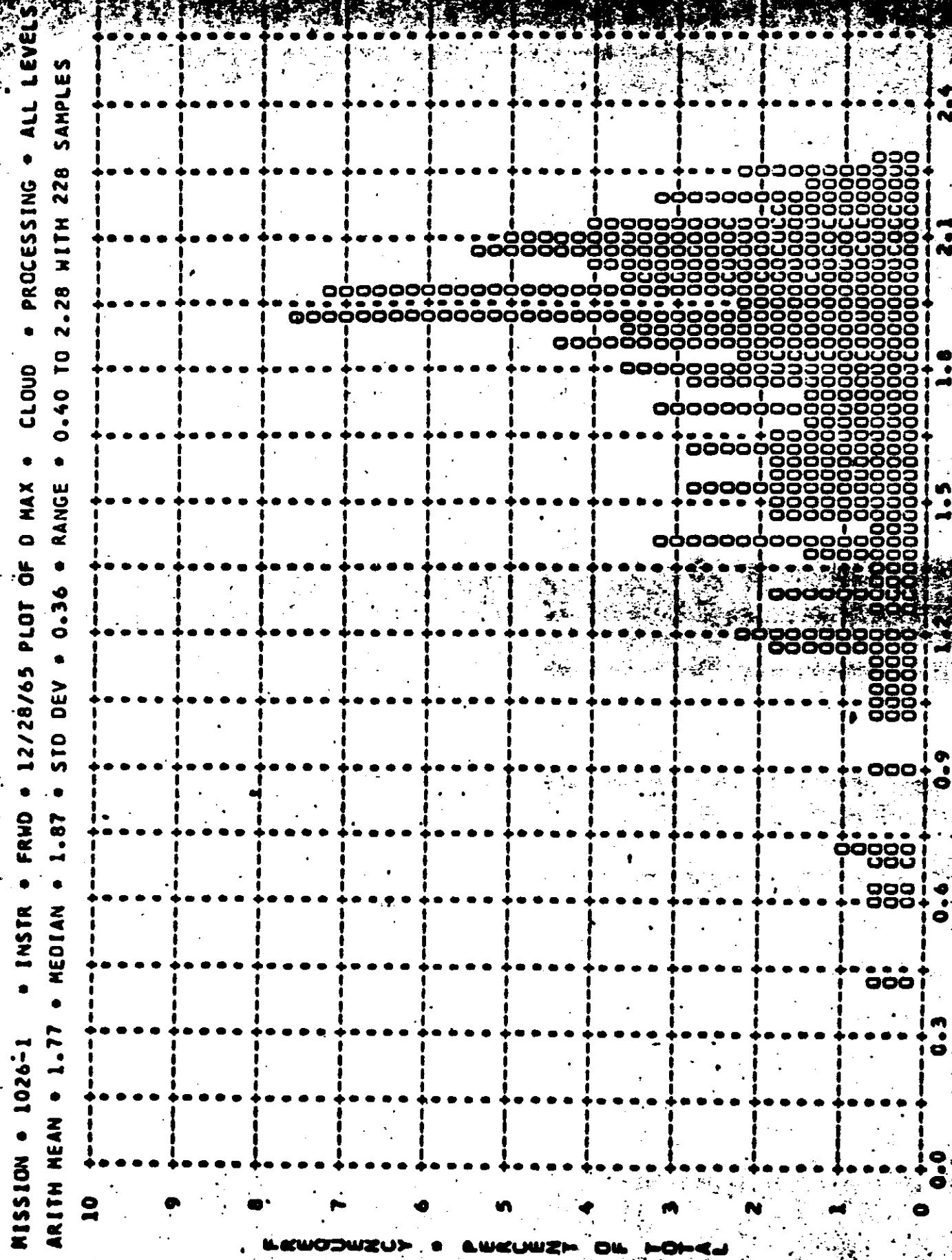


FIGURE A-8

-TOP SECRET

TOP SECRET

CONTROL NO.

MISSION - 1D26-1 INSTRUMENT - AET 12/28/69 DENSITY FREQ DISTA

DENSITY VALUE	PRIMARY			INTERMEDIATE			HIGH LEVEL			LOW LEVEL		
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM
0.01	000	000		000	000		000	000		000	000	
0.02	000	000		000	000		000	000		000	000	
0.03	000	000		000	000		000	000		000	000	
0.04	000	000		000	000		000	000		000	000	
0.05	000	000		000	000		000	000		000	000	
0.06	000	000		000	000		000	000		000	000	
0.07	000	000		000	000		000	000		000	000	
0.08	000	000		000	000		000	000		000	000	
0.09	000	000		000	000		000	000		000	000	
0.10	000	000		000	000		000	000		000	000	
0.11	000	000		000	000		000	000		000	000	
0.12	000	000		000	000		000	000		000	000	
0.13	000	000		000	000		000	000		000	000	
0.14	000	000		000	000		000	000		000	000	
0.15	000	000		000	000		000	000		000	000	
0.16	000	000		000	000		000	000		000	000	
0.17	000	000		000	000		000	000		000	000	
0.18	000	000		000	000		000	000		000	000	
0.19	000	000		000	000		000	000		000	000	
0.20	000	000		000	000		000	000		000	000	
0.21	000	000		000	000		000	000		000	000	
0.22	000	000		000	000		000	000		000	000	
0.23	000	000		000	000		000	000		000	000	
0.24	000	000		000	000		000	000		000	000	
0.25	000	000		000	000		000	000		000	000	
0.26	000	000		000	000		000	000		000	000	
0.27	000	000		000	000		000	000		000	000	
0.28	000	000		000	000		000	000		000	000	
0.29	000	000		000	000		000	000		000	000	
0.30	000	000		000	000		000	000		000	000	
0.31	000	000		000	000		000	000		000	000	
0.32	000	000		000	000		000	000		000	000	
0.33	000	000		000	000		000	000		000	000	
0.34	000	000		000	000		000	000		000	000	
0.35	000	000		000	000		000	000		000	000	
0.36	000	000		000	000		000	000		000	000	
0.37	000	000		000	000		000	000		000	000	
0.38	000	000		000	000		000	000		000	000	
0.39	000	000		000	000		000	000		000	000	
0.40	000	000		000	000		000	000		000	000	
0.41	000	000		000	000		000	000		000	000	
0.42	000	000		000	000		000	000		000	000	
0.43	000	000		000	000		000	000		000	000	
0.44	000	000		000	000		000	000		000	000	
0.45	000	000		000	000		000	000		000	000	
0.46	000	000		000	000		000	000		000	000	
0.47	000	000		000	000		000	000		000	000	
0.48	000	000		000	000		000	000		000	000	
0.49	000	000		000	000		000	000		000	000	
0.50	000	000		000	000		000	000		000	000	
SUBTOTAL	000	000		000	000		000	000		000	000	
	79	5		1	119	7	198	12				

TOP SECRET

CONTROL NO.

TABLE A-2

~~TOP SECRET~~

• CONTROL NO.

TABLE A-2

Subtotal

~~TOP SECRET~~

CONTROL NO.

TABLE A-2

MISSION - 1026-1 INSTRUMENTS AF 12/28/62 DENSITY FREQ DISTR

DENSITY VALUE	PRIMARY			INTERMEDIATE			THERMAL			TOTAL		
	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM	MIN	MAX	LIM
1.51	0	0	0	0	0	0	0	0	0	0	0	0
1.52	0	0	0	0	0	0	0	0	0	0	0	0
1.53	0	0	0	0	0	0	0	0	0	0	0	0
1.54	0	0	0	0	0	0	0	0	0	0	0	0
1.55	0	0	0	0	0	0	0	0	0	0	0	0
1.56	0	0	0	0	0	0	0	0	0	0	0	0
1.57	0	0	0	0	0	0	0	0	0	0	0	0
1.58	0	0	0	0	0	0	0	0	0	0	0	0
1.59	0	0	0	0	0	0	0	0	0	0	0	0
1.60	0	0	0	0	0	0	0	0	0	0	0	0
1.61	0	0	0	0	0	0	0	0	0	0	0	0
1.62	0	0	0	0	0	0	0	0	0	0	0	0
1.63	0	0	0	0	0	0	0	0	0	0	0	0
1.64	0	0	0	0	0	0	0	0	0	0	0	0
1.65	0	0	0	0	0	0	0	0	0	0	0	0
1.66	0	0	0	0	0	0	0	0	0	0	0	0
1.67	0	0	0	0	0	0	0	0	0	0	0	0
1.68	0	0	0	0	0	0	0	0	0	0	0	0
1.69	0	0	0	0	0	0	0	0	0	0	0	0
1.70	0	0	0	0	0	0	0	0	0	0	0	0
1.71	0	0	0	0	0	0	0	0	0	0	0	0
1.72	0	0	0	0	0	0	0	0	0	0	0	0
1.73	0	0	0	0	0	0	0	0	0	0	0	0
1.74	0	0	0	0	0	0	0	0	0	0	0	0
1.75	0	0	0	0	0	0	0	0	0	0	0	0
1.76	0	0	0	0	0	0	0	0	0	0	0	0
1.77	0	0	0	0	0	0	0	0	0	0	0	0
1.78	0	0	0	0	0	0	0	0	0	0	0	0
1.79	0	0	0	0	0	0	0	0	0	0	0	0
1.80	0	0	0	0	0	0	0	0	0	0	0	0
1.81	0	0	0	0	0	0	0	0	0	0	0	0
1.82	0	0	0	0	0	0	0	0	0	0	0	0
1.83	0	0	0	0	0	0	0	0	0	0	0	0
1.84	0	0	0	0	0	0	0	0	0	0	0	0
1.85	0	0	0	0	0	0	0	0	0	0	0	0
1.86	0	0	0	0	0	0	0	0	0	0	0	0
1.87	0	0	0	0	0	0	0	0	0	0	0	0
1.88	0	0	0	0	0	0	0	0	0	0	0	0
1.89	0	0	0	0	0	0	0	0	0	0	0	0
1.90	0	0	0	0	0	0	0	0	0	0	0	0
1.91	0	0	0	0	0	0	0	0	0	0	0	0
1.92	0	0	0	0	0	0	0	0	0	0	0	0
1.93	0	0	0	0	0	0	0	0	0	0	0	0
1.94	0	0	0	0	0	0	0	0	0	0	0	0
1.95	0	0	0	0	0	0	0	0	0	0	0	0
1.96	0	0	0	0	0	0	0	0	0	0	0	0
1.97	0	0	0	0	0	0	0	0	0	0	0	0
1.98	0	0	0	0	0	0	0	0	0	0	0	0
1.99	0	0	0	0	0	0	0	0	0	0	0	0
2.00	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	19	60	42	53	61	113						

TOP SECRET

CONTROL NO.

TABLE A-2

MISSION • 1026-1 • INSTRUMENT • AFT • 12/20/63 • DENSITY FREQ DISTR.

DENSITY PRIMARY INTERMEDIATE TERTIARY
VALUE MIN MAX LIM MIN MAX LIM MIN MAX LIM MIN MAX LIM

2
01
02
03
04
05
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76
77
78
79
80
81
82
83
84
85
86
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88
89
90
91
92
93
94
95
96
97
98
99
SUBTOTAL

~~TOP SECRET~~

- CONTROL NO.

TABLE A-2

MISSION - 1026-1	INSTRUMENT - AFT	DATE - 12/28/65	DENSITY FREQ DIST											
			DENSITY		PRIMARY		INTERMEDIATE		SECONDARY		TERTIARY		QUADRATIC	
VALUE	MIN	MAX	L1%	MIN	MAX	L1%	MIN	MAX	L1%	MIN	MAX	L1%	MIN	MAX
2.51	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.52	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.53	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.54	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.55	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.56	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.57	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.58	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.59	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.60	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.61	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.62	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.63	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.64	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.65	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.66	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.67	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.68	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.69	0	0	0	0	0	0	0	0	0	0	0	0	0	0
2.70	0	0	0	0	0	0	0	0	0	0	0	0	0	0
SUBTOTAL	0	0	0	0	0	0	0	0	0	0	0	0	0	0
TOTAL	0	0	0	90	90	95	150	150	130	240	240	225		

MISSION 1026-1 INSTR - AFT 12/28/65 PROCESSING AND EXPOSURE ANALYS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	90	0 PC	71 PC	27 PC	2 PC	
FULL	150	63 PC	0 PC	37 PC	1 PC	
ALL LEVELS	240	39 PC	27 PC	33 PC	1 PC	
PROCESS LEVEL	BASE + FOG	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	0.91 AND U
INTERMED	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND U
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.90	0.91-1.69	1.70 AND U

TOP SECRET

CONTROL NO.

TABLE A-2

TOP SECRET

CONTROL NO.

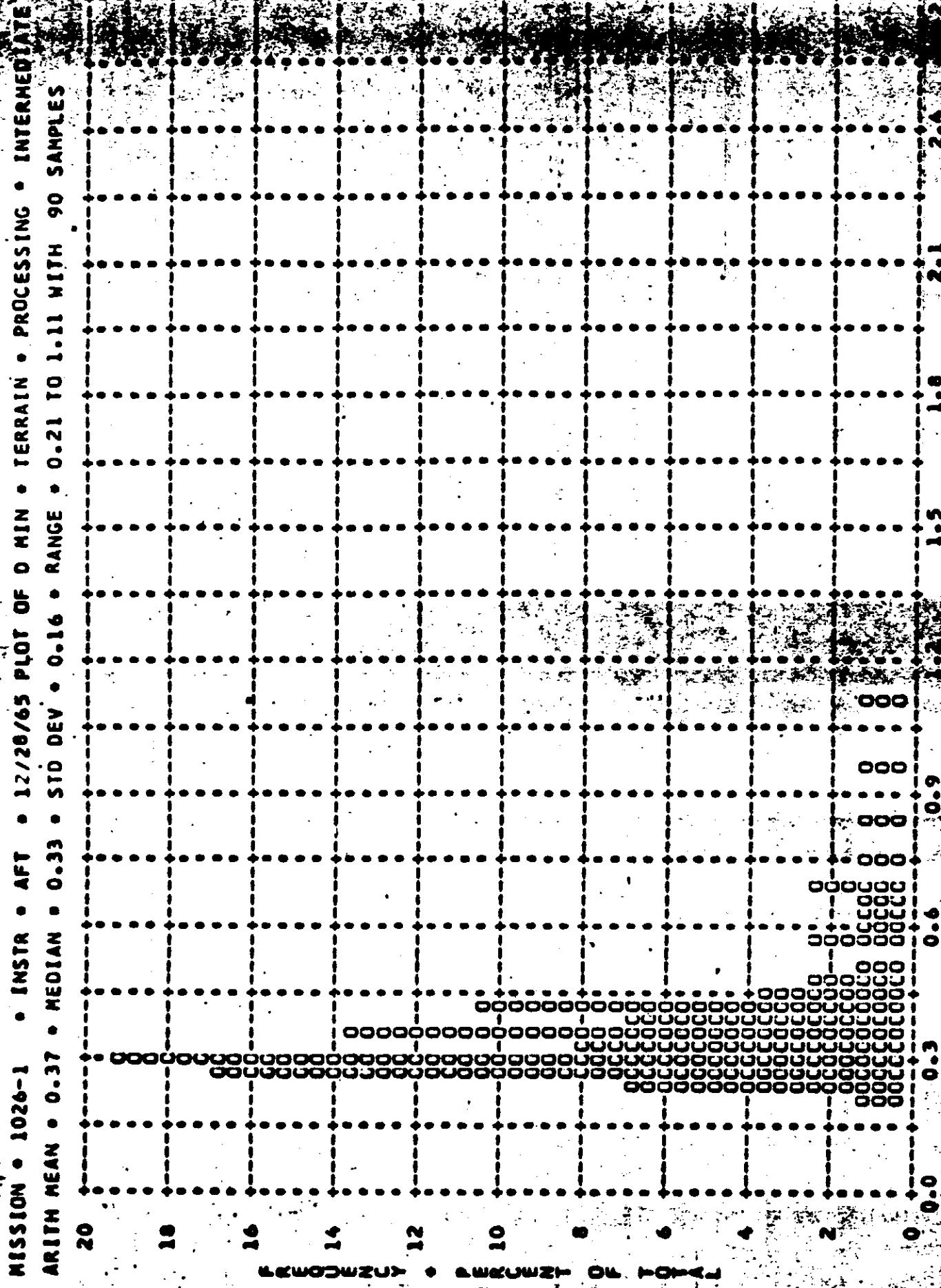


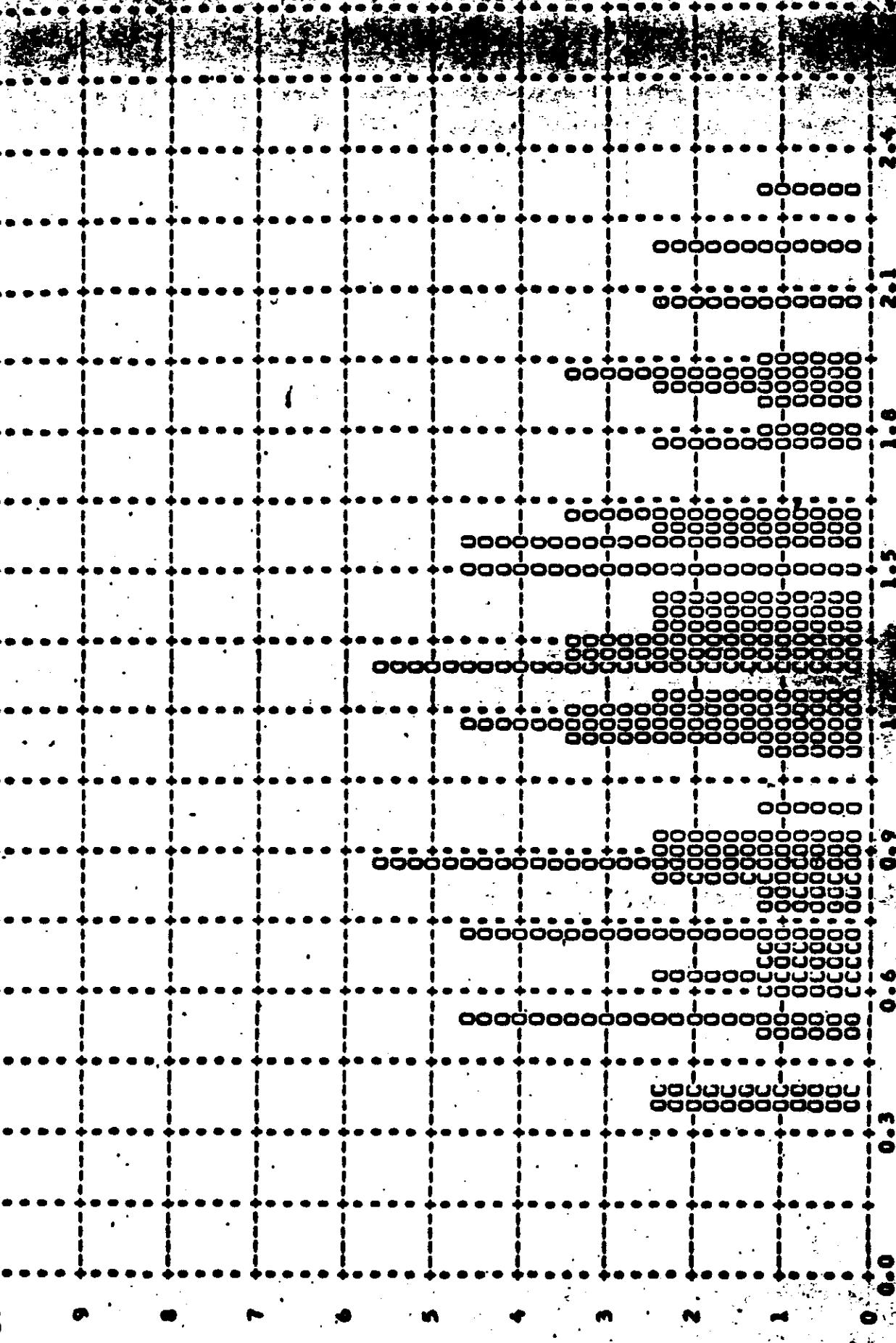
FIGURE A-10

TOP SECRET

- CONTROL NO. [REDACTED]

MISSION • 1026-1 • INSTR • AFT • 12/28/65 PLOT OF D MAX • TERRAIN • PROCESSING • INTERMEDIATE
ARITH MEAN • 1.22 • MEDIAN • 1.27 • STD DEV • 0.48 • RANGE • 0.35 TO 2.31 WITH .90 SAMPLES

10



LEADERWUR • LEADERWUR ON HOWL

FIGURE A-11

TOP SECRET

MISSION • 1026-1 • INSTR • AFT • 12/28/65 PILOT OF D MAX • CLOUD • PROCESSING • INTERMEDIATE
ARITH MEAN • 1.74 • MEDIAN • 1.82 • STD DEV • 0.37 • RANGE • 0.37 TO 2.32 WITH 95 SAMPLES

10

• 7 • 6 • 5 • 4 • 3 • 2 • 1 • 0 •

WATERWELL • SWELLWELL OIL POND

A-24

FIGURE A-12

TOP SECRET

CONT ROLL NO.

MISSION • 1026-1 • INSTR • AFT • 12/28/65 PLOT OF U MIN • TERRAIN • PROCESSING • FULL
ARITH MEAN • 0.41 • MEDIAN • 0.35 • STD DEV • 0.16 • RANGE • 0.20 TO 1.16 WITH 150 SAMPLES

20

18 16 14 12 10 8 6 4 2 0

UWODWZU • UWODWZU ON PAPER

A-25

1.5 1.6 1.8 2.0 2.1 2.2
0.6 0.7 0.8 0.9 0.0 0.3

FIGURE A-13

-TOP SECRET

- CONTROL NO.

MISSION • 1026-1 • INSTR • AFT • 12/28/65 PLOT OF D MAX • TERRAIN • PROCESSING • FULL
ARITH MEAN • 1.33 • MEDIAN • 1.35 • STD DEV • 0.47 • RANGE • 0.24 TO 2.36 WITH 150 SAMPLES

.10

0.0 0.1 0.2 0.3 0.4 0.5 0.6 0.7 0.8 0.9 1.0 1.1 1.2 1.3 1.4 1.5 1.6 1.7 1.8 1.9 2.0 2.1 2.2

- CONTROL NO.

MISSION • 1026-1 • INSTR • AFT • 12/28/65 PLOT OF D MAX • CLOUD • PROCESSING • FULL
ARITH MEAN • 1.81 • MEDIAN • 1.91 • STD DEV • 0.41 • RANGE • 0.64 TO 2.40 WITH 130 SAMPLES

10

DATA SOURCE • EQUATORIAL OCEAN

A-27

FIGURE A-15

- CONTROL NO.

TOECAST

TOP SECRET

CO., ROL NO.

MISSION • 1026-1 • INSTR • AFT • 12/28/65 PLOT OF 0 MIN • TERRAIN • PROCESSING • ALL LEVELS
ARITH MEAN • 0.39 • MEDIAN • 0.34 • STD DEV • 0.16 • RANGE • 0.20 TO 1.16 WITH 240 SAMPLES

20

18

16

14

12

10

8

6

4

2

0

0.00000000 • 0.00000000 HOURS

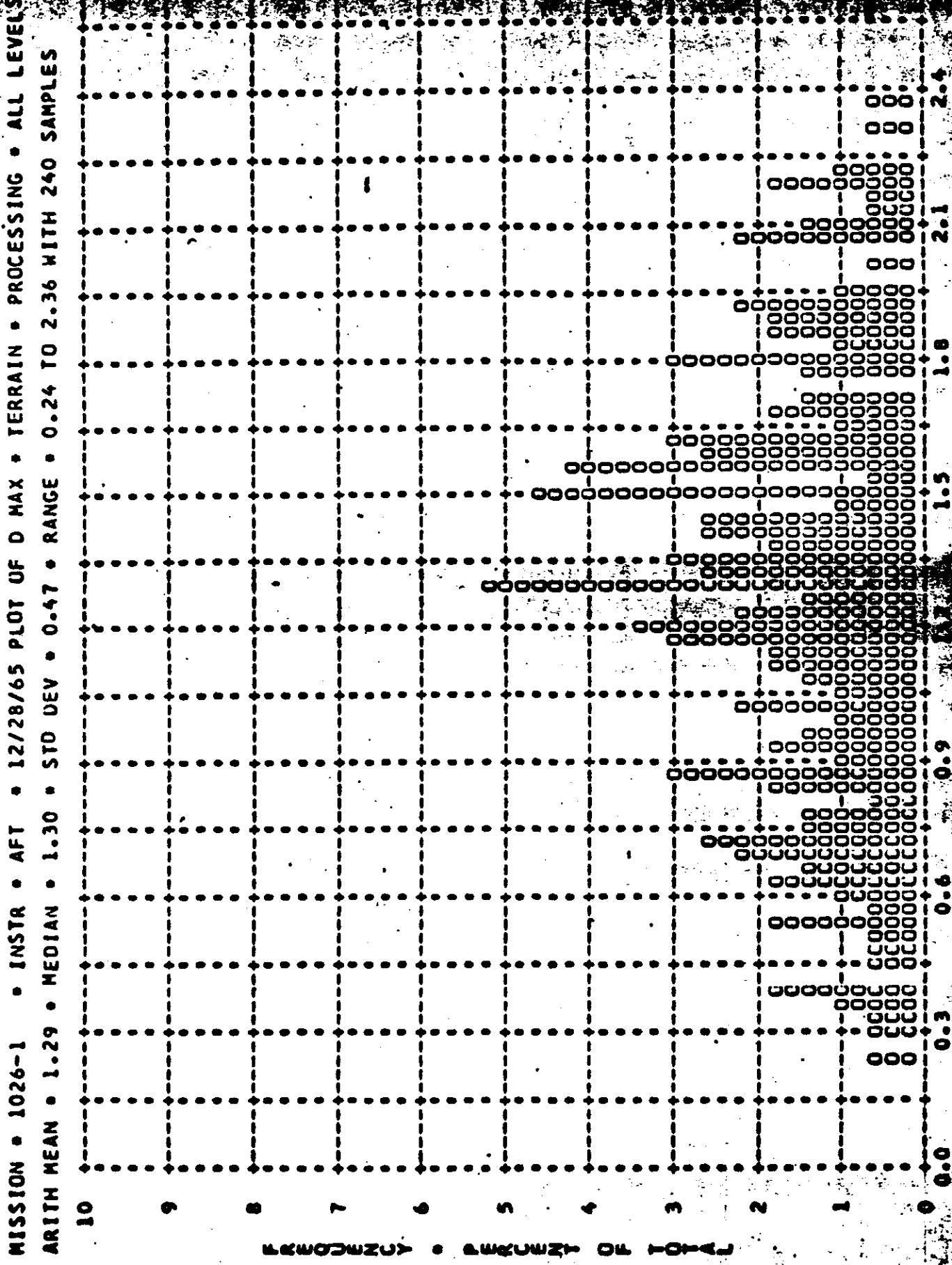
A-28

2.7
2.6
2.5
2.4
2.3
2.2
2.1
2.0
1.9
1.8
1.7
1.6
1.5
1.4
1.3
1.2
1.1
1.0
0.9
0.8
0.7
0.6
0.5
0.4
0.3
0.2
0.1
0.0

FIGURE A-16

TOP SECRET

- CONTROL NO. [REDACTED]



TOP SECRET

CONTROLLING

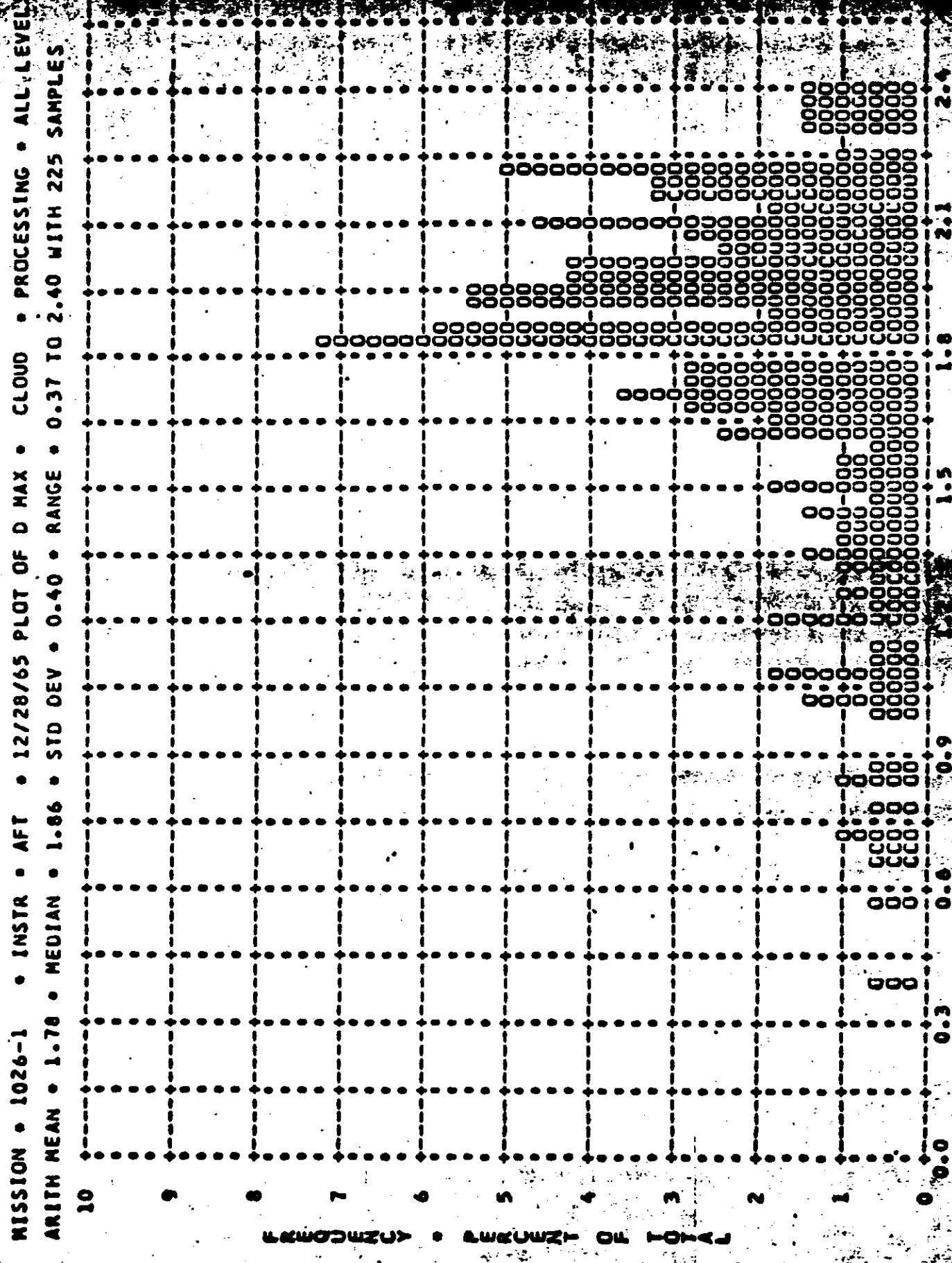


FIGURE A-18

~~TOP SECRET~~

CONTROL NO.

TABLE A-3

MISSION • 1026-2 • INSTRUMENT • FRWD • 12/26/63 • DENSITY FREQ DISTR.

DENSITY PRIMARY SINTERED STATE
VALUE MIN MAX LIM. MIN MAX LIM. MIN MAX LIM.

~~TOP SECRET~~

CONTROL NO.

TABLE A-3

TOP SECRET

CONTROL NO.

TABLE A-3

MISSION #: 102 - INSTRUMENT #: FAUD - 12/20/65 - DENSITY FREQ. DIGIT.

DENSITY	INTERMEDIATE LINE	HIGH MAX LINE	LOW MAX LINE	LINE	HIGH MAX LINE	LOW MAX LINE
1.51						
1.52						
1.53						
1.54						
1.55						
1.56						
1.57						
1.58						
1.59						
1.60						
1.61						
1.62	3					
1.63	4					
1.64	5					
1.65	6					
1.66	7					
1.67	8					
1.68	9					
1.69	10					
1.70	11					
1.71	12					
1.72	13					
1.73	14					
1.74	15					
1.75	16					
1.76	17					
1.77	18					
1.78	19					
1.79	20					
1.80	21					
1.81	22					
1.82	23					
1.83	24					
1.84	25					
1.85	26					
1.86	27					
1.87	28					
1.88	29					
1.89	30					
1.90	31					
1.91	32					
1.92	33					
1.93	34					
1.94	35					
1.95	36					
1.96	37					
1.97	38					
1.98	39					
1.99	40					
2.00	41					
SUBTOTAL	124	34	12	12	34	12

-TOP SECRET-

CONTROL NO.

TABLE A-3

MISSION • 1026-2 • INSTRUMENT • FRWD • 12/28/65 • DENSITY FREQ DIST

DENSITY PRIMARY INTERMEDIATE DEEP TOTAL
VALUE MIN MAX LIM MIN MAX LIM MIN MAX LIM MIN MAX LIM

-TOP SECRET-

CONTROL NO.

TABLE A-3

MISSION	026	INSTRUMENT	FREQ	12/28/65	DENSITY	FREQ DIST
VALUE	MIN	MAX	CNT	MIN MAX	MEAN	MIN MAX
.51	0	0	0	0	0	0
.52	0	0	0	0	0	0
.53	0	0	0	0	0	0
.54	0	0	0	0	0	0
.55	0	0	0	0	0	0
.56	0	0	0	0	0	0
.57	0	0	0	0	0	0
.58	0	0	0	0	0	0
.59	0	0	0	0	0	0
.60	0	0	0	0	0	0
.61	0	0	0	0	0	0
.62	0	0	0	0	0	0
.63	0	0	0	0	0	0
.64	0	0	0	0	0	0
.65	0	0	0	0	0	0
.66	0	0	0	0	0	0
.67	0	0	0	0	0	0
.68	0	0	0	0	0	0
.69	0	0	0	0	0	0
.70	0	0	0	0	0	0
SUBTOTAL						
TOTAL	0	0	0	12	12	26
				224	224	215
				236	236	241

MISSION 1026-2 INSTR - FRWD 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	12	0 PC	33 PC	58 PC	83 PC	100 PC
FULL	224	82 PC	0 PC	17 PC	1 PC	0 PC
ALL LEVELS	236	78 PC	2 PC	19 PC	1 PC	0 PC

PROCESS LEVEL	BASE + FOG	UNDER EXPUSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0.01-0.09	0.01-0.13	0.14-0.39	0.40-0.90	-----	0.91 AND UP
INTERMED	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND UP
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.90	0.91-1.69	1.70 AND UP

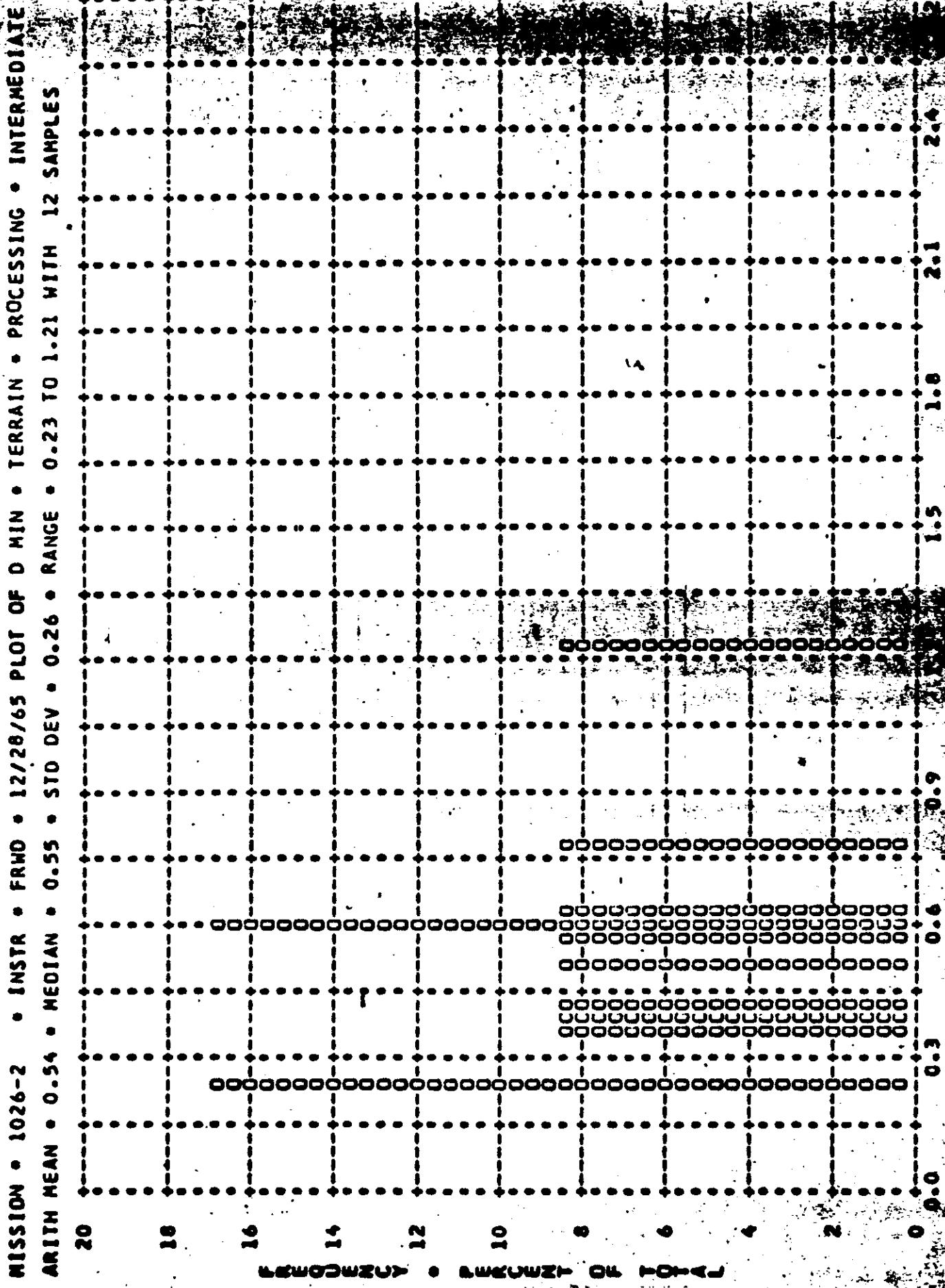
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CONTROL NO.

TABLE A-3

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UNIFROL NO.



TOP SECRET

- CL. / ROL. NO.

MISSION • 1026-2 • INSTR • FRWD • 12/28/65 PLOT OF U MAX • TERRAIN • PROCESSING • INTERMEDIATE
ARITH MEAN • 1.32 • MEDIAN • 1.50 • STD DEV • 0.35 • RANGE • 0.65 TO 1.79 WITH 12 SAMPLES

20

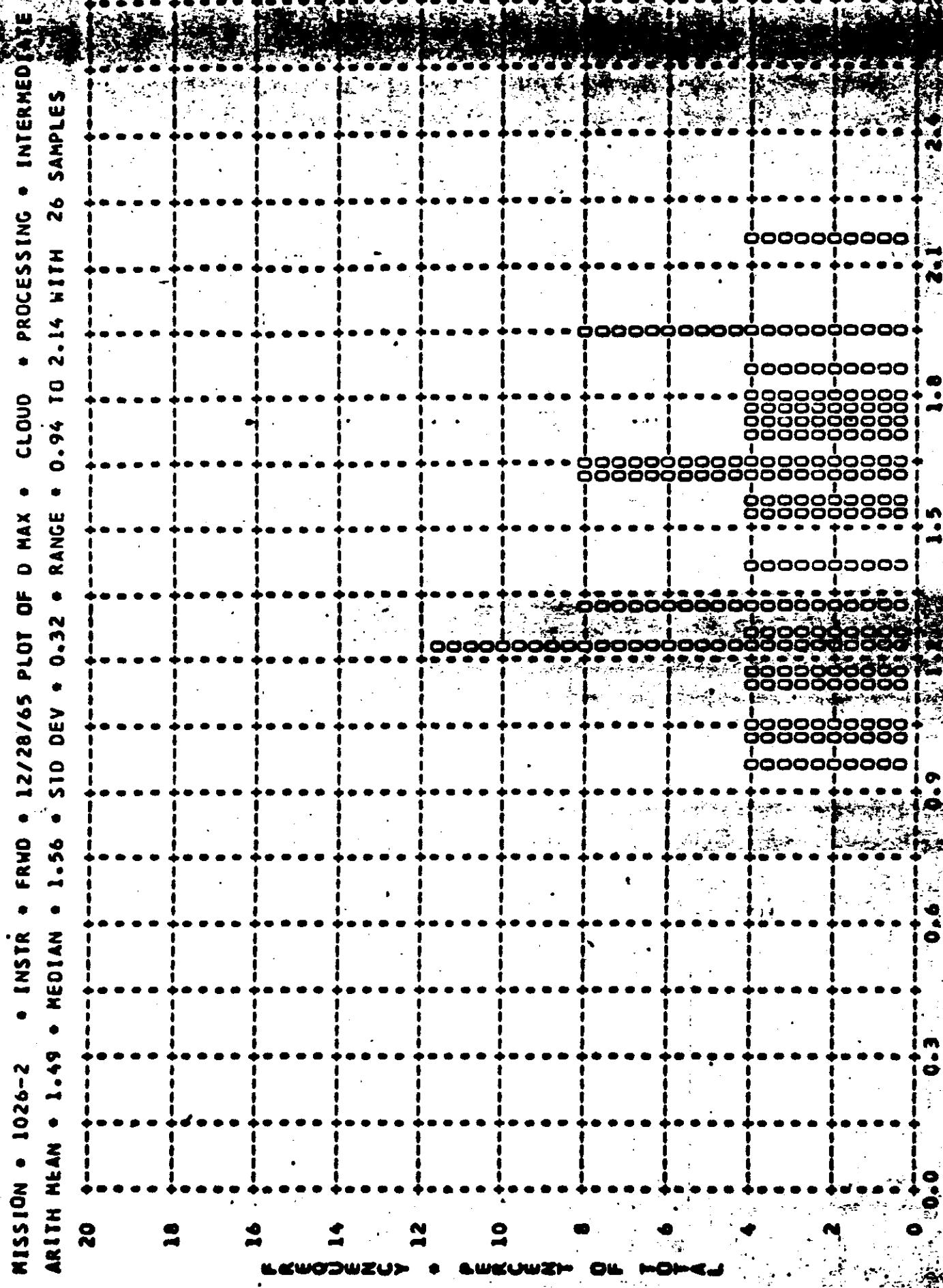
18 16 14 12 10 8 6 4 2 0
MAXIMUM TO MINIMUM • 100% -> 100% ← MINIMUM TO MAXIMUM

FIGURE A-20

TOP SECRET

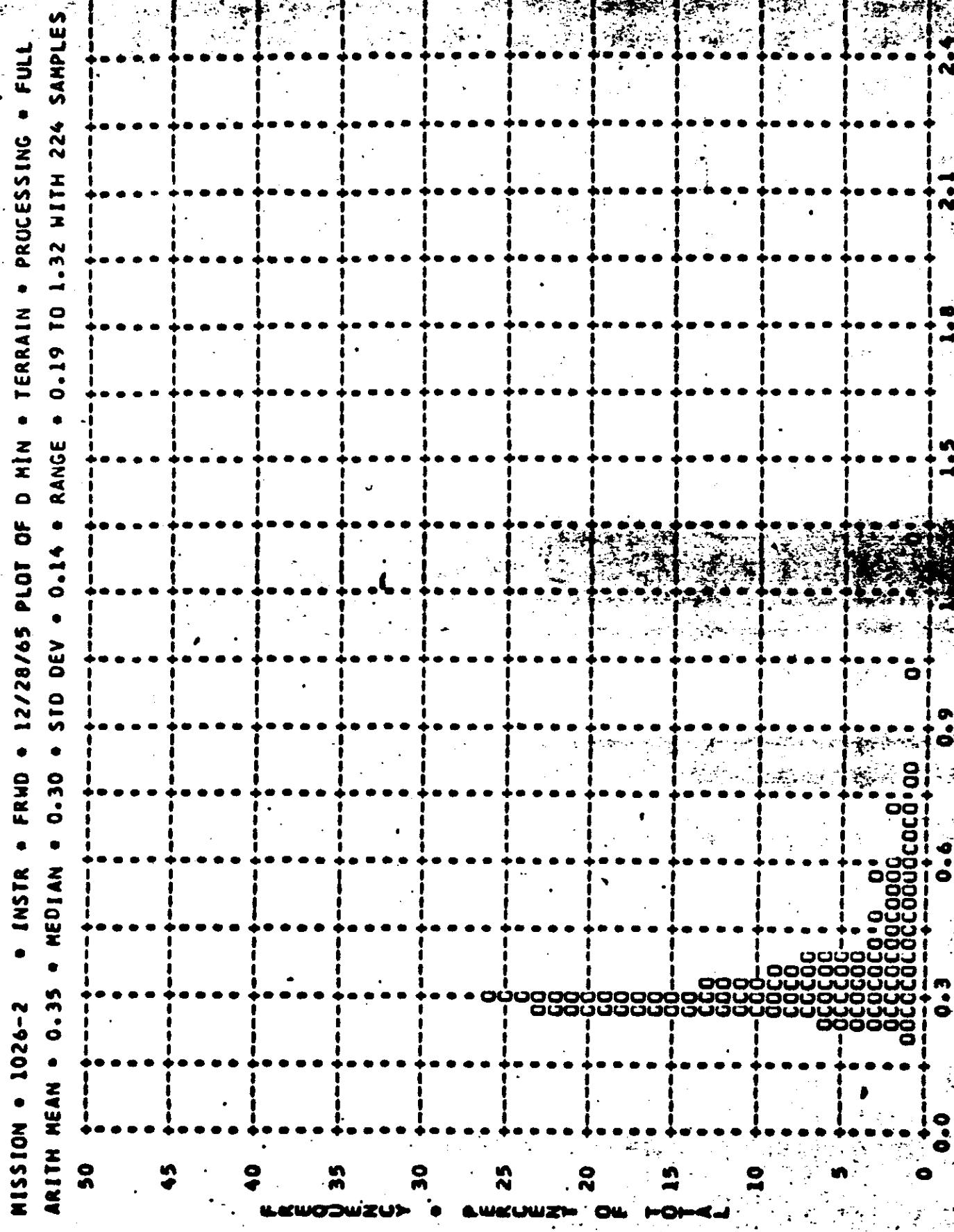
-TOP SECRET

- CL. (ROL NO.



TOP SECRET

C. ROL NO.



A-40

FIGURE A-22

TOP SECRET

TOP SECRET

TOP SECRET

- CONTROL NO.

MISSION • 1026-2 • INSTR • FKWD • 12/28/65 PLOT OF D MAX • TERRAIN • PROCESSING • FULL ARITH MEAN • 1.03 • MEDIAN • 1.01 • STD DEV • 0.45 • RANGE • 0.23 TO 2.17 WITH 224 SAMPLES

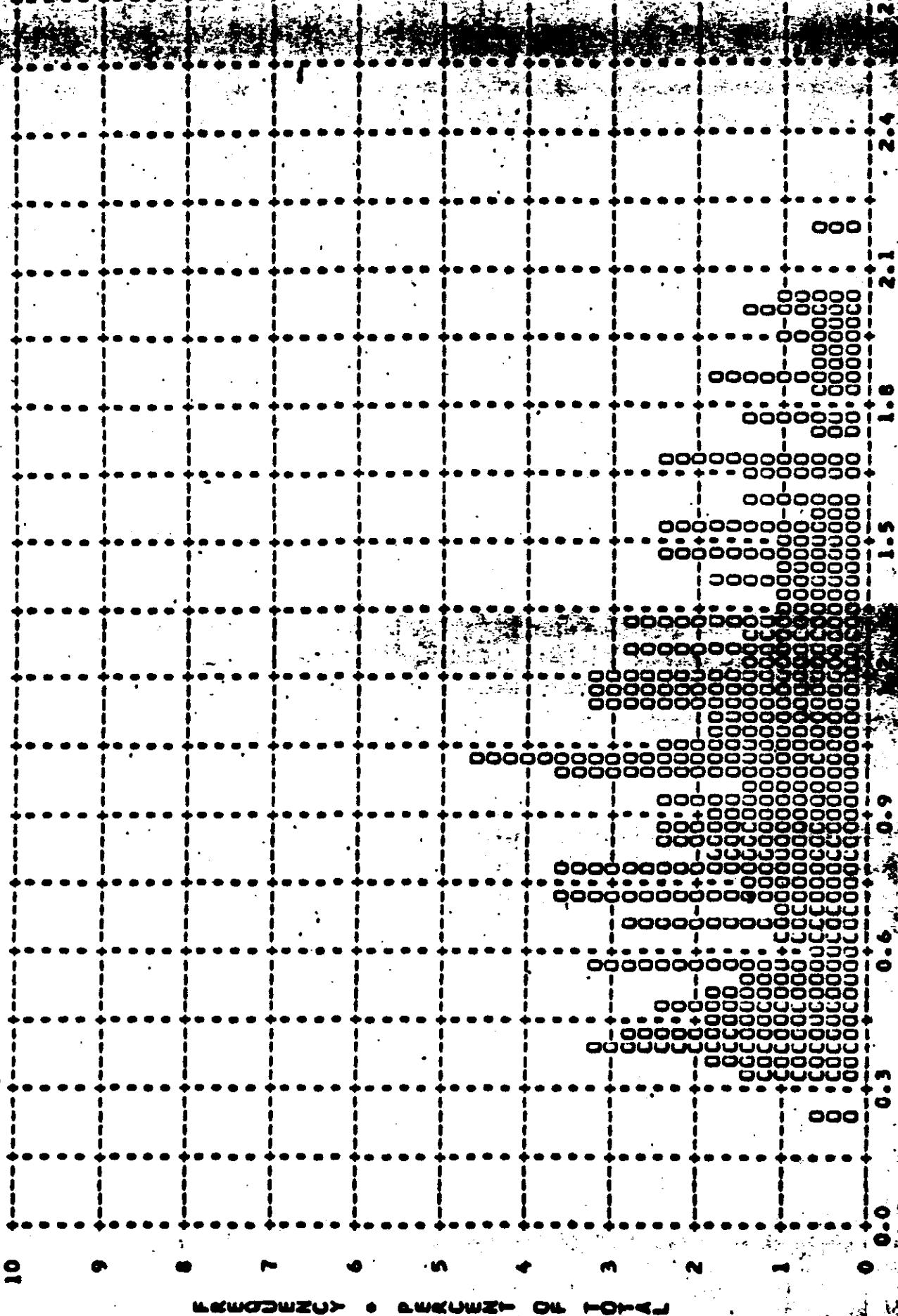
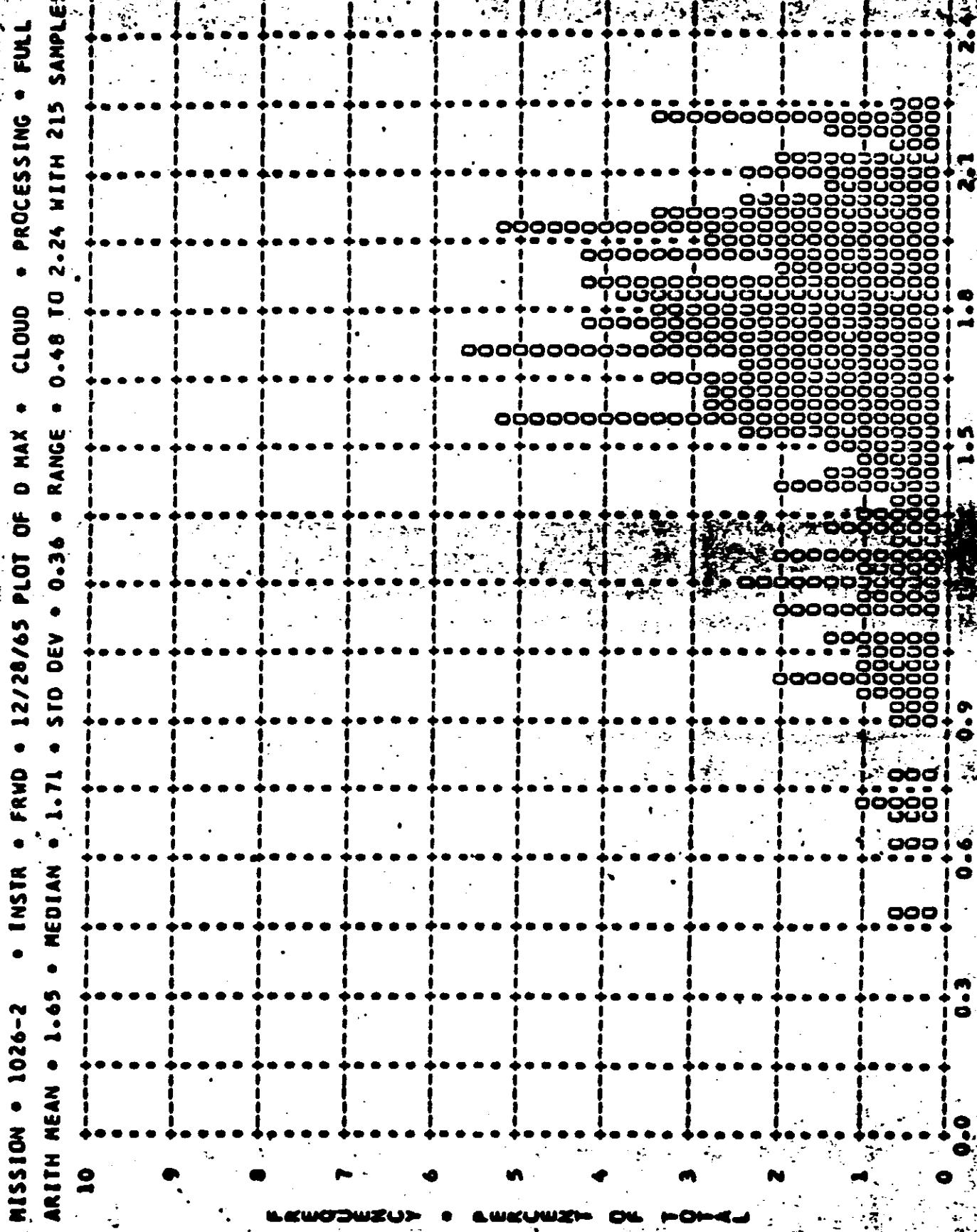


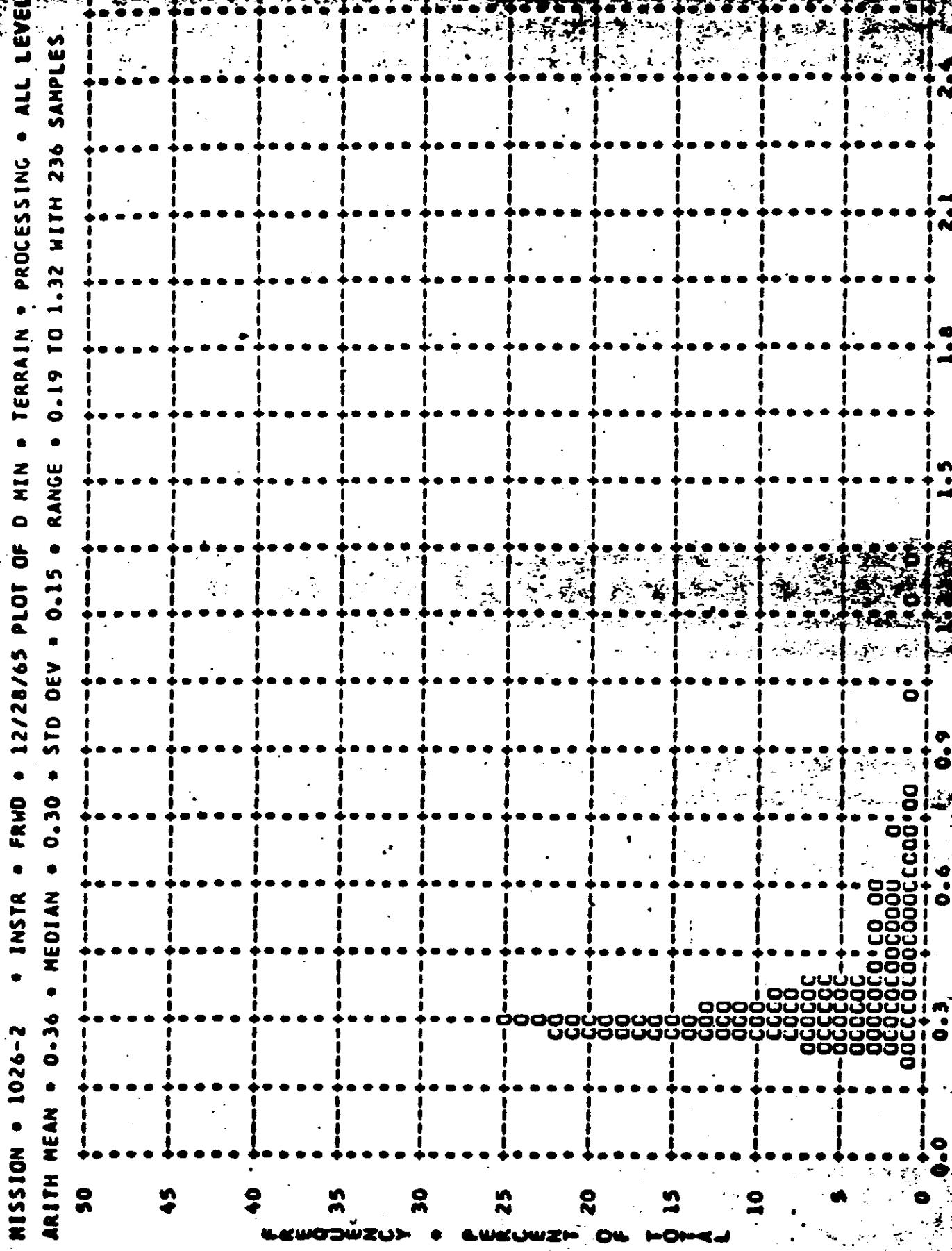
FIGURE A-23

FIGURE A-21



- CONTROL NO.

- TEST



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- CONTROL NO.

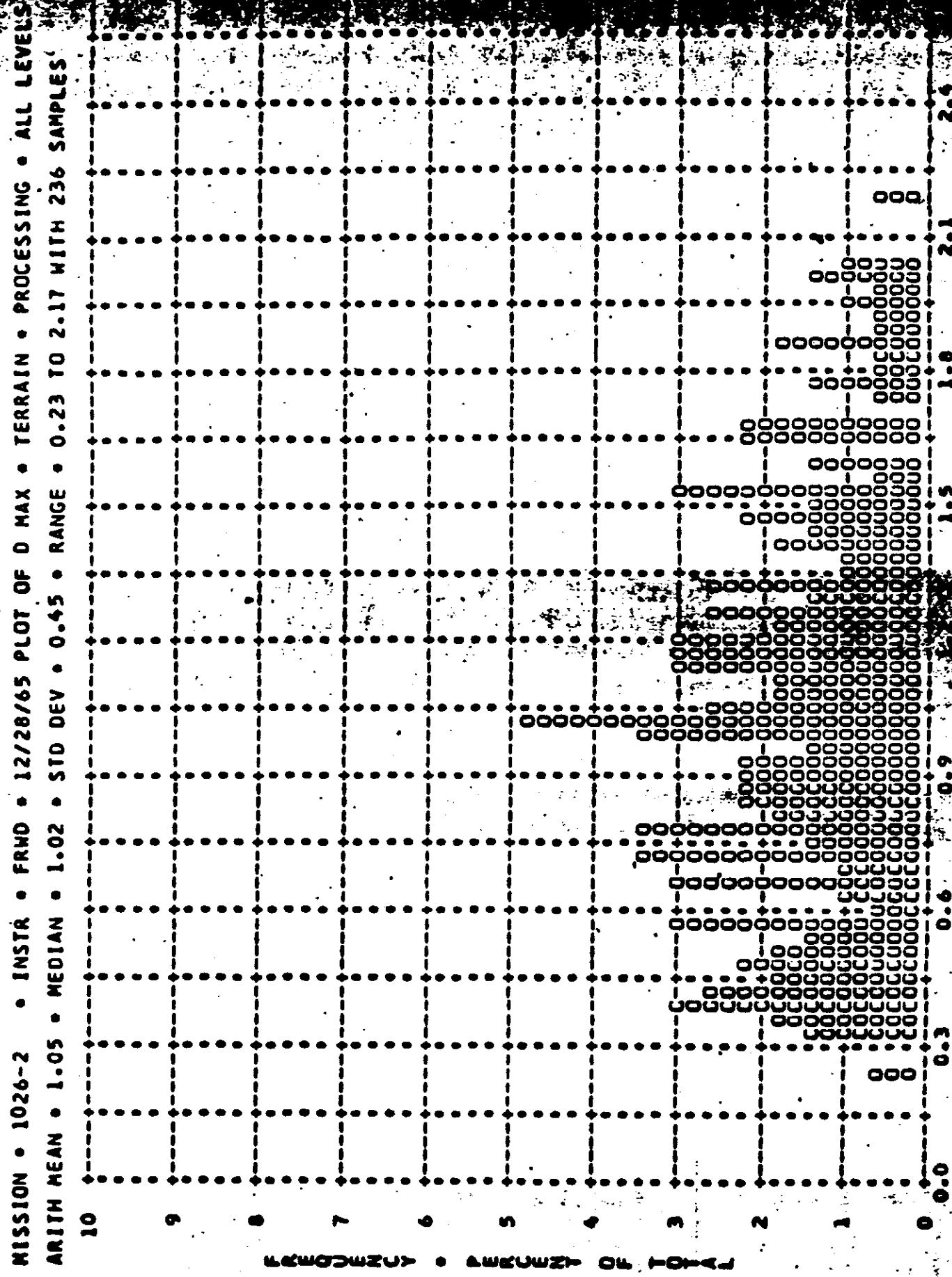


FIGURE A-26

CO. (SQUADRON NO.)

TOP SECRET

MISSION • 1026-2 • INSTR • FRW0 • 12/28/65 PLOT OF D MAX • CLOUD • PROCESSING • ALL LEVELS
ARITH MEAN • 1.63 • MEDIAN • 1.70 • STD DEV • 0.36 • RANGE • 0.48 TO 2.24 WITH 241 SAMPLES

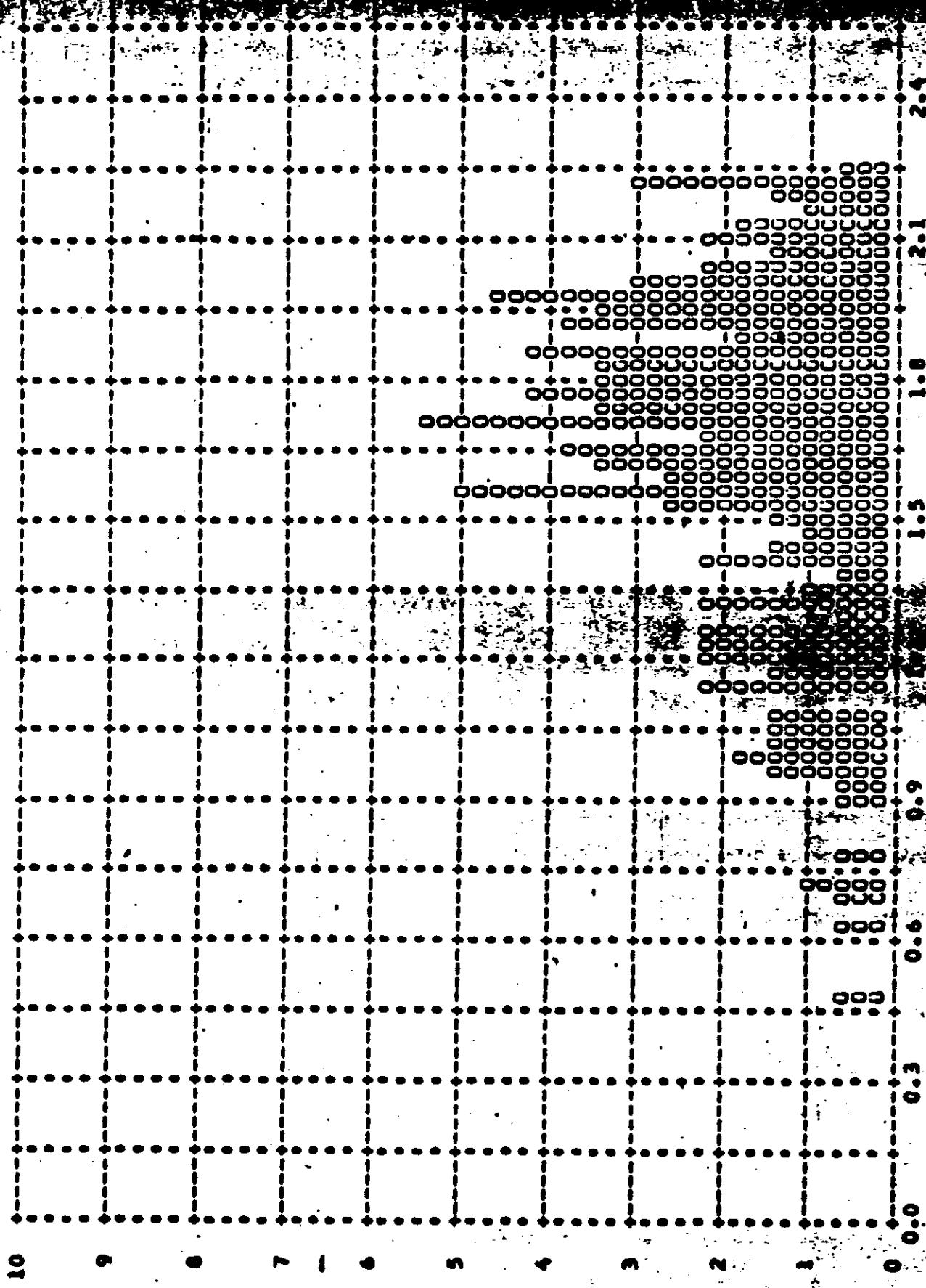


FIGURE A-27

MISSION # 1026-2 INSTRUMENT # AFT DATE 12/26/65 DENSITY FREQ DISTR
 DENSITY 0.01-0.50 MAX LIM 0.01-0.50 MAX LIM 0.01-0.50 MAX LIM
 VALUE 3 NIL MAX LIM 0.01-0.50 MAX LIM 0.01-0.50 MAX LIM
 0.01
 0.02
 0.03
 0.04
 0.05
 0.06
 0.07
 0.08
 0.09
 0.10
 0.11
 0.12
 0.13
 0.14
 0.15
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 0.17
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 0.20
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 0.23
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 0.35
 0.36
 0.37
 0.38
 0.39
 0.40
 0.41
 0.42
 0.43
 0.44
 0.45
 0.46
 0.47
 0.48
 0.49
 0.50
 SUBTOTAL

27	169	22	196	24
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TOP SECRET

CONTROL NO.

TABLE A-4

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SUBTOTAL

TOP SECRET

- CONTROL NO. [REDACTED]

TABLE A-4

2020-2021

CONTROL NO.

TABLE A-4

TOP SECRET

- CONTROL NO.

TABLE A-4

TOP SECRET

- CONTROL NO.

TABLE A-4

MISSION 1026-2 INSTR - AFT 12/28/65 PROCESSING AND EXPOSURE ANALYSIS

PROCESS LEVEL	SAMPLE SIZE	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0	0 PC	0 PC	0 PC	0 PC	0 PC
INTERMEDIATE	34	0 PC	79 PC	12 PC	0 PC	0 PC
FULL	194	70 PC	0 PC	17 PC	0 PC	0 PC
ALL LEVELS	228	60 PC	12 PC	25 PC	0 PC	0 PC
PROCESS LEVEL	BASE + FOG	UNDER EXPOSED	UNDER PROCESSED	CORRECT EXP+PROC	OVER PROCESSED	OVER EXPOSED
PRIMARY	0.01-0.69	0.01-0.13	0.14-0.39	0.40-0.90	-----	0.91 AND UP
INTERMED	0.10-0.17	0.01-0.20	0.21-0.39	0.40-0.90	0.91-1.34	1.35 AND UP
FULL	0.18 AND UP	0.01-0.39	-----	0.40-0.90	0.91-1.69	1.70 AND UP

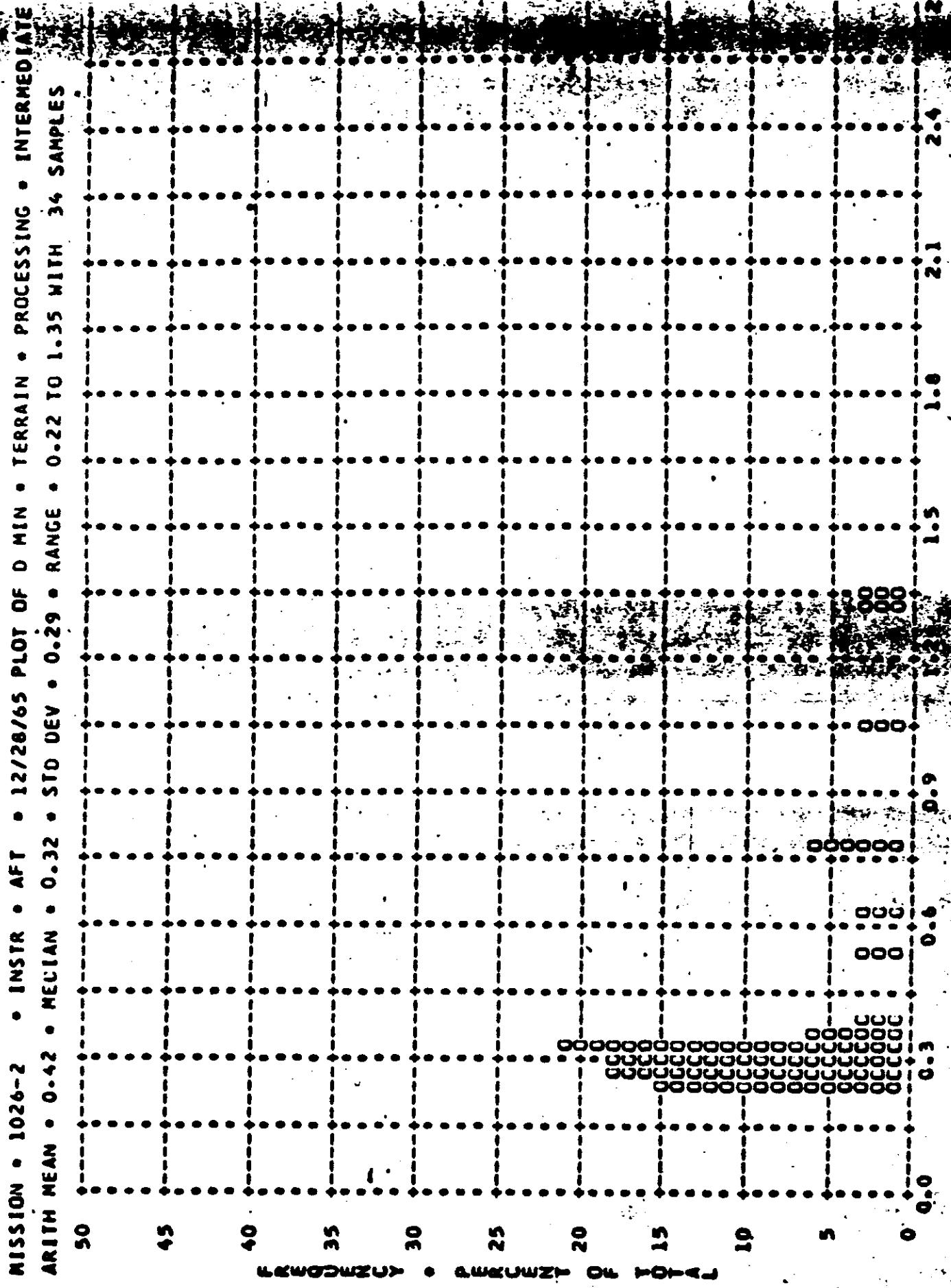
-TOP SECRET

- CONTROL NO.

TABLE A-4

- C. SROL NO.

TOP SECRET



COL/ROL NO.

TOP SECRET

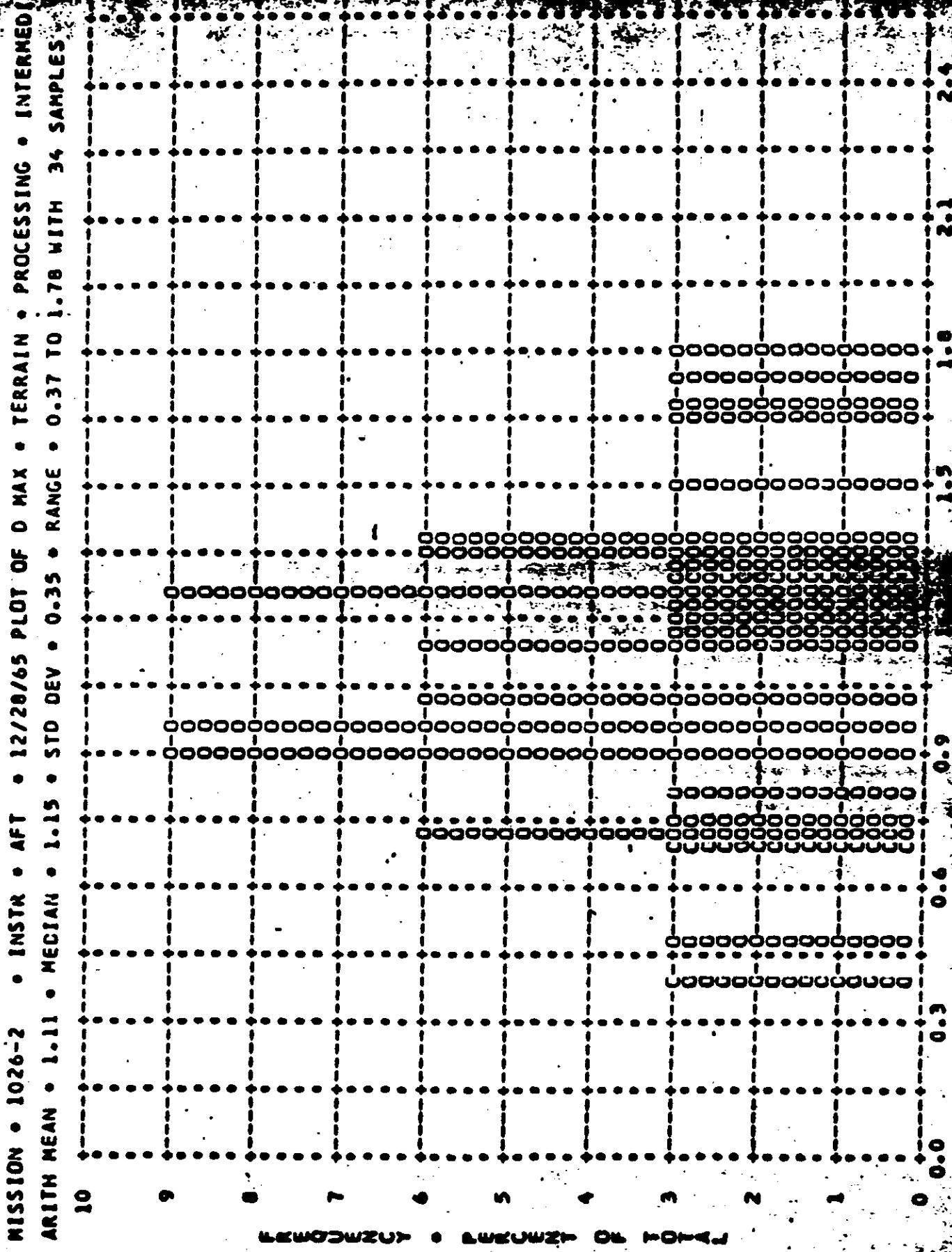


FIGURE A-29

- COL NO.

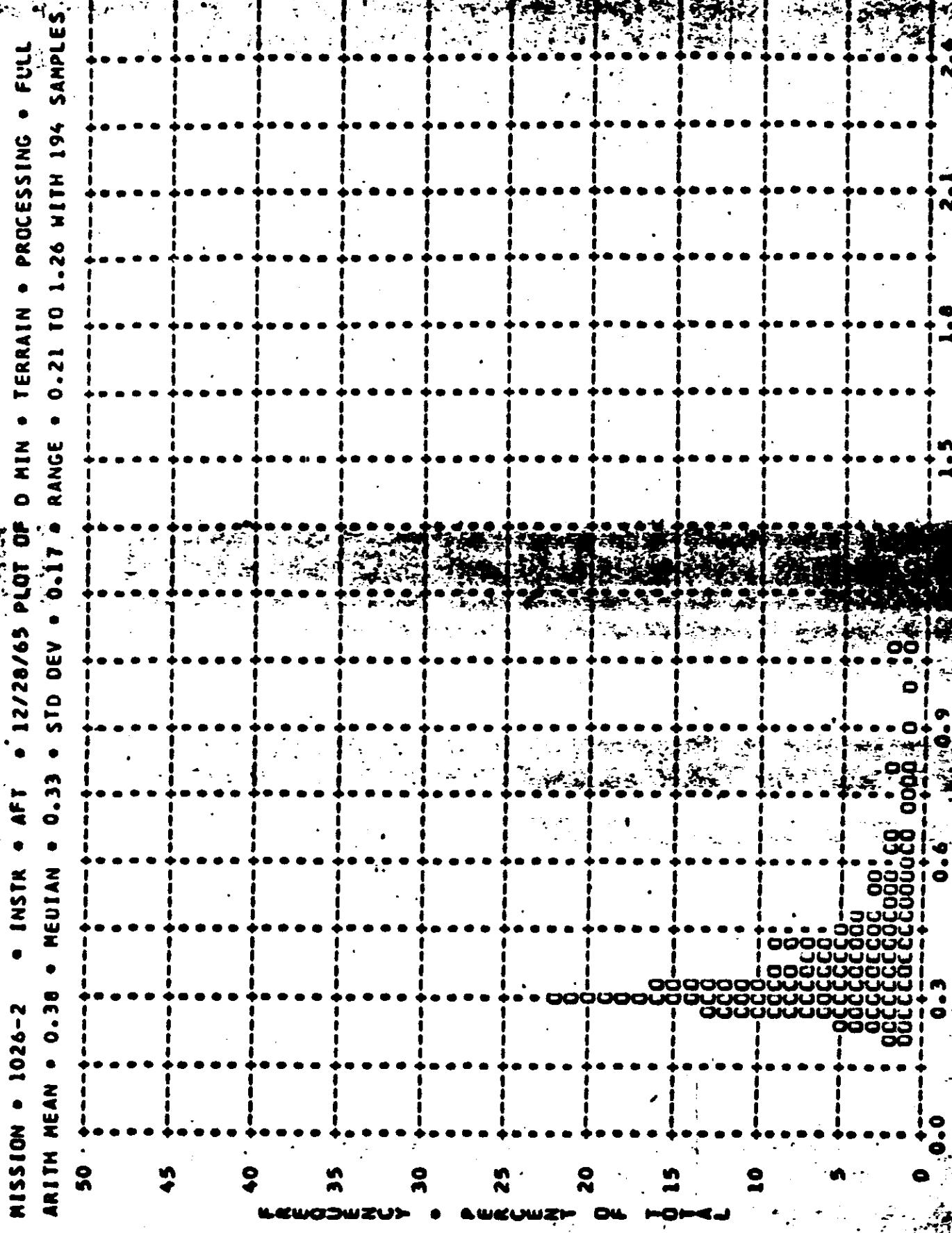
TOP ESCROW

MISSION • 1026-2 • INSTR • AFT • 12/28/65 PLOT OF 0 MAX • CLOUD • PROCESSING • INTERMEDIATE
ARITH MEAN • 1.78 • MEDIAN • 1.86 • STD DEV • 0.35 • RANGE • 0.80 10 2.24 WITH 54 SAMPLES



- CDR IND. NO.

TOP SECRET



- CONTROL NO.

TEST REPORT

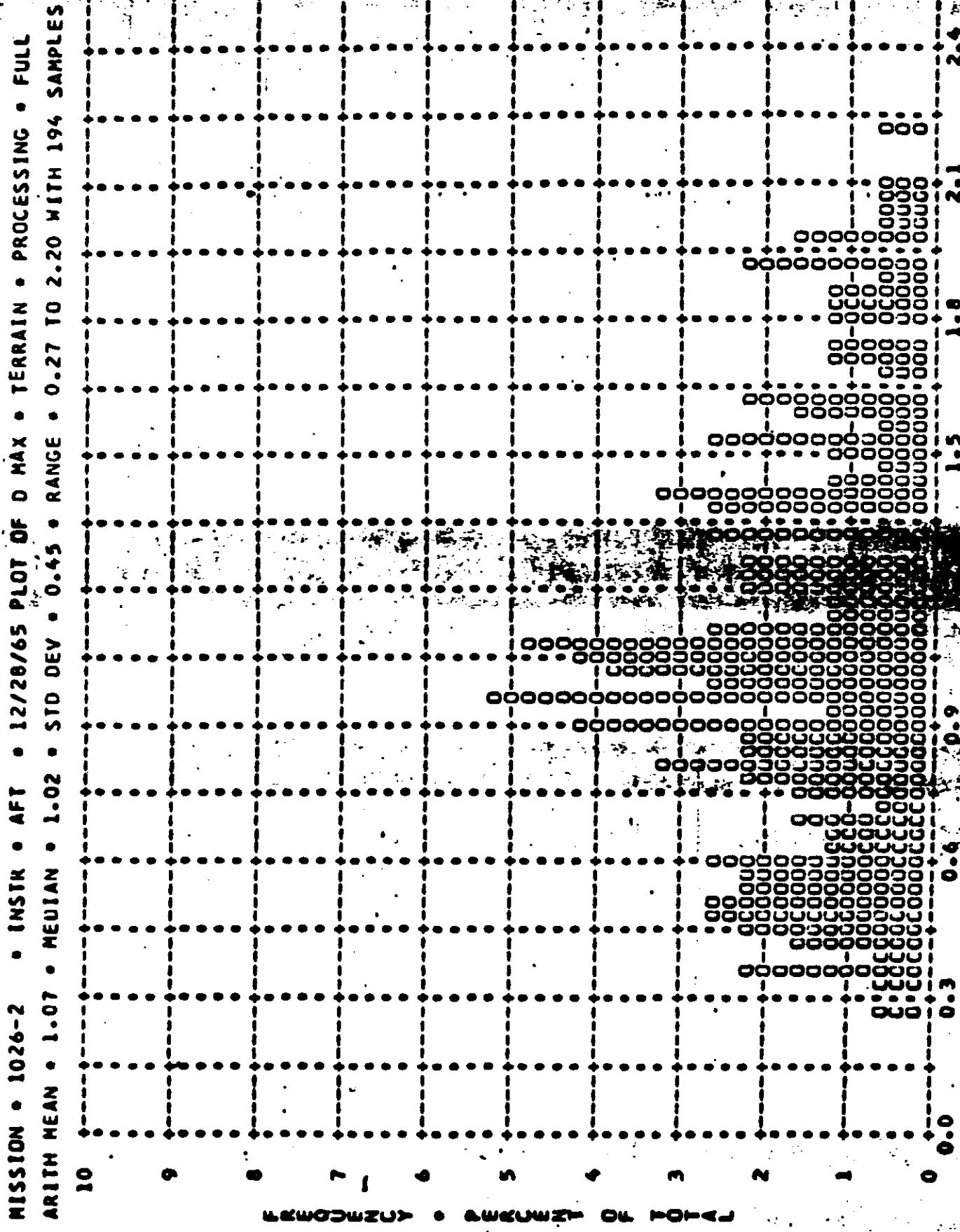
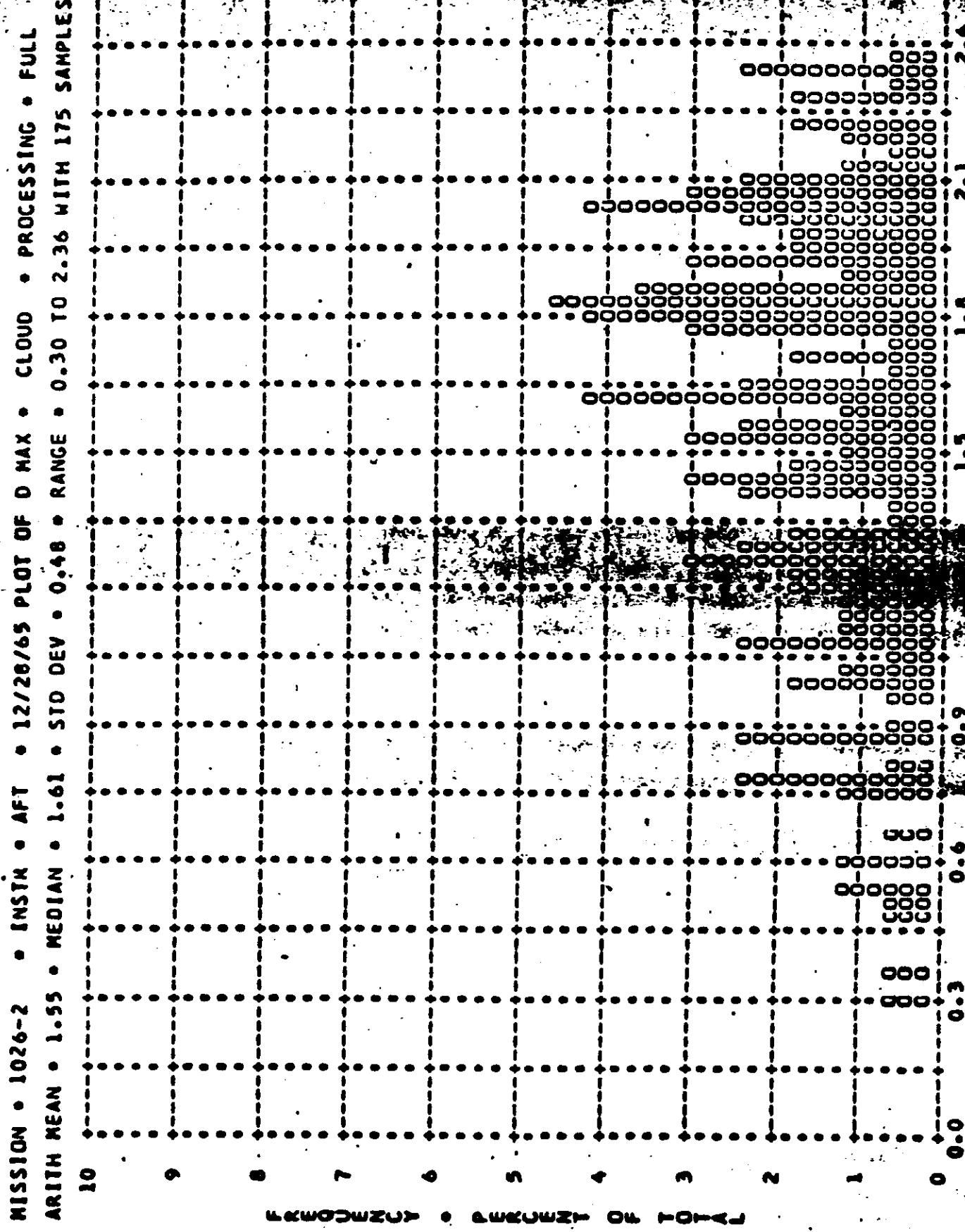


FIGURE A-32

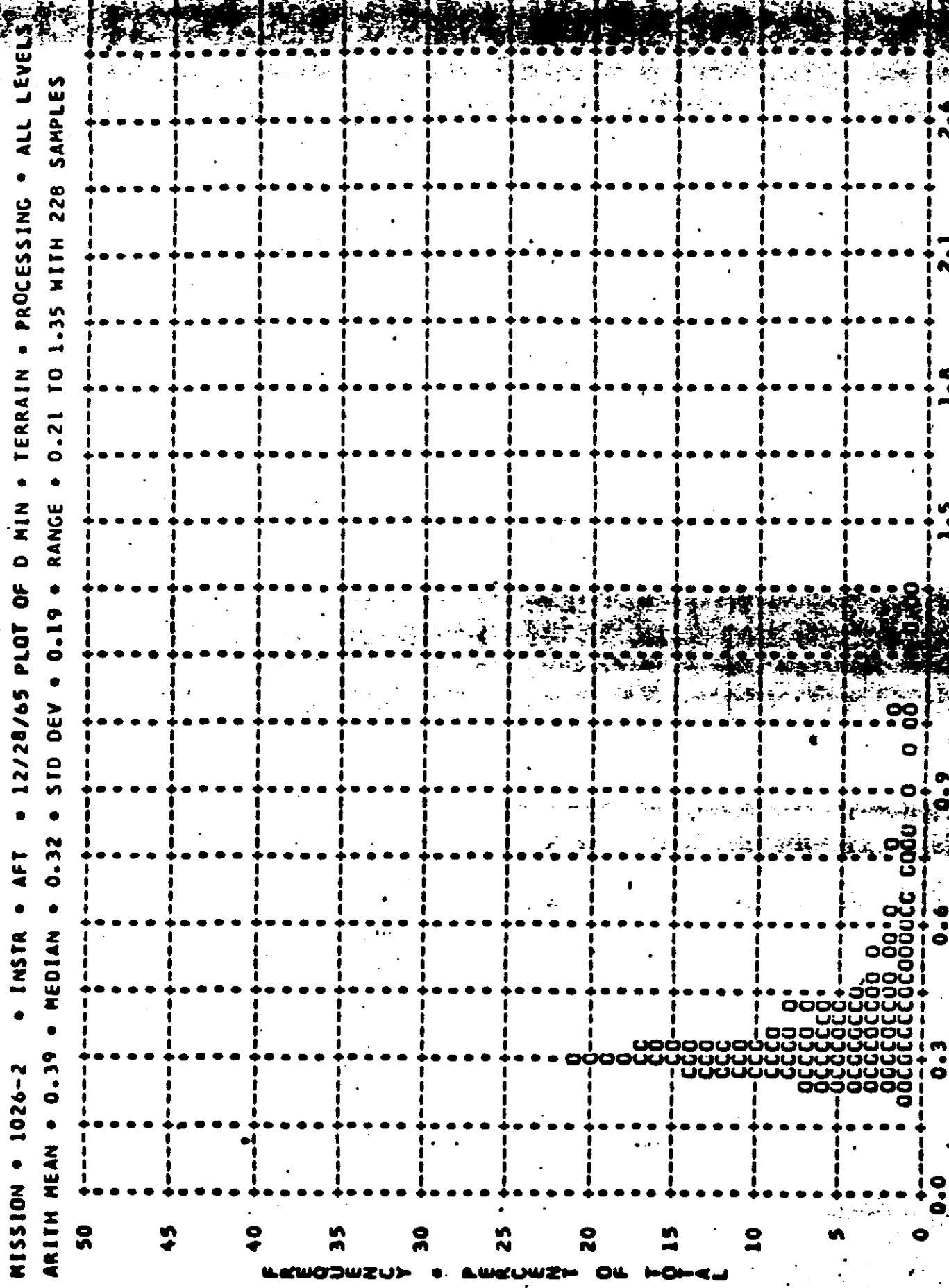
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- CL/DOC NO.



Percent

- CONTROL NO.



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CLASSE NO.

MISSION • 1026-2 • INSTR • AFT • 12/28/65 PLOT OF D MAX • TERRAIN • PROCESSING • ALL LEVEL
ARITH MEAN • 1.06 • MEDIAN • 1.03 • STD DEV • 0.43 • RANGE • 0.27 TO 2.20 WITH 228 SAMPLES

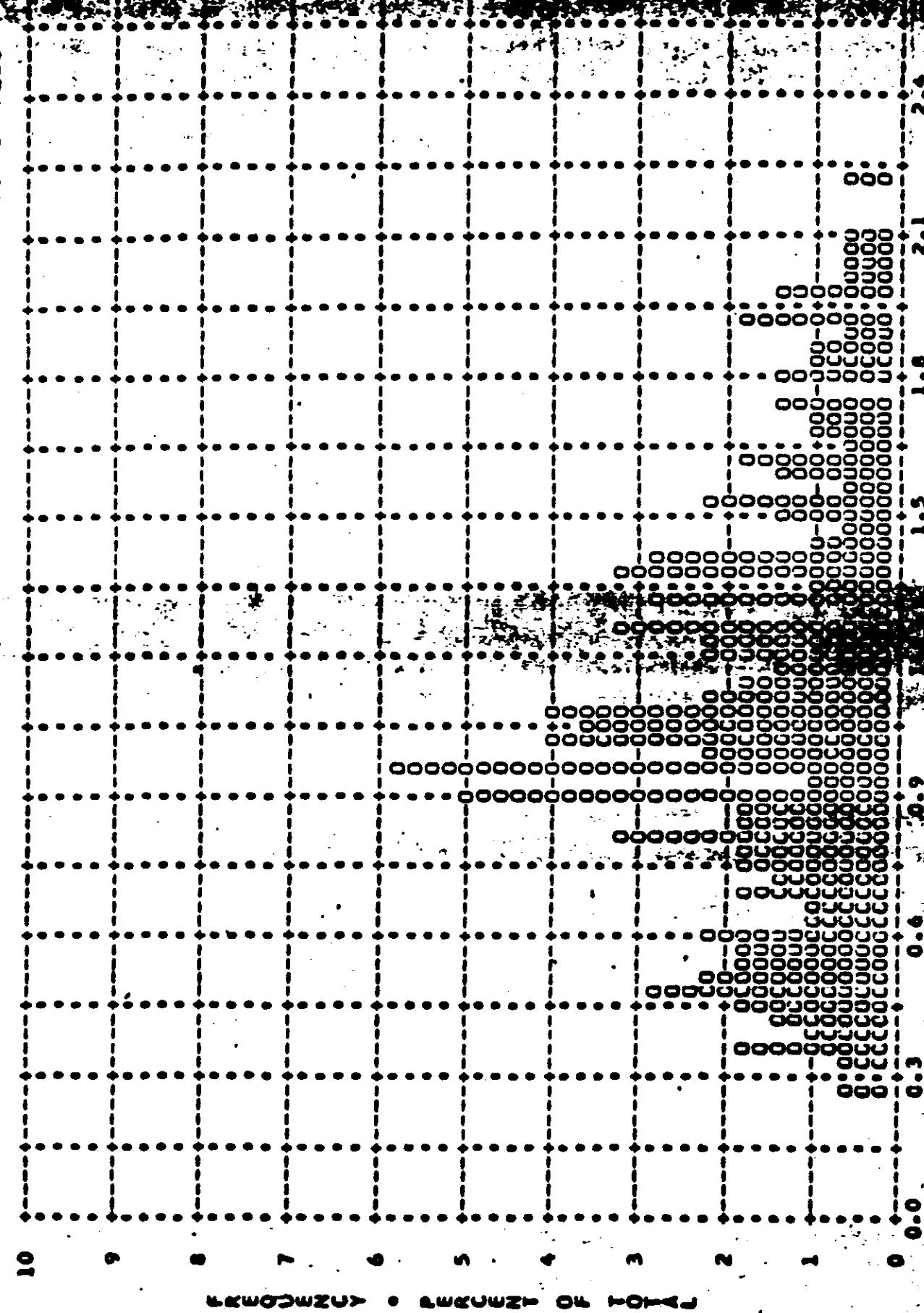


FIGURE A-35

900 SECONDS

MISSION • 1026-2 • INSTR • AFI • 12/28/65
ARITH MEAN • 1.61 • MEDIAN • 1.70 • STD DEV • 0.30 TO 2.36 WITH 229 SAMPLES

D MAX • CLOUD • PROCESSING • ALL

RANGE • 0.30 TO 2.36 WITH 229 SAMPLES

10

MEASUREMENTS • MEASURED ON PAPER

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